**WARNING:** Navigation and terrain separation must NOT be predicated upon the use of the terrain function. The G1000 Terrain Proximity feature is NOT intended to be used as a primary reference for terrain avoidance and does not relieve the pilot from the responsibility of being aware of surroundings during flight. The Terrain Proximity feature is only to be used as an aid for terrain avoidance and is not certified for use in applications requiring a certified terrain awareness system. Terrain data is obtained from third party sources. Garmin is not able to independently verify the accuracy of the terrain data.

**WARNING:** The displayed minimum safe altitudes (MSAs) are only advisory in nature and should not be relied upon as the sole source of obstacle and terrain avoidance information. Always refer to current aeronautical charts for appropriate minimum clearance altitudes.

**WARNING:** The altitude calculated by G1000 GPS receivers is geometric height above Mean Sea Level and could vary significantly from the altitude displayed by pressure altimeters, such as the GDC 74A Air Data Computer, or other altimeters in aircraft. GPS altitude should never be used for vertical navigation. Always use pressure altitude displayed by the G1000 PFD or other pressure altimeters in aircraft.

**WARNING:** Do not use outdated database information. Databases used in the G1000 system must be updated regularly in order to ensure that the information remains current. Pilots using any outdated database do so entirely at their own risk.

**WARNING:** Do not use basemap (land and water data) information for primary navigation. Basemap data is intended only to supplement other approved navigation data sources and should be considered as an aid to enhance situational awareness.

**WARNING:** Traffic information shown on the G1000 Multi Function Display is provided as an aid in visually acquiring traffic. Pilots must maneuver the aircraft based only upon ATC guidance or positive visual acquisition of conflicting traffic.

**WARNING:** Use of the Stormscope is not intended for hazardous weather penetration (thunderstorm penetration). Stormscope information, as displayed on the G1000 MFD, is to be used only for weather avoidance, not penetration.

**WARNING:** GDL 69 Weather should not be used for hazardous weather penetration. Weather information provided by the GDL 69 is approved only for weather avoidance, not penetration.
**WARNING:** NEXRAD weather data is to be used for long-range planning purposes only. Due to inherent delays in data transmission and the relative age of the data, NEXRAD weather data should not be used for short-range weather avoidance.

**WARNING:** The Garmin G1000, as installed in Cessna Nav III aircraft, has a very high degree of functional integrity. However, the pilot must recognize that providing monitoring and/or self-test capability for all conceivable system failures is not practical. Although unlikely, it may be possible for erroneous operation to occur without a fault indication shown by the G1000. It is thus the responsibility of the pilot to detect such an occurrence by means of cross-checking with all redundant or correlated information available in the cockpit.

**WARNING:** For safety reasons, G1000 operational procedures must be learned on the ground.

**WARNING:** The United States government operates the Global Positioning System and is solely responsible for its accuracy and maintenance. The GPS system is subject to changes which could affect the accuracy and performance of all GPS equipment. Portions of the Garmin G1000 utilize GPS as a precision electronic NAVigation AID (NAVAID). Therefore, as with all NAVAIDs, information presented by the G1000 can be misused or misinterpreted and, therefore, become unsafe.

**WARNING:** To reduce the risk of unsafe operation, carefully review and understand all aspects of the G1000 Pilot’s Guide documentation. Thoroughly practice basic operation prior to actual use. During flight operations, carefully compare indications from the G1000 to all available navigation sources, including the information from other NAVAIDs, visual sightings, charts, etc. For safety purposes, always resolve any discrepancies before continuing navigation.

**WARNING:** The illustrations in this guide are only examples. Never use the G1000 to attempt to penetrate a thunderstorm. Both the FAA Advisory Circular, Subject: Thunderstorms, and the Airman’s Information Manual (AIM) recommend avoiding “by at least 20 miles any thunderstorm identified as severe or giving an intense radar echo."

**CAUTION:** The GDU 1040 and GDU 1044B PFD/MFD displays use a lens coated with a special anti-reflective coating that is very sensitive to skin oils, waxes, and abrasive cleaners. CLEANERS CONTAINING AMMONIA WILL HARM THE ANTI-REFLECTIVE COATING. It is very important to clean the lens using a clean, lint-free cloth and an eyeglass lens cleaner that is specified as safe for anti-reflective coatings.
**CAUTION:** The Garmin G1000 does not contain any user-serviceable parts. Repairs should only be made by an authorized Garmin service center. Unauthorized repairs or modifications could void both the warranty and the pilot’s authority to operate this device under FAA/FCC regulations.

**NOTE:** When using Stormscope, there are several atmospheric phenomena in addition to nearby thunderstorms that can cause isolated discharge points in the strike display mode. However, clusters of two or more discharge points in the strike display mode do indicate thunderstorm activity if these points reappear after the screen has been cleared.

**NOTE:** All visual depictions contained within this document, including screen images of the G1000 panel and displays, are subject to change and may not reflect the most current G1000 system. Depictions of equipment may differ slightly from the actual equipment.

**NOTE:** This device complies with part 15 of the FCC Rules. Operation is subject to the following two conditions: (1) this device may not cause harmful interference, and (2) this device must accept any interference received, including interference that may cause undesired operation.

**NOTE:** The GDU 1040 and GDU 1044B PFD/MFD may require a warm-up time of up to 30 minutes when exposed to -40°C for an extended period. A warm-up time of up to 15 minutes may be required when exposed to -30°C for an extended period.

**NOTE:** This product, its packaging, and its components contain chemicals known to the State of California to cause cancer, birth defects, or reproductive harm. This notice is being provided in accordance with California’s Proposition 65. If you have any questions or would like additional information, please refer to our web site at www.garmin.com/prop65.
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SECTION 1: SYSTEM OVERVIEW

The purpose of this Cockpit Reference Guide is to provide the pilot a resource with which to find operating instructions on the major features of the G1000 system more easily. **It is not intended to be a comprehensive operating guide.** Complete operating procedures for the complete system are found in the Garmin G1000 Pilot's Guide for the Cessna Nav III (190-00498-01).

This guide gives the pilot abbreviated operating instructions for the Primary Flight Display (PFD), Multi Function Display (MFD), and the GMA 1347 Audio Panel System.

**NOTE:** The pilot should read and thoroughly understand the Cessna Pilot’s Operating Handbook (POH) for limitations, procedures and operational information not contained in this Cockpit Reference Guide. The Cessna POH always takes precedence over the information found in this guide.
1.1 PFD/MFD CONTROLS

Figure 1-1  PFD/MFD Controls
PFD and MFD controls function the same.

1. **NAV VOL/ID Knob** – Controls the NAV audio level. Press to turn the Morse code identifier ON and OFF. Volume level is shown in the field as a percentage.

2. **NAV Frequency Transfer Key** – Transfers the standby and active NAV frequencies.

3. **Dual NAV Knob** – Tunes the MHz (large knob) and kHz (small knob) standby frequencies for the NAV receiver. Press to switch the tuning box (light blue box) between the NAV1 and NAV2 fields.

4. **Heading Knob** – Turn to manually select a heading on the HSI. When pressed, it synchronizes the heading bug with the compass lubber line. Selected Heading provides the heading reference to the Flight Director while operating in Heading Select mode.

5. **Joystick** – Changes the map range (distance top to bottom of map display) when rotated. Activates the map pointer when pressed.

6. **CRS/BARO Knob** – The large knob sets the altimeter barometric pressure and the small knob adjusts the course. The course is only adjustable when the HSI is in VOR1, VOR2, or OBS/SUSP mode. Pressing this knob centers the CDI on the currently selected VOR. Selected Course provides course reference to the Flight Director when operating in Navigation and Approach modes.

7. **Dual COM Knob** – Tunes the MHz (large knob) and kHz (small knob) standby frequencies for the COM transceiver. Pressing this knob switches the tuning box (light blue box) between the COM1 and COM2 fields.

8. **COM Frequency Transfer Key** – Transfers the standby and active COM frequencies. Pressing and holding this key for two seconds automatically tunes the emergency frequency (121.5 MHz) in the active frequency field.

9. **COM VOL/SQ Knob** – Controls COM audio level. Pressing this knob turns the COM automatic squelch ON and OFF. Audio volume level is shown in the field as a percentage.

10. **Direct-to Key** – Allows the user to enter a destination waypoint and establish a direct course to the selected destination (specified by the identifier, chosen from the active route, or taken from the map pointer position).

11. **FPL Key** – Displays the active Flight Plan Page for creating and editing the active flight plan, or for accessing stored flight plans.

12. **CLR Key (DFLT MAP)** – Erases information, cancels an entry, or removes page menus. To display the Navigation Map Page immediately, press and hold CLR (MFD only).

13. **Dual FMS Knob** – Used to select the page to be viewed (only on the MFD). The large knob selects a page group (MAP, WPT, AUX, NRST), while the small knob selects a specific page within the page group. Pressing the small knob turns the selection cursor ON and OFF. When the cursor is ON, data may be entered in the different fields using the small and large knobs. The large knob is used to move the cursor on the page, while the small knob is used to select individual characters for the highlighted cursor location. When the G1000 displays a list that is too long for the display screen, a scroll bar appears along the right
side of the display, indicating the availability of additional items within the selected category. Press the small **FMS** Knob to activate the cursor and turn the large **FMS** Knob to scroll through the list.

**MENU Key** – Displays a context-sensitive list of options. This list allows the user to access additional features, or to make setting changes that relate to certain pages.

**PROC Key** – Selects approaches, departures and arrivals from the flight plan. If a flight plan is used, available procedures for the departure and/or arrival airport are automatically suggested. If a flight plan is not used, the desired airport and the desired procedure may be selected. This key selects IFR departure procedures (DPs), arrival procedures (STARs) and approaches (IAPs) from the database and loads them into the active flight plan.

**ENT Key** – Accepts a menu selection or data entry. This key is used to approve an operation or complete data entry. It is also used to confirm selections and information entries.

**Dual ALT Knob** – Sets the selected altitude in the box located above the Altimeter. The large knob selects the thousands, while the small knob selects the hundreds. Altitude Select is used by the Automatic Flight Control System in certain modes, in addition to the standard G1000 Altitude Alerter function.

**AP Key** – (182 & 206 only) Engages/disengages the Autopilot and Flight Director. Pressing the **AP** Key activates the Flight Director and engages the Autopilot in the default pitch axis and roll axis modes. Pressing the **AP** Key again disengages the autopilot and deactivates the Flight Director.

**HDG Key** – (182 & 206 only) Selects/deselects the Heading Select mode.

**NAV Key** – (182 & 206 only) Selects/deselects the Navigation mode.

**APR Key** – (182 & 206 only) Selects/deselects the Approach mode.

**VS Key** – (182 & 206 only) Selects/deselects the Vertical Speed mode.

**FLC Key** – (182 & 206 only) Selects/deselects the Flight Level Change mode.

**FD Key** – (182 & 206 only) Activates/deactivates the Flight Director only. Pressing the **FD** Key turns on the Flight Director in the default pitch axis and roll axis modes. Pressing the **FD** Key again deactivates the Flight Director and removes the command bars, unless the Autopilot is engaged. If the Autopilot is engaged, the **FD** Key is disabled.

**ALT Key** – (182 & 206 only) Selects/deselects the Altitude Hold mode.

**VNV Key** – (182 & 206 only) Selects/deselects Vertical Navigation mode.

**BC Key** – (182 & 206 only) Selects/deselects Back Course mode.

**NOSE UP/NOSE DN Keys** – (182 & 206 only) Controls the active pitch reference for the Pitch Hold, Vertical Speed, and Flight Level Change modes.
1.2 PFD SOFTKEYS

**INSET** – Press to display the Inset Map in the lower left corner of the PFD.

**OFF** – Press to remove the Inset Map.

**DCLTR** (3) – Press momentarily to select the desired amount of map detail. The declutter level appears adjacent to the **DCLTR** Softkey.
- No declutter: All map features are visible
- Declutter – 1: Declutters land data
- Declutter – 2: Declutters land and SUA data
- Declutter – 3: Removes everything except the active flight plan

**TRAFFIC** – Press to display traffic on the inset map.

**TOPO** – Press to display topographical data (i.e., coastlines, terrain, rivers, lakes) and elevation scale on the inset map.

**TERRAIN** – Press to display terrain information on the inset map.

**STRMSCP** (optional) – Press to display the Stormscope lightning data on the inset map (within a 200 nm radius of the aircraft).

**NEXRAD** (optional) – Press to display NEXRAD precipitation data on the inset map.

**XM LTNG** (optional) – Press to display the XM Radio lightning data on the inset map.

**BACK** – Press to return to the previous level softkey configuration.

**ALERTS** – Press to display the Alerts Window.

**PFD** – Press to display the additional softkeys for additional configuration of the PFD.

**METRIC** – Press to display the current and reference altitudes in meters, in addition to feet. Pressing the metric softkey also changes the barometric setting to hectopascals.

**DFLTS** – Press to reset default settings on the PFD.

**DME** (optional) – Press to display the DME Information Window.

**BRG1** (bearing) – Press to cycle through the following Nav sources, making the pointer the indicator for the corresponding source and displaying the appropriate information.
- **NAV1** – Displays NAV1 waypoint frequency or identifier and DME information in the BRG1 Information Window.
- **GPS** – Displays GPS waypoint identifier and GPS distance information in the BRG1 Information Window.
- **ADF** – Displays ADF in the BRG1 Information Window.
- **OFF** – Removes the BRG1 Information Window.

![Figure 1-2 PFD Top Level Softkeys](Image)

![Figure 1-3 PFD Softkey Flow Chart – 1](Image)
**BRG2** (bearing) – Press to cycle through the following Nav sources, making the pointer the indicator for the corresponding source and displaying the appropriate information.

**NAV2** – Displays NAV2 waypoint frequency or identifier and DME information in the BRG2 Information Window.

**GPS** – Displays GPS waypoint identifier and GPS distance information in the BRG2 Information Window.

**ADF** – Displays ADF in the BRG2 Information Window.

**OFF** – Removes the BRG2 Information Window.

**STD BARO** – Press to set the barometric pressure to 29.92 inches of mercury (1013 hPa by pressing the **METRIC** Softkey).

**BACK** – Press to return to the previous level softkeys.

**ALERTS** – Press to display the Alerts Window.

**OBS** – Press to select OBS mode on the CDI when navigating by GPS (only available with active leg).

**CDI** – Press to change navigation mode on the CDI between GPS, VOR1, and VOR2.

**DME** (optional) – Press to display the DME Tuning Window.

**XPDR** – Press to display the transponder mode selection softkeys.

**STBY** – Press to select standby mode.

**ON** – Press to select mode A.

**ALT** – Press to select altitude reporting mode.

**GND** – Press to select ground mode.

**VFR** – Press to automatically squawk 1200 (only in the U.S.A., refer to ICAO standards for VFR codes in other countries).

**CODE** – Press to display transponder code selection softkeys 0-7.

0 through 7 – Press numbers to enter code.
SECTION 1
SYSTEM OVERVIEW

**1.3 MFD SOFTKEYS**

**ENGINE** – Pressing this softkey makes available the LEAN and SYSTEM Softkeys which in turn access the Lean Page and the System Page, respectively.

**MAP** – Pressing this softkey enables the following softkeys:

- **TRAFFIC** – Pressing this softkey displays/removes Traffic on the Navigation Map.
- **TOPO** – Pressing this softkey displays or removes topographic information on the Navigation Map.
- **TERRAIN** – Pressing this softkey displays/removes terrain and obstacle data on the Navigation Map.

**IDENT** – Press to provide special aircraft position identification to Air Traffic Control (ATC).

**BKSP** – Press to remove numbers entered one at a time.

**BACK** – Press to return to the previous level softkeys.

**IDENT** – Press to provide special aircraft position identification to Air Traffic Control (ATC).

**BACK** – Press to return to the previous level softkeys.

**ALERTS** – Press to display the Alerts Window.

**IDENT** – Press to provide special aircraft position identification to Air Traffic Control (ATC).

**BACK** – Press to return to the previous level softkeys.

**ALERTS** – Press to display the Alerts Window.
**Figure 1-6 MFD Softkeys**

**STRMSCP** (optional) – Pressing this softkey displays/removes Stormscope lightning data on the Navigation Map.

**NEXRAD** (optional) – Pressing this softkey displays/removes precipitation data on the Navigation Map.

**XM LTNG** (optional) – Pressing this softkey displays/removes XM Radio lightning data on the Navigation Map.

**BACK** – Pressing this softkey displays the ENGINE and MAP top level softkeys.

**DCLTR** (declutter) – Pressing this softkey removes map information in three levels.

**CHKLIST** (checklist)(optional) – Pressing the CHKLIST Softkey displays the Checklist Page.
1.4 **MFD PAGE GROUPS**

1) Turn the large **FMS** Knob until the desired page group is selected.

2) Turn the small **FMS** Knob to select pages within the group. See Figure 1-7.

![Figure 1-7 Page Group Icon](image)

**Nearest Group**

**Auxiliary Page Group**

**Waypoint Page Group**

**Map Page Group**

**MAP WPT AUX NRST**

**Number of Pages in Current Group**

**Selected Page**

1.5 **VERTICAL NAVIGATION**

The G1000 system can use altitude constraints associated with lateral waypoints to give guidance for vertical navigation. These altitudes are, depending on the specific instance, entered by the pilot or retrieved from the published altitudes in the navigation database.

The navigation database only contains altitudes for procedures that call for “Cross at” altitudes. If the procedure states “Expect to cross at,” then the altitude will not be in the database. In this case the altitude may be entered manually.

**NOTE:** All arrival procedure altitudes contained in the navigation database are for turbojet aircraft only. If necessary, alter or enter altitudes to be consistent with the requirements of the Cessna Nav III aircraft.

When activating or loading an arrival or approach procedure into an active flight plan, the VNAV ‘ALT’ fields will be populated with any altitudes that can be retrieved from the navigation database.

Since altitudes loaded with an arrival procedure are published only for turbojet aircraft, the altitudes are displayed as white text indicating that the altitudes are displayed for reference only. An arrival waypoint altitude may be used (or “designated”) as is, or changed to a different altitude. An altitude is designated by placing the cursor on the desired altitude and pressing the **ENT** Key or entering a different value and press the **ENT** Key. The altitude will now be displayed as blue text, indicating that the altitude is now designated to give vertical speed and deviation guidance.

Altitudes associated with an approach procedure are “auto-designated”. This means the system will automatically use the altitudes loaded with the approach for giving vertical speed and deviation guidance. Note that these altitudes will be displayed as blue text up to, but not including the FAF. The FAF will always be a “reference only” altitude and cannot be designated.

Altitudes that have been designated for use in vertical guidance may also be made “non-designated” by placing the cursor over the desired altitude and pressing the **CLR** Key. Other displayed altitudes may change due to re-calculation.
or rendered invalid as a result of manually changing an altitude to a non-designated altitude.

To help interpret the meanings of how the altitudes are presented, keep the following points in mind:

- When the altitude is displayed in light blue, the system is using that altitude (designated) to determine vertical speed and deviation guidance.
- When the altitude is displayed in white, it is not being used by the system (non-designated) to determine the vertical speed and deviation guidance.
- An altitude displayed as small text is an altitude that is published in the navigation database.
- Altitudes displayed as a light blue halftone cannot be used in the current vertical navigation calculations.

Refer to Figure 1-8 and Table 1-1 for more detail regarding the significance of text size and color.

### Table 1-1 VNAV Altitude Text Size and Color

<table>
<thead>
<tr>
<th>White Text</th>
<th>Light Blue Text</th>
<th>Light Blue Halftone Text</th>
</tr>
</thead>
<tbody>
<tr>
<td><strong>Large Text</strong></td>
<td>Altitude calculated by the system estimating the altitude of the aircraft as it passes over the navigation point. This altitude is provided as a reference and is not designated to be used in determining vertical speed and deviation guidance.</td>
<td>Altitude has been entered by the pilot. Altitude is designated for use in giving vertical speed and deviation guidance. Altitude does not match the published altitude in navigation database or no published altitude exists.</td>
</tr>
<tr>
<td><strong>Small Text</strong></td>
<td>Altitude is not designated to be used in determining vertical speed and deviation guidance. Altitude has been retrieved from the navigation database and is provided as a reference.</td>
<td>Altitude is designated for use in giving vertical speed and deviation guidance. Altitude has been retrieved from the navigation database or has been entered by the pilot and matches a published altitude in the navigation database.</td>
</tr>
</tbody>
</table>
Some altitudes retrieved from the database have associated restrictions indicating to stay ‘At’, ‘At or Above’, or ‘At or Below’ a specific altitude. These restrictions are indicated using a ‘bar’ above and/or below the appropriate altitude as shown in Figure 1-9.

![Figure 1-9 Altitude Restrictions](image)

See Section 7 - Navigation, for a sample flight plan which further illustrates vertical navigation in more detail.

### 1.6 BACKLIGHTING

**Manually adjust the backlight for the PFD and MFD:**

1) Press the **MENU** Key on the PFD to display the PFD Setup Menu window.

2) Press the small **FMS** Knob to activate the cursor. ‘PFD DSPL > AUTO’ is now highlighted.

3) Turn the small **FMS** Knob to display the selection window.

4) Turn the **FMS** Knob to select ‘MANUAL’, then press the **ENT** Key.

5) With the intensity value now highlighted, turn the small **FMS** Knob to select the desired backlighting.

6) Turn the large **FMS** Knob to highlight ‘MFD DSPL > AUTO’ and repeat steps 3 through 5.

### 1.7 XM RADIO - AUTOMATIC MUTING

XM Radio audio will be automatically muted when the aircraft groundspeed exceeds approximately 30 kts **and** the airspeed is less than approximately 80 kts. The audio must be manually unmuted once the aircraft is outside the applicable speed range. Refer to the G1000 Optional Equipment section of the pilot’s guide.
1.8 DATABASE UPDATES

The G1000 System uses Secure Digital (SD) cards to load and store various types of data. For basic flight operations, SD cards are required for database storage as well as Jeppesen aviation and ChartView database updates.

The following procedures pertain to updating the G1000 system with SD cards obtained as outlined in Appendix B of the G1000 Cessna Nav III Pilot’s Guide.

Jeppesen Aviation Database

NOTE: After the aviation database is installed, the card may be removed after loading the update to each LRU.

Updating the Jeppesen aviation database:

1) With the G1000 System OFF, insert the SD card containing the aviation database update into the top card slot of the PFD to be updated (Label of SD card facing left).

2) Turn the G1000 System ON. A prompt similar to the following is displayed in the upper left corner of the PFD:

![Figure 1-11 Database Update Prompt](image)

3) Press the ENT Key to start the database update. A prompt similar to the following is displayed:

![Figure 1-12 Database Update Confirmation](image)

4) After the update completes, the PFD starts in normal mode.

5) Turn the G1000 System OFF and remove the SD card.

6) Repeat steps 1 through 4 for the MFD. The MFD and PFD databases are now updated. Remove the SD card when finished.

7) Verify that the correct update cycle is loaded during startup of the MFD.
Garmin Databases

Since these databases are not stored internally in the MFD or PFD, a Supplemental Data Card containing identical database versions must be kept in each display unit.

---

**NOTE:** The data contained in the terrain and obstacle databases comes from government agencies. Garmin accurately processes and cross-validates the data, but cannot guarantee the accuracy and completeness of the data.

---

1) Insert one SD card in the bottom card slot of the MFD and one in the bottom card slot of the PFD. The SD card containing the ChartView or FliteCharts database must be inserted into the bottom slot on the MFD.

2) Apply power to the G1000 System. View the MFD power-up splash screen. Check that the databases are initialized and displayed on the scrolling window of the splash screen. When updating the terrain and FliteCharts databases, an ‘in progress’ message may be seen. If this message is present, wait for the system to finish loading before verifying the correct databases are initialized, then proceed to step 3.

---

3) Acknowledge the Power-up Page agreement by pressing the **ENT** Key or the right most softkey.

4) Turn the large **FMS** Knob to select the AUX Page group on the MFD.

5) Turn the small **FMS** Knob to select the System Status Page.

6) Press the **DBASE** Softkey to place the cursor in the ‘DATABASE’ box.

7) Turn either **FMS** Knob to scroll through the database list and check that all databases are current and there are no errors.

8) Power down the G1000.
SECTION 2: FLIGHT INSTRUMENTS

The following discussions pertain to the Primary Flight Display, unless otherwise indicated.

Figure 2-1 Default PFD Information

- 1. NAV Frequency Box
- 2. Airspeed Indicator
- 3. True Airspeed Box
- 4. Heading Box
- 5. Ground Track Bug
- 6. Horizontal Situation Indicator
- 7. Outside Air Temperature Box
- 8. System Time Box
- 9. Transponder Status Box
- 10. Turn Rate Indicator
- 11. Barometric Setting Box
- 12. Vertical Speed Indicator
- 13. Altimeter
- 14. Altitude Reference Box
- 15. COM Frequency Box
- 16. Navigation Status Box
- 17. Slip/Skid Indicator
- 18. Attitude Indicator
Figure 2-2 Additional PFD Information

1. Traffic Annunciation
2. Vspeed References
3. Selected Heading Box
4. Wind Data Window
5. Inset Map
6. BRG1 Information Window
7. DME Information Window
8. BRG2 Information Window
9. Flight Plan Window
10. Barometric Minimums Box
11. Altitude Reference Bug
12. Selected Course Box
13. Barometric Minimums Bug
14. Vertical Deviation/Glidepath (WAAS enabled systems only)/Glidesope Indicator
15. Marker Beacon Annunciation

Figure 2-3 PFD Navigation Status Box

Garmin G1000 Cockpit Reference Guide for the Cessna Nav III
2.1 AIRSPEED INDICATOR

![Figure 2-4 Airspeed Indicator](image)

**Speed Indication**

The indicated airspeed is displayed inside the black pointer. The pointer will become red upon reaching Vne.

![Figure 2-5 Red Pointer at Vne](image)

**Speed Ranges**

The color coded speed range strip denotes flaps operating range, normal operating range, and never exceed speed (Vne). A red range is also present for low speed awareness. Refer to the Pilot's Operating Handbook (POH) for airspeed limitations and indicator markings.

**Airspeed Trend Vector**

The end of the trend vector displays approximately what the airspeed will be in 6 seconds if the current rate of acceleration/deceleration is maintained.

2.2 ATTITUDE INDICATOR

![Figure 2-6 Attitude Indicator](image)

**Vspeed References**

Vspeed References are turned on or off in the Timer/References Window. Press the TMR/REF Softkey to display the widow. When active (ON), the Vspeeds are displayed at their respective locations to the right of the airspeed scale. To activate the Vspeed References, display the Timer/Reference Window and turn the large FMS Knob to place the cursor in the ON/OFF field. Turn the small FMS Knob to select ON or OFF.

**Speed Indication**

The indicated airspeed is displayed inside the black pointer. The pointer will become red upon reaching Vne.

**Speed Ranges**

The color coded speed range strip denotes flaps operating range, normal operating range, and never exceed speed (Vne). A red range is also present for low speed awareness. Refer to the Pilot's Operating Handbook (POH) for airspeed limitations and indicator markings.

**Airspeed Trend Vector**

The end of the trend vector displays approximately what the airspeed will be in 6 seconds if the current rate of acceleration/deceleration is maintained.
2.3 ALTIMETER

Altitude Reference Bug

The Altitude Reference Bug is displayed at the Selected Altitude or the edge of the tape (whichever is closer to the current altitude) to provide increased altitude awareness and to set the desired hold altitude for the autopilot.

**Set the Altitude Reference Bug:**

Turn the ALT Knobs to set the Altitude Reference Bug. The small ALT Knob sets the hundreds and the large ALT Knob sets the thousands. This altitude also appears in the Altitude Reference Box above the Altimeter.

Altitude Trend Vector

The end of the trend vector displays approximately what the altitude will be in six seconds if the current rate of vertical speed is maintained.

Barometric Setting Box

**Select barometric pressure:**

1) Turn the BARO Knob to select the desired setting.

2) Push the BARO Knob to select standard pressure 29.92.

Altitude Alerting

<table>
<thead>
<tr>
<th>Within 1000 ft</th>
<th>Within 200 ft</th>
<th>Deviation of ±200 ft</th>
</tr>
</thead>
<tbody>
<tr>
<td>🟢 4000</td>
<td>🟢 4000</td>
<td>🟢 4000</td>
</tr>
</tbody>
</table>

**Figure 2-8 Altitude Alerting Visual Annunciations**

Visual annunciations appear in the Altitude Reference Box. Whenever the setting is changed, the Altitude Alerter is reset. The Altitude Alerter is independent of the Automatic Flight Control System.

Metric Display

**Display altitude in meters and barometric pressure in hectopascals:**

1) Press the PFD Softkey to display the second level softkeys.

2) Press the METRIC Softkey to display altitude in meters and barometric pressure in hectopascals. Press the BACK Softkey to return to the top level softkeys.
2.4 VERTICAL DEVIATION/GLIDEPATH/GLIDESLOPE INDICATOR

The Vertical Deviation and Required Vertical Speed Indicators appear when vertical guidance is being given prior to executing an approach (see Figure 2-10). In systems that are WAAS enabled, the Glidepath Indicator appears at a point prior to the FAF when executing an LPV or LNAV+V approach (see Figure 2-11).
The Glideslope Indicator appears when an ILS approach has been activated and an ILS is tuned in the active NAV receiver field (see Figure 2-12).

**2.5 MARKER BEACON ANNUNCIATIONS**

The actual vertical speed is displayed inside the pointer.

When the Flight Director is placed in Vertical Speed Mode (by pressing the VS Key) the Vertical Speed Bug is displayed. Press the NOSE UP or NOSE DN Key to adjust.

**2.7 BAROMETRIC ALTITUDE MINIMUMS**

The desired barometric altitude minimums can be set in the Timer/References Window. The altitude ranges from 0 to 16,000 feet in 10-foot increments. The minimums are reset anytime the power is cycled.
Visual annunciations alert the pilot when approaching the MDA:

- When the aircraft altitude descends to within 2500 feet of the MDA setting, the Barometric Minimum Box appears with the altitude in light blue text. The bug appears on the tape in light blue once in range.
- When the aircraft passes through 100 feet of the MDA, the bug and text turn white.
- Once the aircraft descends past the MDA, the bug and text turn yellow and the aural alert, “Minimums Minimums”, is generated.

Alerting is inhibited while the aircraft is on the ground. If the aircraft climbs after having reached the MDA, once it reaches 50 feet above the MDA, alerting is disabled.

2.8 HORIZONTAL SITUATION INDICATOR (HSI)

The HSI compass can be displayed as a 360° rose or 140° arc by pressing the PFD Softkey, followed by the 360 HSI or the ARC HSI Softkey.

Set the barometric altitude minimums:

1) From the Timer References Window, turn the large FMS Knob to highlight the BARO MIN field.
2) Turn the small FMS Knob to select the desired barometric altitude minimums and press the ENT Key. The ON/OFF field is now highlighted.
3) Turn the small FMS Knob in the direction of the green arrowhead.
**Turn Rate Indicator and Heading Trend Vector**

Tick marks to the left and right of the lubber line denote half-standard and standard turn rates. A magenta turn rate trend vector shows the current turn rate. The end of the trend vector gives the heading predicted in six seconds, based on the present turn rate. At rates greater than 4 deg/sec, an arrowhead appears at the end of the magenta trend vector and the prediction is no longer valid.

![Figure 2-18 Turn Rate Indicator and Trend Vector](image)

**Course Pointer**

The course pointer is a single line arrow (GPS, VOR1 and LOC1) or double line arrow (VOR2 and LOC2) which points in the direction of the set course.

![Figure 2-20 Course Pointer](image)
Course Deviation Indicator (CDI)

The CDI scale automatically adjusts to the current phase of flight (enroute 2.0 nm, terminal area 1.0 nm, or approach 0.3 nm). Scaling may be selected manually from the MFD System Setup Page. See the System Setup section in the G1000 Pilot’s Guide for more details.

Bearing Pointers and Information Windows

Pressing the PFD Softkey provides access to the BRG1 and BRG2 Softkeys. The BRG1 pointer is a single line pointer. The BRG2 pointer is a double line pointer. Press the BRG1 or BRG2 Softkey to cycle through selecting NAV1/2, GPS, or ADF for display using the corresponding pointer.

DME (optional)

To display the DME Information Window, press the PFD Softkey followed by the DME Softkey.

Navigation Source

Change CDI navigation sources:

1) Press the CDI Softkey to change from GPS to VOR1/LOC1.
2) Press the CDI Softkey again to change from VOR1/LOC1 to VOR2/LOC2.
3) Press the CDI Softkey a third time to return to GPS.

When using GPS as the navigation source, the following may appear:
• LOI - GPS position integrity is inadequate for the current procedure being flown. If GPS is being used as primary navigation, and LOI is annunciated, other means of primary navigation will be required, such as VHF. LOI is also displayed during GPS position initialization.
• WARN – GPS detects a position error.
• SUSP – Displayed when in OBS Mode indicating GPS waypoint sequencing is suspended.
• DR – Navigating using Dead Reckoning due to an error in the GPS solution.

3) Press the OBS Softkey again to disable OBS mode.

NOTE: The OBS Softkey is only displayed when navigating an active leg using GPS.

2.9 WIND DATA

When the window is selected for display, but wind information is invalid or unavailable, the window shows “NO WIND DATA”. Wind data can be displayed in three different ways:
• Longitudinal and lateral components (Option 1)
• Total wind direction and speed (Option 2)
• Total direction with head and crosswind speed components (Option 3)

Displaying wind data:
1) Press the PFD Softkey.
2) Press the WIND Softkey to display wind data below the Selected Heading.
3) Press one of the OPTN softkeys to change how wind data is displayed.
4) To remove the Wind Data Window, press the OFF Softkey.
SECTION 3: ENGINE INDICATION SYSTEM (EIS)

NOTE: Refer to the Pilot’s Operating Handbook (POH) for limitations.

EIS information is presented in three displays, accessed using the ENGINE Softkey on the MFD:

- **Engine Display** – Default display, shows all critical engine, fuel, and electrical indicators
- **Lean Display** – Provides engine leaning information
- **System Display** – Shows numeric readouts of critical engine, fuel, and electrical indicators

Green and white bands indicate normal ranges of operation; yellow and red bands indicate caution and warning, respectively. If sensory data to an instrument becomes invalid or unavailable, a red “X” is shown across the instrument.

### 3.1 ENGINE DISPLAY

The Engine Display is the default EIS display and can be displayed after viewing other EIS displays by pressing the ENGINE softkey.

The EIS automatically defaults back to the Engine Display from the Lean or System Display when certain parameters are exceeded. Fluctuations in engine speed and fuel quantity above certain levels, depending on the airframe, also cause reversion back to the Engine Display.

1. **Engine Manifold Pressure Gauge (MAN IN)**  
   Models 182T, T182T, 206H, T206H – Displays engine power in inches of mercury (in Hg). Turbocharged aircraft have a red portion of the gauge indicating the maximum manifold pressure range. A white tick mark is displayed indicating the cruise manifold pressure (Model T182T only).

2. **Tachometer (RPM)** – Shows propeller speeds in revolutions per minute (rpm). Red range indicates propeller overspeed warning; a white high-rpm range indicates above normal operating speeds (Models 172S, 206H, and T206H)

3. **Fuel Flow Indicator (FFLOW GPH)** – Shows the current fuel flow in gallons per hour (gph). For turbocharged aircraft, the indicator displays a small stand-alone green band indicating maximum takeoff fuel flow. A white tick mark indicates the maximum cruise fuel flow (Model T182T only).

4. **Oil Pressure Indicator (OIL PRES)** – Displays pressure of the oil supplied to the engine in pounds per square inch (psi).

5. **Oil Temperature Indicator (OIL TEMP)** – Displays the engine oil temperature in degrees Fahrenheit (°F).

6. **Cylinder Head Temperature Indicator (CHT)**  
   Models 182T, T182T, 206H, T206H – Shows the head temperature of the hottest cylinder (number shown in triangular pointer) in degrees Fahrenheit (°F).

7. **Exhaust Gas Temperature Indicator (EGT)**  
   Normally-aspirated Aircraft – Displays the exhaust gas temperature of the hottest cylinder (number shown in triangular pointer) in degrees Fahrenheit (°F).

8. **Turbine Inlet Temperature Indicator (TIT)**  
   Turbocharged Aircraft – Displays the temperature at the turbine inlet in degrees Fahrenheit (°F).

9. **Vacuum Pressure Indicator (VAC)**  
   Models 172R and 172S – Displays vacuum pressure.
**Fuel Quantity Indicator (FUEL QTY GAL)**
- Shows the quantity of fuel in the tanks, in gallons, ranging from zero to full (F) for each fuel tank (left–L and right–R). When full, the indicator displays to 35 gallons per side (26 gallons for the Models 172R and 172S).

**Engine Hours (Tach) (ENG HRS)** Models 172R and 172S
- A numeric readout gives the time (in hours) the engine has been in service.

**Voltmeter (M, E BUS VOLTS)** – Displays the main and essential bus voltages.

**Ammeter (M, S BATT AMPS)** – Shows the main and standby battery load in amperes.

---

*Figure 3-1 Engine Display (Normally-aspirated Aircraft)*

Garmin G1000 Cockpit Reference Guide for the Cessna Nav III
SECTION 3 – ENGINE INDICATION SYSTEM (EIS)

Figure 3-2  Engine Display (Turbocharged Aircraft)
3.2 LEAN DISPLAY

NOTE: The pilot should follow the engine manufacturer’s recommended leaning procedures in the Pilot’s Operating Handbook (POH).

1 Engine Manifold Pressure Gauge (MAN IN) Models 182T, T182T, 206H, T206H – Displays engine power in inches of mercury (in Hg). Turbocharged aircraft have a red portion of the gauge indicating the maximum manifold pressure range. A white tick mark is displayed indicating the cruise manifold pressure (Model T182 only).

2 Tachometer (RPM) – Shows propeller speeds in revolutions per minute (rpm). Red range indicates propeller overspeed warning; a white high-rpm range indicates above normal operating speeds (Models 172S, 206H and T206H).

3 Fuel Flow (FFLOW GPH) – Shows the current fuel flow in gallons per hour (gph).

4 Turbine Inlet Temperature Indicator (TIT) Models T182T and T206H – Displays the temperature at the turbine inlet in degrees Fahrenheit (°F). When the ASSIST Softkey is pressed, the TIT deviation from peak (ΔPEAK) is displayed below the indicator.

5 Exhaust Gas Temperature Bar Graph (EGT °F) – Displays the exhaust gas temperature of all cylinders in degrees Fahrenheit (°F); a readout for the selected cylinder (by default, the hottest cylinder) is shown below the bar graph. The selected cylinder is indicated in light blue. Cylinders whose EGTs are in the normal range appear in white. Pressing the ASSIST Softkey causes the EGT deviation from peak (ΔPEAK) for the selected cylinder to be displayed below the indicator.

6 Cylinder Head Temperature Indicator (CHT) – Shows the head temperatures of all cylinders in degrees Fahrenheit (°F); a readout for the selected cylinder (by default, the hottest cylinder) is shown below the bar graph. The selected cylinder is indicated in light blue. Cylinders whose CHTs are in the normal range appear in white. Cylinders whose CHTs enter the caution and warning ranges appear in yellow and red, respectively.

7 Fuel Quantity Indicator (FUEL QTY GAL) – Shows the quantity of fuel in the left and right fuel tanks (left–L and right–R). When full, the indicator displays to 35 gallons per side (26 gallons for the Models 172R and 172S).
Figure 3-3  Lean Display (Normally-aspirated Aircraft)
The Lean Display is accessed by pressing the **ENGINE** Softkey followed by the **LEAN** Softkey and provides information for performing engine leanning.

From the Lean Display, the pilot can utilize the **CYL SLCT** and **ASSIST** softkeys to obtain information about specific cylinders. Pressing the **CYL SLCT** (Cylinder Select) Softkey cycles through the cylinders (i.e., changes the cylinder indicated on the bar graphs in light blue). This softkey is disabled when the **ASSIST** Softkey is pressed or when a cylinder experiences a caution or warning condition; the softkey remains disabled until the temperature returns to normal.
The **ASSIST** Softkey aids in the leaning process by identifying the peak of the first cylinder whose temperature falls. If the temperature of the peaked cylinder exceeds the peak value, the peak value is not updated. Press the **ASSIST** Softkey to stop peak monitoring.

### Normally-aspirated Aircraft

For normally-aspirated aircraft, when a cylinder peaks, its peak is represented by a hollow block on the EGT Bar Graph. The EGT readout for the peaked cylinder, indicated on the bar graph in light blue, appears directly beneath the bar graph. The system automatically switches to the first peak obtained and displays the temperature deviation from peak ($\Delta$PEAK) in degrees Fahrenheit ($^\circ$F) below the EGT readout.

### Turbocharged Aircraft

Leaning for turbocharged aircraft is done with reference to the Turbine Inlet Temperature (TIT). When the temperature peaks, the numeric readout ($\Delta$PEAK) appears below the TIT Indicator and displays the difference between peak and current TITs, in degrees Fahrenheit ($^\circ$F). If a peak is not displayed, underscores are shown until one is established.

### 3.3 SYSTEM DISPLAY

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**NOTE:** Fuel calculations do not use the aircraft fuel quantity indicators and are calculated from the last time the fuel was reset.

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**NOTE:** The pilot should refer to the Pilot’s Operating Handbook (POH) for fuel values and limitations. The displayed fuel remaining can be adjusted up to 53 gal (Models 172R, 172S) or 87 gal (Models 182T, T182T, 206H, and T206H).
1. **Engine Manifold Pressure Gauge (MAN IN)**
   *Models 182, T182, 206, T206* – Displays engine power in inches of mercury (in Hg). Turbocharged aircraft have a red portion of the gauge indicating the maximum manifold pressure range. A white tick mark is displayed indicating the cruise manifold pressure (*Model T182 only*).

2. **Tachometer (RPM)** – Shows propeller speeds in revolutions per minute (rpm). Red range indicates propeller overspeed warning; a white high-rpm range indicates above normal operating speeds (*Models 172S, 206 and T206*).

3. **Oil Pressure (OIL PSI)** – Displays pressure of the oil supplied to the engine in pounds per square inch (psi).

4. **Oil Temperature (OIL °F)** – Displays the engine oil temperature in degrees Fahrenheit (°F).

5. **Engine Hours (Tach) (ENG HRS)**
   *Models 182, T182, 206, T206* – A numeric readout gives the time (in hours) the engine has been in service.

6. **Vacuum Pressure Indicator (VAC)**
   *Models 182, T182, 206, T206* – Displays vacuum pressure.

7. **Fuel Flow (FFLOW GPH)** – Shows the current fuel flow in gallons per hour (gph).

8. **Calculated Fuel Used (GAL USED)** – Shows quantity of fuel used in gallons based on fuel flow since last reset.

9. **Set Fuel Remaining (GAL REM)** – Shows current fuel remaining in gallons as set by the pilot and adjusted for fuel burn since last set.

10. **Fuel Quantity Indicator (FUEL QTY GAL)** – Shows the quantity of fuel in the tanks, in gallons, ranging from zero to full (F) for each fuel tank (left–L and right–R). When full, the indicator displays to 35 gallons per side (26 gallons for the *Models 172R and 172S*).


12. **Ammeter (M, S BATT AMPS)** – Shows the main and standby battery load in amperes.
SECTION 3 – ENGINE INDICATION SYSTEM (EIS)

Figure 3-5  System Display (Normally-aspirated Aircraft)
Figure 3-6 System Display (Turbocharged Aircraft)
SECTION 4: NAV/COM AND TRANSPONDER

The NAV/COM controls and frequency boxes share the same locations on the Primary Flight Display and the Multi-Function Display.

Figure 4-1  G1000 VHF NAV/COM Interface (PFD shown)
SECTION 4 – NAV/COM & TRANSPONDER

SECTION 4 – NAV/COM & TRANSPONDER

Figure 4-2 Frequency Fields

Figure 4-3 Frequency Transfer Arrow and Tuning Box

Figure 4-4 NAV/COM Controls

- Turn to tune in desired frequencies.
- Press to change tuning box positions.
4.1 RADIO STATUS INDICATIONS

- **TX** – When a COM radio is transmitting, a white TX indication appears to the right of the corresponding COM frequency.

- **ID** – When the Morse code identifier is ON for a NAV radio, a white ID indication appears to the left of the corresponding active NAV frequency. The Morse code identifier can be heard if the corresponding NAV radio is selected on the audio panel.

![Figure 4-5 Radio Status Indications](image)

4.2 VOLUME

‘VOLUME’ is displayed in place of the associated radio name (i.e., ‘COM1’ or ‘NAV2’) for two seconds after the volume level is last changed. The percentage of maximum volume is displayed in place of the standby frequency selected by the tuning box.

![Figure 4-6 COM Volume Level](image)

4.3 AUTOMATIC SQUELCH

Automatic squelch can be disabled for a COM radio by pressing the **COM** Knob to place the tuning box on the desired COM's standby frequency, then by pressing the **VOL/PUSH SQ** Knob.

4.4 QUICKLY ACTIVATING 121.500 MHZ

Pressing and holding the **COM Frequency Transfer** Key for approximately two (2) seconds automatically tunes the selected COM radio to the emergency frequency.

4.5 OPTIONAL NAV RADIOS

**DME Radio (optional)**

The DME Tuning Window is displayed by pressing the DME Softkey.

![Figure 4-7 DME Selection Window](image)

**Change the DME tuning source:**

1) From the tuning window, turn the large **FMS** Knob to highlight the DME source field.

2) Turn the small **FMS** Knob to display the selection window. Turn the **FMS** Knob to select the desired mode and press the **ENT** Key.

**ADF Radio (optional)**

The G1000 does not support tuning of the ADF in Nav III aircraft. ADF tuning is accomplished through the Bendix/King KR 87 ADF Radio. ADF volume must also be adjusted through the KR 87.
4.6 FREQUENCY AUTO-TUNING

Auto-tuning on the PFD

Figure 4-9 Nearest Airports Window (PFD)

1) Press the NRST Softkey to display the Nearest Airports Window.
2) Turn either FMS Knob to highlight the desired frequency.
3) Press the ENT Key to place the frequency in the standby field of the active COM.
4) Press the Frequency Transfer Key to place the frequency in the active field.

NAV frequencies are entered automatically in the NAV frequency active or standby field (depending in CDI selection) upon approach loading or approach activation.

Auto-tuning on the MFD

Auto-tuning on the MFD is done in much the same way as on the PFD. Use the FMS Knobs to select the desired frequency on any of the information pages. Pressing the ENT Key then loads the selected frequency in the tuning box as a standby frequency.

4.7 TRANSPONDER

Mode Selection

The STBY, ON and ALT Softkeys can be accessed by pressing the XPDR Softkey.

Ground Mode (Automatic)

GND is displayed when the aircraft is on the ground.

Figure 4-10 Ground Mode

Reply Status

When the transponder sends replies to interrogations, an “R” indication appears momentarily in the reply status field.

Figure 4-11 Reply Indication

Code Selection

1) Press the XPDR Softkey to display the transponder Mode Selection softkeys.
2) Press the CODE Softkey to display the transponder Code Selection softkeys, which includes the digit softkeys.
3) Press the appropriate digit softkeys to enter the code in the four-digit code field of the Transponder Status bar. Five seconds after the fourth digit has been entered, the transponder code becomes activated.

When entering a code, press the BKSP Softkey as needed to back up and change code digits.
Flight ID Reporting

If so configured, the Flight ID may be entered in the Timer/Reference Window.

1) Press the TMR/REF Softkey to display the Timer/References Window.
2) Turn the large FMS Knob to place the cursor in the Flight ID field.
3) Turn the small FMS Knob to enter the desired first character.
4) Turn the large FMS knob to place the cursor in the next desired field.
5) Turn the small FMS Knob to enter the next desired character.
6) Repeat steps 4 and 5 until the desired Flight ID is entered.
7) Press the ENT Key. ‘updating’ will be displayed as the Flight ID is loaded into the system.
SECTION 5: AUDIO PANEL

Figure 5-1 Front Panel Controls
5.1 COM RADIO SELECTION

Pressing the COM1 MIC or COM2 MIC Key selects the active transmitter (i.e., microphone). The associated receiver audio (COM1 or COM2) also becomes selected when the COM MIC Key is pressed.

To prevent deselecting the desired received audio when pressing another COM MIC Key, press the already selected COM1 or COM2 Key before pressing the other COM MIC Key.

Figure 5-2 Transceivers

5.2 MARKER BEACON RECEIVER

The marker beacon receiver is always on. Only the marker beacon audio can be turned off. Figure 5-3 shows the marker beacon annunciators on the PFD.

When the MKR/MUTE Key is pressed, the key annunciator is lit and the audio tone can be heard over the speaker or headsets during marker beacon reception.

When the tone is active, pressing the MKR/MUTE Key once mutes the audio but does not affect the marker annunciator. The audio returns when the next marker signal is received.

To turn off the marker beacon audio, press the MKR/MUTE Key once when there is no marker indication present, or press twice when an indication is present. The key annunciator will be extinguished when the marker beacon audio is turned off.

Figure 5-3 Marker Beacon Annunciators on the PFD

Marker Beacon Signal Sensitivity

The HI SENS Key can be pressed for increased marker beacon signal sensitivity.

Figure 5-4 Marker Beacon
### 5.3 NAV RADIO AUDIO SELECTION

Pressing **DME**, **ADF**, **NAV1**, or **NAV2** selects and deselects the audio source and activates the annunciator. Selected audio can be heard over the headset and the speakers. These four keys can be selected individually or together.

![Navigation Radios](image)

**Figure 5-5 Navigation Radios**

### Table 5-1 ICS Isolation Modes

<table>
<thead>
<tr>
<th>Mode</th>
<th>PILOT KEY ANNUNCIATOR</th>
<th>COPLT KEY ANNUNCIATOR</th>
<th>Pilot Hears</th>
<th>Copilot Hears</th>
<th>Passenger Hears</th>
</tr>
</thead>
<tbody>
<tr>
<td>ALL</td>
<td>OFF</td>
<td>OFF</td>
<td>Selected radios; pilot; copilot; passengers; MUSIC 1</td>
<td>Selected radios; pilot; copilot; passengers; MUSIC 1</td>
<td>Selected radios; pilot; copilot; passengers; MUSIC 2</td>
</tr>
<tr>
<td>PILOT</td>
<td>ON</td>
<td>OFF</td>
<td>Selected radios; pilot</td>
<td>Copilot; passengers; MUSIC 1</td>
<td>Copilot; passengers; MUSIC 2</td>
</tr>
<tr>
<td>COPILOT</td>
<td>OFF</td>
<td>ON</td>
<td>Selected radios; pilot; passengers; MUSIC 1</td>
<td>Copilot</td>
<td>Selected radios; pilot; passengers; MUSIC 2</td>
</tr>
<tr>
<td>CREW</td>
<td>ON</td>
<td>ON</td>
<td>Selected radios; pilot; copilot</td>
<td>Selected radios; pilot; copilot</td>
<td>Passengers; MUSIC 2</td>
</tr>
</tbody>
</table>

### 5.4 INTERCOM SYSTEM (ICS) ISOLATION

Press the **PILOT** and/or **COPLT** Key to select who is isolated from hearing the Nav/Com radios and music. Selection scenarios are addressed in Table 5-1.

![ICS Isolation](image)

**Figure 5-6 ICS Isolation**

Table 5-1 ICS Isolation Modes
5.5 INTERCOM SQUELCH CONTROL

Select manual squelch for intercom audio by pressing the MAN SQ Key to light the annunciator.

Pressing the small VOL/SQ Knob now switches between volume and squelch adjustment by lighting VOL or SQ respectively.

![Figure 5-7 Volume/Squelch Control](image)

5.6 DIGITAL CLEARANCE RECORDER AND PLAYER

Each reception of primary active COM audio is automatically recorded in a memory block. When the next transmission is received, it is recorded in the next memory block, and so on. Once the 2.5 minutes of recording time has been reached, the recorder begins recording over the stored memory blocks, starting from the oldest block. Powering off the unit automatically clears all recorded blocks.

![Figure 5-8 Clearance Recorder Play Key](image)

- Pressing MKR/MUTE while playing a memory block stops play.
- Pressing the PLAY Key during play begins playing the previously recorded memory block. Each subsequent press of the PLAY Key will begin playing the next previously recorded block.

If a COM input signal is detected while playing, play is halted and the new COM input signal is recorded as the latest block.

- Pressing PLAY once plays the latest recorded memory block, then returns to normal operation.
SECTION 6: AUTOMATIC FLIGHT CONTROL

**NOTE:** The Aircraft Flight Manual (AFM) always supersedes the information in this guide. This section only applies to the GFC 700 Automatic Flight Control System (AFCS).

**NOTE:** The GFC 700 AFCS is not available for the Cessna 172.

6.1 AFCS CONTROLS

The following dedicated AFCS keys are located on the bezels of the PFD and MFD:

![Dedicated AFCS Controls](image)

Figure 6-1  Dedicated AFCS Controls

The following AFCS controls are located in the cockpit separately from the MFD:

- **AP DISC Switch (Autopilot Disconnect)**
  Disengages the autopilot and interrupts pitch trim operation. The red AP DISC switch is located forward of the MET Switch on the pilot’s control wheel left grip. This switch may be used to acknowledge an autopilot disconnect and mute the associated aural tone.

- **CWS Button (Control Wheel Steering)**
  Momentarily disengages the autopilot and synchronizes the flight director's Command Bars with the current aircraft pitch (if not in Glideslope Mode) and roll (if in Roll Hold Mode). The CWS button is located on the top of the pilot’s control wheel right grip. Upon release of the CWS button, the flight director may establish new reference points, depending on the current pitch and roll modes.

- **GA Switch (Go-Around)**
  Disengages the autopilot, selects flight director Go-Around Mode, and activates the missed approach. The GA Switch is located on the instrument panel above the throttle.

- **MET Switch (Manual Electric Trim)**
  The MET Switch is located on the pilot’s control wheel left grip. This composite switch is split into left and right sides. The left switch is the ARM contact and the right switch controls the DN (forward) and UP (rearward) contacts. The MET ARM switch can be used to disengage the autopilot and to acknowledge an autopilot disconnect alert and mute the associated aural tone. Manual trim commands are generated only when both sides of the switch are operated simultaneously. If either side of the switch is active separately for more than three seconds, MET function is disabled and ‘PTRM’ is displayed as the AFCS Status Annunciation on the PFD. The function remains disabled until both sides of the switch are inactivated.
6.2 FLIGHT DIRECTOR OPERATION

With the flight director activated, the aircraft can be hand-flown to follow the path shown by the Command Bars. Maximum commanded pitch (+20°/-15°) and bank (22°) angles, vertical acceleration, and roll rate are limited to values established during AFCS certification. The flight director also provides commands to the autopilot.

Activating the Flight Director

Pressing the FD or AP Key (when the flight director is not active) activates the flight director in default pitch/roll modes. Pushing the GA Switch or any flight director mode key activates the flight director in the respective mode(s). The flight director may be turned off by pressing the FD Key.

Command Bars

Upon activation of the flight director, Command Bars are displayed on the PFD as a single cue. If the attitude information sent to the flight director becomes invalid or unavailable, the Command Bars are removed from the display. The Command Bars do not override the aircraft symbol.
AFCS Status Box

Flight director roll modes are shown on the left and pitch on the right. Armed modes are annunciated in white and active in green. Autopilot status is displayed in the center of the AFCS Status Box.

6.3 FLIGHT DIRECTOR MODES

Flight director modes are normally selected independently for the pitch and roll axes. Unless otherwise specified, all mode keys are alternate action (i.e., press on, press off). In the absence of specific mode selection, the flight director reverts to the default pitch and/or roll mode(s).

Armed modes are annunciated in white and active in green in the AFCS Status Box. Under normal operation, when the control for the active flight director mode is pressed, the flight director reverts to the default mode(s) for the axis(es). Automatic transition from armed to active mode is indicated by the white armed mode annunciation moving to the green active mode field and flashing for ten seconds.

A flashing yellow mode annunciation and annunciator light indicate loss of sensor (AHRS, ADC, IAU) or navigation data (VOR, LOC, GPS, VNAV, WAAS) required to compute commands. When such a loss occurs, the system automatically begins to roll the wings level or maintain the pitch angle, depending on the affected axis. The flashing annunciation stops when the affected mode key is pressed or another mode for the axis is selected. If after ten seconds no action is taken, the flashing annunciation stops and the flight director enters the default mode for the affected axis.

If the information required to compute a flight director mode becomes invalid or unavailable, the flight director automatically reverts to the default mode for that axis. The flight director is automatically disabled if the attitude information required to compute the default flight director modes becomes invalid or unavailable.

Pitch Modes

- **Pitch Hold** (default mode)—Holds the current aircraft pitch attitude; may be used to climb/descend to the Selected Altitude
- **Selected Altitude Capture**—Captures the Selected Altitude
- **Altitude Hold**—Holds the current Altitude Reference
- **Vertical Speed**—Maintains the current aircraft vertical speed; may be used to climb/descend to the Selected Altitude
- **Flight Level Change**—Maintains the current aircraft airspeed while the aircraft is climbing/descending to the Selected Altitude
- **Vertical Path Tracking**—Follows an active vertical profile for enroute and terminal phases of flight
- **VNAV Target Altitude Capture**—Captures the VNAV Target Altitude
- **Glidepath**—Intercepts and tracks the WAAS glidepath on approach (only available in installations with GIA 63W Integrated Avionics Units and when WAAS is available)
- **Glideslope**—Intercepts and tracks the ILS glideslope on approach
- **Go Around**—Automatically disengages the autopilot and commands a constant pitch angle and wings level while in the air
Table 6-1 lists the pitch modes with their corresponding controls and annunciations. The mode reference (shown with default measurement units) is displayed next to the active mode annunciation for Altitude Hold, Vertical Speed, and Flight Level Change modes. The NOSE UP/NOSE DN Keys can be used to change the pitch mode reference while operating under Pitch Hold, Vertical Speed, or Flight Level Change Mode.

<table>
<thead>
<tr>
<th>Pitch Mode</th>
<th>Control</th>
<th>Annunciation</th>
<th>Reference Range</th>
<th>Reference Change Increment</th>
</tr>
</thead>
<tbody>
<tr>
<td>Pitch Hold</td>
<td>(default)</td>
<td>PIT</td>
<td>-20° to +15°</td>
<td>0.5°</td>
</tr>
<tr>
<td>Selected Altitude Capture</td>
<td>*</td>
<td>ALTS</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Altitude Hold</td>
<td>ALT Key</td>
<td>ALT nnnn FT</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Vertical Speed</td>
<td>VS Key</td>
<td>VS nnnn FPM</td>
<td>-3000 to +1500 fpm</td>
<td>100 fpm</td>
</tr>
<tr>
<td>Flight Level Change, IAS Hold</td>
<td>FLC Key</td>
<td>FLC nnn KT</td>
<td>70 to 165 kt</td>
<td>1 kt</td>
</tr>
<tr>
<td>Vertical Path Tracking</td>
<td>VNV Key</td>
<td>VPTH</td>
<td></td>
<td></td>
</tr>
<tr>
<td>VNAV Target Altitude Capture</td>
<td>**</td>
<td>ALTV</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Glidepath</td>
<td>APR Key</td>
<td>GP</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Glideslope</td>
<td>APR Key</td>
<td>GS</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Go Around (in air)</td>
<td>GA Switch</td>
<td>GA</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

* ALTS is armed automatically when PIT, VS, FLC, or GA is active, and under VPTH when the Selected Altitude is to be captured instead of the VNAV Target Altitude.

** ALTV is armed automatically under VPTH when the VNAV Target Altitude is to be captured instead of the Selected Altitude.

Table 6-1  Flight Director Pitch Modes
**Pitch Hold Mode (PIT)**

When the flight director is activated (the FD Key is pressed), Pitch Hold Mode is selected by default. Pitch Hold Mode is indicated as the active pitch mode by the green annunciation 'PIT'. This mode may be used for climb or descent to the Selected Altitude (shown above the Altimeter), since Selected Altitude Capture Mode is automatically armed when the mode is activated.

In Pitch Hold Mode, the flight director maintains a constant pitch attitude, the pitch reference. The pitch reference is set to the aircraft attitude at the moment of mode selection. If the aircraft pitch attitude exceeds the flight director pitch command limitations, the flight director commands a pitch angle equal to the nose-up/down limit.

**Changing the Pitch Reference**

When operating in Pitch Hold Mode, the pitch reference can be adjusted by:
- Using the NOSE UP/NOSE DN Keys
- Pressing the CWS Button, hand-flying the aircraft to establish a new pitch reference, then releasing the CWS Button

---

Figure 6-5 Pitch Hold Mode
SECTION 6 – AUTOMATIC FLIGHT CONTROL

Selected Altitude Capture Mode (ALTS)

Selected Altitude Capture Mode arms automatically when the flight director is in Pitch Hold, Vertical Speed, Flight Level Change, or Go Around Mode. This mode is also armed automatically under Vertical Path Tracking Mode when the Selected Altitude is to be captured instead of the VNAV Target Altitude. The white ‘ALTS’ annunciation indicates Selected Altitude Capture Mode is armed (see Figure 6-5 for example).

The ALT Knob is used to set the Selected Altitude, shown above the Altimeter until Selected Altitude Capture Mode becomes active.

As the aircraft nears the Selected Altitude, the flight director automatically transitions to Selected Altitude Capture Mode with Altitude Hold Mode armed (Figure 6-7). This automatic transition is indicated by the green ‘ALTS’ annunciation flashing for up to ten seconds and the appearance of the white ‘ALT’ annunciation. The Selected Altitude is shown as the Altitude Reference beside the ‘ALTS’ annunciation.

At 50 ft from the Selected Altitude, the flight director automatically transitions from Selected Altitude Capture to Altitude Hold Mode and holds the Selected Altitude (shown as the Altitude Reference). As Altitude Hold Mode becomes active, the white ‘ALT’ annunciation moves to the active pitch mode field and flashes green for ten seconds to indicate the automatic transition.

Changing the Selected Altitude

NOTE: Pressing the CWS Button while in Selected Altitude Capture Mode does not cancel the mode.

Use of the ALT Knob to change the Selected Altitude while Selected Altitude Capture Mode is active causes the flight director to revert to Pitch Hold Mode with Selected Altitude Capture Mode armed for the new Selected Altitude.

Altitude Hold Mode (ALT)

Altitude Hold Mode can be activated by pressing the ALT Key; the flight director maintains the current aircraft altitude (to the nearest ten feet) as the Altitude Reference. The flight director’s Altitude Reference is shown in the AFCS Status Box and is independent of the Selected Altitude, displayed above the Altimeter. Altitude Hold Mode active is indicated by a green ‘ALT’ annunciation in the AFCS Status Box.

Altitude Hold Mode is automatically armed when the flight director is in Selected Altitude Capture Mode. Selected Altitude Capture Mode automatically transitions to Altitude Hold Mode when the altitude error is less than 50 ft. In this case, the Selected Altitude becomes the flight director’s Altitude Reference.

Changing the Altitude Reference

NOTE: Turning the ALT Knob while in Altitude Hold Mode changes the Selected Altitude, but not the flight director’s Altitude Reference and does not cancel the mode.

With the CWS Button depressed, the aircraft can be hand-flown to a new Altitude Reference. When the CWS Button is released at the desired altitude, the new altitude is established as the Altitude Reference.
If the Selected Altitude is reached during CWS maneuvering, the Altitude Reference is not changed. To adjust the Altitude Reference in this case, the CWS Button must be pressed again after the Selected Altitude is reached.

**Vertical Speed Mode (VS)**

In Vertical Speed Mode, the flight director acquires and maintains a Vertical Speed Reference. Current aircraft vertical speed (to the nearest 100 fpm) becomes the Vertical Speed Reference at the moment of Vertical Speed Mode activation. Vertical Speed Mode does not consider the relative position of the Selected Altitude in relation to the current aircraft altitude at the time of mode activation, so it is possible to use Vertical Speed Mode while not climbing/descending to the Selected Altitude.

Vertical Speed Mode is activated by pressing the VS Key; the ‘VS’ annunciation appears in the AFCS Status Box to indicate the active pitch mode, along with the Vertical Speed Reference to the right. The Vertical Speed Reference is also displayed above the Vertical Speed Indicator. A Vertical Speed Reference Bug corresponding to the Vertical Speed Reference is shown on the indicator.
Changing the Vertical Speed Reference

The Vertical Speed Reference (shown both in the AFCS Status Box and above/below the Vertical Speed Indicator) may be changed by:

- Using the NOSE UP/NOSE DN Keys
- By pressing the CWS Button, hand-flying the aircraft to attain a new Vertical Speed Reference, then releasing the CWS Button

Flight Level Change Mode (FLC)

**NOTE:** The Selected Altitude should be set before selecting Flight Level Change Mode.

Flight Level Change Mode is selected by pressing the FLC Key. When Flight Level Change Mode is active, the flight director continuously monitors Selected Altitude, airspeed, and altitude. This mode acquires and maintains the Airspeed Reference while climbing or descending to the Selected Altitude (shown above the Altimeter). The Airspeed Reference is set to the current airspeed upon mode activation. Flight Level Change Mode is indicated by an ‘FLC’ annunciation beside the Airspeed Reference in the AFCS Status Box. The Airspeed Reference is also displayed directly above the Airspeed Indicator, along with a bug corresponding to the Airspeed Reference along the tape.

Engine power must be adjusted to allow the autopilot to fly the aircraft at a pitch attitude corresponding to the Airspeed Reference and the desired flight profile (climb or
descent). The flight director maintains the current altitude until either engine power or the Airspeed Reference are adjusted and does not allow the aircraft to climb or descend away from the Selected Altitude.

**Changing the Airspeed Reference**

The Airspeed Reference (shown in both the AFCS Status Box and above the Airspeed Indicator) may be adjusted:
- Using the **NOSE UP/NOSE DN** Keys
- By pressing the **CWS** Button, hand-flying the aircraft to a new airspeed, then releasing the **CWS** Button to establish the new Airspeed Reference

![Figure 6-9 Flight Level Change Mode](image)
Vertical Navigation Modes (VPTH, ALTV)

**NOTE:** Pressing the CWS Button while Vertical Path Tracking Mode is active does not cancel the mode. The autopilot guides the aircraft back to the descent path upon release of the CWS Button.

**NOTE:** VNAV flight director pitch modes are available only in conjunction with GPS roll modes.

**NOTE:** The Selected Altitude takes precedence over any other vertical constraints.

Vertical Navigation (VNAV) flight control is available for enroute/terminal cruise and descent operations when VNAV has been enabled and a VNAV flight plan (with at least one vertical waypoint) or vertical direct-to has been activated. Refer to the GPS Navigation Section for more information on VNAV flight plans. The flight director may be armed for VNAV at any time, but no target altitudes are captured during a climb.

The Command Bars provide vertical profile guidance based on specified altitudes (entered manually or loaded from the database) at waypoints in the active flight plan or vertical direct-to. The appropriate VNAV flight control modes are sequenced by the flight director to follow the path defined by the vertical profile. Upon reaching the last waypoint in the VNAV flight plan, the flight director transitions to Altitude Hold Mode and cancels any armed VNAV modes.

**Vertical Path Tracking Mode (VPTH)**

**NOTE:** If another pitch mode key is pressed while Vertical Path Tracking Mode is selected, Vertical Path Tracking Mode reverts to armed.

When a vertical profile (VNAV flight plan) is active and the VNV Key is pressed, Vertical Path Tracking Mode is armed in preparation for descent path capture. ‘VPTH’ (or ‘/V’ when Glidepath or Glideslope Mode is concurrently armed) is annunciated in white in addition to previously armed modes. If applicable, the appropriate altitude capture mode is armed for capture of the next VNAV Target Altitude (ALTV) or the Selected Altitude (ALTS), whichever is greater.

![Figure 6-10 Vertical Path Tracking Armed Annunciations](image)

Prior to descent path interception, the Selected Altitude must be set below the current aircraft altitude by at least 75 ft. For the flight director to transition from Altitude Hold to Vertical Path Tracking Mode, acknowledgment is required within five minutes of descent path capture by:

- Pressing the VNV Key
- Adjusting the Selected Altitude

If acknowledgment is not received within one minute of descent path interception, the white ‘VPTH’ annunciation and the VNV Key annunciator light start to flash. Flashing continues until acknowledged or the descent path is intercepted. If the descent is not confirmed by the time of interception, Vertical Path Tracking Mode remains armed and the descent is not captured.
In conjunction with the “TOD [top of descent] within 1 minute” annunciation in the Navigation Data Box, VNAV indications (VNAV Target Altitude, vertical deviation, and vertical speed required) appear on the PFDs in magenta (Figure 6-11).
When a descent leg is captured (Figure 6-12), Vertical Path Tracking becomes active and tracks the descent profile. An altitude capture mode (‘ALTS’ or ‘ALTV’) is armed as appropriate.

**Figure 6-12 Vertical Path Tracking Mode**

- **Vertical Path Tracking Active**
- **VNAV Target Altitude Capture Armed**
- **VNAV Target Altitude**
- **Vertical Deviation Indicator**
- **Required Vertical Speed Bug**
- **HSI Set to GPS**
- **Terminal Phase of Flight**

### Automatic Pitch Hold Reversion

Several situations can occur while Vertical Path Tracking Mode is active which cause the flight director to revert to Pitch Hold Mode. Vertical Path Tracking and the appropriate altitude capture modes are armed for possible descent profile recapture if the vertical deviation:

- Exceeds 200 ft during an overspeed condition
- Experiences a discontinuity exceeding 200 ft due to a flight plan change
- Becomes invalid due to excessive cross-track error, track angle error
- Cannot be computed for a leg type (such as a hold or procedure turn)

The following circumstances cause mode reversion without arming Vertical Path Tracking Mode:

- Navigation source manually changed from GPS
- **CNCL VNV** Softkey selected on the Active Flight Plan Page (MFD)
- All remaining vertical waypoints deleted from the flight plan
- Displays entering Reversionary Mode

### Non-Path Descents

Pitch Hold, Vertical Speed, and Flight Level Change modes can also be used to fly non-path descents while
VNAV flight control is selected. If the VS or FLC Key is pressed while Vertical Path Tracking Mode is selected, Vertical Path Tracking Mode reverts to armed along with the appropriate altitude capture mode to allow profile re-capture.

**Figure 6-13 Flight Level Change VNAV Non-Path Descent**

To prevent immediate profile re-capture, the following must be satisfied:

- At least ten seconds have passed since the non-path transition was initiated
- Vertical deviation from the profile has exceeded 250 ft, but is now less than 200 ft

Pressing the VNV Key twice re-arms Vertical Path Tracking for immediate profile re-capture.

**VNAV Target Altitude Capture Mode (ALTV)**

**NOTE:** Armed VNAV Target Altitude and Selected Altitude capture modes are mutually exclusive. However, Selected Altitude Capture Mode is armed implicitly (not annunciated) whenever VNAV Target Altitude Capture Mode is armed. This ensures the Selected Altitude is not violated during a change from VNAV Target Altitude Capture to Selected Altitude Capture Mode close to Selected Altitude interception.

VNAV Target Altitude Capture is analogous to Selected Altitude Capture Mode and is armed automatically after the VNV Key is pressed and the next VNAV Target Altitude is to be intercepted before the Selected Altitude. The annunciation ‘ALTV’ indicates that the VNAV Target Altitude is to be captured. VNAV Target Altitudes are shown in the active flight plan or vertical direct-to, and can be entered manually or loaded from a database (see the GPS Navigation Section for details). At the same time as “TOD within 1 minute” is annunciated in the Navigation Data Box, the VNAV Target Altitude is displayed above the Vertical Speed Indicator (see Figure 6-12). VNAV Target Altitudes can be modified until VNAV Target Altitude Capture Mode becomes active.

As the aircraft nears the VNAV Target Altitude, the flight director automatically transitions to VNAV Target Altitude Capture Mode with Altitude Hold Mode armed. This automatic transition is indicated by the green ‘ALTV’ annunciation flashing for up to ten seconds and the appearance of the white ‘ALT’ annunciation. The VNAV Target Altitude is shown as the Altitude Reference beside the ‘ALT’ annunciation.

At 50 ft from the VNAV Target Altitude, the flight director automatically transitions from VNAV Target Altitude Capture to Altitude Hold Mode and tracks the level leg. As Altitude Hold Mode becomes active, the white ‘ALT’ annunciation moves to the active pitch mode field and flashes green for ten seconds to indicate the automatic transition. The flight director automatically arms Vertical Path Tracking, allowing upcoming descent legs to be captured and subsequently tracked.

**Changing the VNAV Target Altitude**

**NOTE:** Pressing the CWS Button while in VNAV Target Altitude Capture Mode does not cancel the mode.
Changing the current VNAV Target Altitude while VNAV Target Altitude Capture Mode is active causes the flight director to revert to Pitch Hold Mode. Vertical Path Tracking and the appropriate altitude capture mode are armed in preparation to capture the new VNAV Target Altitude or the Selected Altitude, depending on which altitude is to be intercepted first.

VNAV target altitudes can be changed while editing the active flight plan (see the GPS Navigation Section for details).

**Glidepath Mode (GP)**

**NOTE:** Pressing the CWS Button while Glidepath Mode is active does not cancel the mode. The autopilot guides the aircraft back to the glidepath upon release of the CWS Button.

Glidepath mode is used to track the WAAS-based glidepath. Arming Glidepath Mode (annunciated in white as ‘GP’) requires:
- Approach supporting WAAS vertical guidance is loaded into the flight plan
- Expected availability of vertical guidance
- GPS Approach Mode is armed, after acquiring clearance for approach, prior to intercepting the WAAS glidepath (GPS is the selected navigation source and the APR Key is pressed; see GPS Approach Mode)

---

**Figure 6-15 Glidepath Mode**
Glideslope Mode (GS)

**NOTE:** Pressing the CWS Button while Glideslope Mode is active does not cancel the mode. The autopilot guides the aircraft back to the glideslope upon release of the CWS Button.

Glideslope Mode is available for LOC/ILS approaches to capture and track the glideslope. Glideslope Mode is armed when:
- A valid localizer frequency is tuned
- LOC Approach Mode is armed (the APR Key is pressed and either LOC is the selected navigation source or a LOC/ILS approach is loaded into the flight plan; see LOC Approach Mode)

Once the localizer has been set as the navigation source, the localizer and glideslope can be captured. Upon reaching the glideslope, the flight director transitions to Glideslope Mode and begins to intercept and track the glideslope.
Go Around (GA) Mode

Pushing the GA Switch engages the flight director in a wings level, 7° pitch-up attitude, allowing the execution of a missed approach or a go around. This mode is a couple pitch and roll mode and is annunciated as ‘GA’ in both the pitch and roll active mode fields. Go Around Mode disengages the autopilot and arms Altitude Hold Mode automatically. Subsequent autopilot engagement is allowed. Attempts to modify the aircraft attitude (i.e., with the CWS Button or NOSE UP/NOSE DN keys) result in reversion to Pitch and Roll Hold modes.

![Go Around Mode](image)

**Figure 6-19 Go Around Mode**
Roll Modes

- **Roll Hold** (default mode) — Holds the current aircraft roll attitude or rolls the wings level, depending on the commanded bank angle
- **Heading Select** — Captures and tracks the Selected Heading
- **Navigation** (GPS, VOR, LOC) — Captures and tracks the selected navigation source
- **Backcourse** — Captures and tracks a localizer signal for backcourse approaches
- **Approach** (GPSa, VAPP, LOC) — Captures and tracks the selected navigation source with greater sensitivity for approach
- **Go Around** — Commands a constant pitch angle and wings level while in the air

The following table relates each roll mode to its respective control and annunciation. Refer to the pitch modes section for information regarding Go Around and Takeoff Modes.

<table>
<thead>
<tr>
<th>Roll Mode</th>
<th>Control</th>
<th>Annunciation</th>
</tr>
</thead>
<tbody>
<tr>
<td>Roll Hold</td>
<td>(default)</td>
<td>ROL</td>
</tr>
<tr>
<td>Heading Select</td>
<td>HDG Key</td>
<td>HDG</td>
</tr>
<tr>
<td>Navigation, GPS Arm/Capture/Track</td>
<td>NAV Key</td>
<td>GPS</td>
</tr>
<tr>
<td>Navigation, VOR Enroute Arm/Capture/Track</td>
<td>NAV Key</td>
<td>VOR</td>
</tr>
<tr>
<td>Navigation, LOC Arm/Capture/Track (No Glideslope)</td>
<td>NAV Key</td>
<td>LOC</td>
</tr>
<tr>
<td>Backcourse Arm/Capture/Track</td>
<td>BC Key</td>
<td>BC</td>
</tr>
<tr>
<td>Approach, GPS Arm/Capture/Track</td>
<td>APR Key</td>
<td>GPSa</td>
</tr>
<tr>
<td>Approach, VOR Arm/Capture/Track</td>
<td>APR Key</td>
<td>VAPP</td>
</tr>
<tr>
<td>Approach, ILS Arm/Capture/Track</td>
<td>APR Key</td>
<td>LOC</td>
</tr>
<tr>
<td>(Glideslope Mode automatically armed)</td>
<td>APR Key</td>
<td>LOC</td>
</tr>
<tr>
<td>Go Around (in air)</td>
<td>GA Switch</td>
<td>GA</td>
</tr>
</tbody>
</table>

Table 6-2 Roll Modes

The **CWS** Button does not change lateral references for Heading Select, Navigation, Backcourse, or Approach modes. The autopilot guides the aircraft back to the Selected Heading/Course upon release of the **CWS** Button.

**Roll Hold Mode (ROL)**

**NOTE:** *If Roll Hold Mode is activated as a result of a mode reversion, the flight director rolls the wings level.*

When the flight director is activated (the **FD** or **AP** Key is pressed), Roll Hold Mode is selected by default. This mode is annunciated as ‘ROL’ in the AFCS Status Box. The current aircraft bank angle is held, subject to the bank angle conditions listed in Table 6-3.
SECTION 6 – AUTOMATIC FLIGHT CONTROL

Bank Angle | Flight Director Response
---|---
< 6° | Rolls wings level
6° to 22° | Maintains current aircraft roll attitude
> 22° | Limits bank to 22°

Table 6-3 Roll Hold Mode Responses

Changing the Roll Reference

The roll reference can be changed by pressing the CWS Button, establishing the desired bank angle, then releasing the CWS Button.

Heading Select Mode (HDG)

Heading Select Mode is activated by pressing HDG Key. Heading Select Mode acquires and maintains the Selected Heading. The Selected Heading is shown by a light blue bug on the HSI and in the box to the upper left of the HSI.

Changing the Selected Heading

NOTE: Pressing the HDG Knob synchronizes the Selected Heading to the current heading.

The Selected Heading is adjusted using the HDG Knob on either display. Pressing the CWS Button and hand-flying the aircraft does not change the Selected Heading. The autopilot guides the aircraft back to the Selected Heading upon release of the CWS Button.
Turns are commanded in the same direction as Selected Heading Bug movement, even if the bug is turned more than 180° from the present heading (e.g., a 270° turn to the right). However, Selected Heading changes of more than 340° at a time result in turn reversals.

**Navigation Mode (GPS, VOR, LOC)**

**NOTE:** The selected navigation receiver must have a valid VOR or LOC signal or active GPS course for the flight director to enter Navigation Mode.

Pressing the NAV Key selects Navigation Mode. Navigation Mode acquires and tracks the selected navigation source on the HSI (GPS, VOR, LOC). The flight director follows GPS roll steering commands when GPS is the selected navigation source.

When the HSI is coupled to VOR or LOC, the flight director creates roll steering commands from the Selected Course and deviation. Navigation Mode can also be used to fly non-precision GPS and LOC approaches where glideslope capture is not required.

If the Course Deviation Indicator (CDI) shows greater than one dot when the NAV Key is pressed, the selected mode is armed. The armed annunciation appears in white to the left of the active roll mode. For cases where the projected course is offset a large distance from the present course for turn anticipation, GPS Navigation Mode can be activated with crosstrack error up to 10 nm when the NAV Key is pressed.
When the CDI has automatically switched from GPS to LOC during a LOC/ILS approach, GPS Navigation Mode remains active, providing GPS steering guidance until the localizer signal is captured. LOC Navigation Mode is armed in anticipation of localizer signal capture if the APR Key is not pressed prior to the automatic source switch.

If Navigation Mode is active and either of the following occur, the flight director reverts to Roll Hold Mode (wings rolled level):

- Different VOR is tuned while in VOR Navigation Mode (VOR Navigation Mode reverts to armed)
- Navigation source is manually switched
- Localizer signal is not captured by the final approach fix (FAF) while in LOC Navigation Mode

**Changing the Selected Course**

The Selected Course on the PFD is controlled using the CRS Knob. Pressing the CWS Button and hand-flying the aircraft does not change the Selected Course while in Navigation Mode. The autopilot guides the aircraft back to the Selected Course (or GPS flight plan) when the CWS Button is released.

**Approach Mode (GPS, VAPP, LOC)**

*NOTE: The selected navigation receiver must have a valid VOR or LOC signal or active GPS course for the flight director to enter Approach Mode.*

Approach Mode is activated when the APR Key is pressed. Approach Mode acquires and tracks the selected navigation receiver on the HSI (GPS, VOR, or LOC), depending on the loaded approach. This mode uses the selected navigation receiver deviation and desired course inputs to fly the approach. Approach Mode provides greater sensitivity for signal tracking than Navigation Mode.

Pressing the APR Key when the CDI is greater than one dot arms the selected approach mode (annunciated in white to the left of the active roll mode). If the selected navigation receiver is GPS, pressing the APR Key arms GPS Approach Mode, provided that a GPS approach has been loaded into the flight plan. If the loaded approach provides WAAS-based vertical guidance, Glidepath Mode is also armed (Figure 6-16). If GPS Approach Mode is selected while in GPS Navigation Mode, capture can occur with crosstrack error of up to 2 nm.

![Figure 6-24 GPS Approach Mode Armed](image)

LOC Approach Mode allows the autopilot to fly a LOC/ILS approach with a glideslope. LOC Approach Mode is armed (along with Glideslope Mode; see Figure 6-17) when the APR Key is pressed and either of the following have been done:

- Navigation source is set to LOC
- A LOC/ILS approach is loaded into the flight plan and the corresponding localizer frequency tuned (even if the selected navigation source is GPS)

Localizer capture is suppressed until the navigation source is changed to LOC.

If Approach Mode is active and either of the following occur, the flight director reverts to Roll Hold Mode (wings rolled level):

- Vectors-to-Final is activated
- Navigation source is manually switched
- Localizer signal is not captured by the final approach fix (FAF) while in LOC Navigation Mode

**Changing the Selected Course**

The Selected Course on the PFD is controlled using the CRS Knob. Pressing the CWS Button and hand-flying the aircraft does not change the Selected Course while in Approach Mode. The autopilot guides the aircraft back to the Selected Course (or GPS flight plan) when the CWS Button is released.
**Backcourse Mode (BC)**

**NOTE:** When making a backcourse approach, set the Selected Course to the localizer front course.

Backcourse Mode captures and tracks a localizer signal. The mode may be selected by pressing the **BC** Key. Backcourse Mode is armed if the CDI is greater than one dot when the mode is selected. The flight director creates roll steering commands from the Selected Course and deviation when in Backcourse Mode.

**Changing the Selected Course**

The Selected Course on the PFD is controlled using the **CRS** Knob. Pressing the **CWS** Button and hand-flying the aircraft does not reset any reference data while in Backcourse Mode. The autopilot guides the aircraft back to the Selected Course (or GPS flight plan) when the **CWS** Button is released.

![Image of Backcourse Mode](image_url)
6.4 AUTOPilot OPERATION

NOTE: Refer to the AFM for specific instructions regarding emergency procedures.

Cessna Nav III's autopilot operates flight control surface servos to provide automatic flight control. The autopilot controls the aircraft pitch and roll attitudes following commands received from the flight director. Pitch autotrim provides trim commands to the pitch trim adapter to relieve any sustained effort required by the pitch servo.

Flight Control

Pitch and roll commands are provided to the servos, based on the active flight director modes. Servo motor control limits the maximum servo speed and torque. The servo mounts are equipped with slip-clutches set to certain values. This allows the servos to be overridden in case of an emergency.

Pitch Axis and Pitch Trim

The autopilot pitch axis uses pitch rate to stabilize the aircraft pitch attitude during upsets and flight director maneuvers. Flight director pitch commands are rate- and attitude-limited, combined with pitch damper control, and sent to the pitch servo motor. The pitch servo measures the output effort (torque) and provides this signal to the pitch trim adapter. The pitch trim servo commands the motor to reduce the average pitch servo effort.

When the autopilot is not engaged, the pitch trim servo may be used to provide manual electric trim. This allows the aircraft to be trimmed using a control stick switch rather than the trim wheel. Manual trim commands are generated with the MET Switch. Trim speeds are scheduled with airspeed to provide more consistent response.

Roll Axis

The autopilot roll axis uses roll rate to stabilize aircraft roll attitude during upsets and flight director maneuvers. The flight director roll commands are rate- and attitude-limited, combined with roll damper control, and sent to the roll servo motor.

Engaging the Autopilot

NOTE: Autopilot engagement/disengagement is not equivalent to servo engagement/disengagement. Use the CWS Button to disengage the pitch and roll servos while the autopilot remains active.

When the AP Key is pressed, the autopilot and flight director (if not already engaged) are activated. Engagement is indicated by a green ‘AP’ annunciation in the center of the AFCS Status Box. The flight director engages in Pitch and Roll Hold modes when initially activated.

Figure 6-26 Autopilot Engaged
Control Wheel Steering

During autopilot operation, the aircraft may be hand-flown without disengaging the autopilot. Pressing and holding the CWS Button disengages the pitch and roll servos from the flight control surfaces and allows the aircraft to be hand-flown. At the same time, the flight director is synchronized to the aircraft attitude during the maneuver. The ‘AP’ annunciation is temporarily replaced by ‘CWS’ in white for the duration of CWS maneuvers.

In most scenarios, releasing the CWS Button reengages the autopilot with a new reference. Refer to the flight director modes section for CWS behavior in each mode.

Disengaging the Autopilot

Automatic disengagement occurs due to:

- System failure
- Inability to compute default flight director modes (FD also disengages automatically)
- Invalid sensor data
- Stall warning

Automatic autopilot disengagement is indicated by a flashing red ‘AP’ annunciation and by the autopilot disconnect aural alert, which continue until acknowledged by pushing the AP DISC or MET Switch.

The autopilot may be manually disengaged by pushing the AP DISC or GA Switch or the AP Key on the MFD. Manual disengagement is indicated by a five-second flashing yellow ‘AP’ annunciation and a three-second autopilot disconnect aural alert. The AP DISC or MET Switch may be used to cancel the aural alert.
**6.5 EXAMPLE PROCEDURES**

**NOTE:** The diagrams in this section are for instructional purposes only and should not be used for navigation.

This section provides a scenario-based set of procedures showing various GFC 700 AFCS modes used during a flight. In this scenario, the aircraft departs Reid-Hillview Airport (KRHV) from runway 13L, enroute to Fresno Yosemite International Airport (KFAT). Per the filed flight plan, the aircraft climbs to an altitude of 7000 feet MSL at a heading of 070° to intercept airway V107.

Airway V107 is flown to Panoche VORTAC (PXN) using VOR navigation, then airway V230 is flown using a GPS flight plan. On this leg, the aircraft descends to 3000 feet prior to reaching Clovis VORTAC (CZQ) and the ILS approach to R29R at KFAT is loaded. The ILS approach is flown and a missed approach is executed.

### Departure

**Climbing to the Selected Altitude and flying an assigned heading:**

1) Upon takeoff, set the Selected Altitude to 7000 feet using the ALT Knob.

To have the autopilot capture this altitude, use Pitch Hold, Vertical Speed, or Flight Level Change Mode. In this example, Vertical Speed Mode is used.

2) Upon reaching the desired vertical speed of 1000 fpm, enable the flight director in Vertical Speed Mode.

   a) Press the VS Key to activate Vertical Speed Mode and arm Selected Altitude Capture Mode.

![Flight Plan Overview](Figure 6-30 Flight Plan Overview)
The Vertical Speed Reference may be adjusted after Vertical Speed Mode is selected using the **NOSE UP/NOSE DN** keys or pushing the **CWS** Button while hand-flying the aircraft to establish a new Vertical Speed Reference.

b) Press the **AP** Key to engage the autopilot in Roll Hold and Vertical Speed modes.

### Figure 6-31 Departure

3) Activate Heading Select Mode while the autopilot is engaged in the climb.

a) Use the **HDG** Knob to set the Selected Heading to 070°.

b) Press the **HDG** Key. The autopilot follows the Selected Heading Bug on the HSI and turns the aircraft to the desired heading.

4) As the aircraft nears the Selected Altitude, the flight director transitions to Selected Altitude Capture Mode, indicated by the green ‘ALTS’ annunciation flashing for up to ten seconds.

At 50 ft from the Selected Altitude, the green ‘ALT’ annunciation flashes for up to ten seconds; the autopilot transitions to Altitude Hold Mode and levels the aircraft.
Intercepting a VOR

During climb-out, the autopilot continues to fly the aircraft in Heading Select Mode. Airway V107 to Panoche VORTAC (PXN) should now be intercepted.

**Intercepting a VOR:**

1) Tune the VOR frequency and activate Navigation Mode.
   a) Use the **CDI** Softkey to set the navigation source to VOR.
   b) Use the **CRS** Knob to set the Selected Course to 116°. Note that at this point, the flight director is still in Heading Select Mode and the autopilot continues to fly 070°.
   c) Press the **NAV** Key to arm Navigation Mode. The white annunciation ‘VOR’ appears as the armed roll mode.

2) When the CDI shows less than one dot, the flight director changes from Heading Select to Navigation Mode, and the green annunciation ‘VOR’ flashes. The autopilot begins turning to intercept the Selected Course.

3) The autopilot continues the turn until the aircraft is established on the Selected Course.

![Figure 6-32 Intercepting a VOR](image)
Flying a Flight Plan/GPS Course

**NOTE:** Changing the navigation source cancels the flight director’s Navigation Mode and causes the flight director to automatically revert to Roll Hold Mode, wings level.

As the aircraft closes on Panoche VORTAC, a GPS flight plan is entered for the next leg, airway V230. To do this, PXN, CZQ, and KFAT are entered into the active flight plan.

The aircraft is currently tracking inbound on the 116° radial of Panoche VORTAC.

**Flying a GPS flight plan:**

1) Enter Panoche VORTAC (PXN) and Clovis VORTAC (CZQ) into the GPS flight plan, in preparation to intercept this leg.

2) Transition from VOR to GPS Navigation Mode.

3) Once the leg is intercepted, the autopilot continues to track GPS guidance to the end of the flight plan.

a) Use the **CDI** Softkey to select GPS as the navigation source. This causes the flight director to lose the VOR signal, and the VOR annunciation flashes yellow as the autopilot rolls the wings level.

   Flashes 10 sec, Transitions to Roll Hold Mode
   Unless **NAV** Key is Pressed

b) Press the **NAV** Key to activate GPS Navigation Mode and cause the autopilot to steer the aircraft toward the active flight plan leg.

   If the **NAV** Key is not pressed within ten seconds of the VOR signal loss, the flight director reverts to Roll Hold Mode

Figure 6-33 Transition to GPS Flight Plan
**Descent**

While flying the GPS leg from Panoche VORTAC to Clovis VORTAC, the aircraft is cleared to make a descent to 3000 feet in preparation for the approach to KFAT. Three examples are presented:

- Flight Level Change descent
- Vertical Path Tracking descent
- Non-path descent using Flight Level Change Mode

**Making a Flight Level Change descent:**

1) Enter Flight Level Change Mode.
   
a) Using the **ALT** Knob, set the Selected Altitude to 3000 feet.
   
b) Press the **FLC** Key to activate Flight Level Change Mode. The annunciation ‘FLC’ appears next to the Airspeed Reference, which defaults to the current aircraft airspeed, 142 knots. Altitude Hold Mode is armed automatically.

2) Reduce power to allow descent in Flight Level Change Mode. The autopilot maintains the Airspeed Reference.

3) As the aircraft nears the Selected Altitude, the flight director transitions to Selected Altitude Capture Mode, indicated by the green ‘ALTS’ annunciation flashing for up to ten seconds.

The green ‘ALT’ annunciation flashes for up to ten seconds upon reaching 50 ft from the Selected Altitude; the autopilot transitions to Altitude Hold Mode and levels the aircraft.

**Vertical Path Tracking descent to VNAV Target Altitude:**

1) Select VNAV flight control:
   
a) Using the **ALT** Knob, set the Selected Altitude below the VNAV Target Altitude of 3000 ft.
   
b) Press the **VNV** Key to arm Vertical Path Tracking Mode. The white annunciation ‘VPTH’ appears.
c) If Vertical Path Tracking Mode is armed more than five minutes prior to descent path capture, acknowledgment is required for the flight director to transition from Altitude Hold to Vertical Path Tracking Mode. To proceed with descent path capture if the white ‘VPTH’ annunciation begins flashing, Press the **VNV** Key

OR:

Turn the **ALT** Knob to adjust the Selected Altitude

If the descent is not confirmed by the time of interception, Vertical Path Tracking Mode remains armed and the descent is not captured.

2) When the top-of-descent (TOD) is reached, the flight director transitions to Vertical Path Tracking Mode and begins the descent to the VNAV Target Altitude. Intention to capture the VNAV Target Altitude is indicated by the white ‘ALTV’ annunciation.

3) As the aircraft nears the VNAV Target Altitude, the flight director transitions to VNAV Target Altitude Capture Mode, indicated by the green ‘ALTV’ annunciation flashing for up to ten seconds.

The green ‘ALT’ annunciation flashes for up to ten seconds upon reaching 50 ft from the VNAV Target Altitude; the autopilot transitions to Altitude Hold Mode and levels the aircraft at the vertical waypoint.

Non-path descents can be flown while under VNAV guidance using Pitch Hold, Vertical Speed, or Flight Level Change Mode. Flight Level Change Mode is used for this example.

**Non-path descent:**

1) Command a non-path descent using Flight Level Change Mode:

a) Using the **ALT** Knob, set the Selected Altitude below the current aircraft altitude to 5,000 ft.
b) Press the FLC Key before the planned TOD during a vertical profile level leg (tracked using Altitude Hold Mode). The Airspeed Reference defaults to the current aircraft airspeed. Vertical Path Tracking and Selected Altitude Capture Mode is armed automatically.

| GPS | AP | FLC 142KT ALTS VPTH |

2) Reduce power to allow descent in Flight Level Change Mode. The autopilot maintains the Airspeed Reference.

3) As the aircraft nears the Selected Altitude, the flight director transitions to Selected Altitude Capture Mode, indicated by the green ‘ALTS’ annunciation flashing for up to ten seconds.

| GPS | AP | ALTS 3000FT ALT VPTH |

The green ‘ALT’ annunciation flashes for up to ten seconds upon reaching 50 ft from the Selected Altitude; the autopilot transitions to Altitude Hold Mode and levels the aircraft.

4) When the next TOD is reached, Vertical Path Tracking becomes active (may require acknowledgment to allow descent path capture).

| GPS | AP | VPTH ALTV |

5) As the aircraft nears the VNAV Target Altitude, the flight director transitions to VNAV Target Altitude Capture Mode, indicated by the green ‘ALTV’ annunciation flashing for up to ten seconds.

| GPS | AP | ALTV 2000FT ALT |

The green ‘ALT’ annunciation flashes for up to ten seconds upon reaching 50 ft from the VNAV Target Altitude; the autopilot transitions to Altitude Hold Mode and levels the aircraft at the vertical waypoint.

| GPS | AP | ALT 2000FT |

Figure 6-36 Non-path Descent
Approach

Flying an ILS approach:
1) Transition from GPS Navigation Mode to Heading Select Mode.
   a) Load the Runway 29R ILS approach for KFAT into the flight plan.
   b) Select ‘VECTORS’ for the transition.
   c) Use the HDG Knob to set the Selected Heading after getting vectors from ATC.
   d) Press the HDG Key. The autopilot turns the aircraft to the desired heading.

2) Use Heading Select Mode to comply with ATC vectors as requested.
3) Tune the localizer frequency.
4) Press the APR Key when cleared for approach to arm Approach and Glideslope modes. ‘LOC’ and ‘GS’ appear in white as armed mode annunciations.

<table>
<thead>
<tr>
<th>LOC</th>
<th>HDG</th>
<th>AP</th>
<th>ALT 3000FT</th>
<th>GS</th>
</tr>
</thead>
</table>

The navigation source on the HSI automatically switches to LOC. After this switch occurs, the localizer signal can be captured and the autopilot and flight director determine when to begin the turn to intercept the final approach course.

5) There are two options available at this point, as the autopilot flies the ILS approach:
   • Push the AP DISC Switch at the Decision Height and land the aircraft.
   • Use the GA Switch to execute a missed approach.

Figure 6-37  ILS Approach to KFAT
NOTE: Support for WAAS precision approaches is available only in installations with GIA 63W Integrated Avionics Units and when WAAS is available.

Flying a WAAS precision approach:
1) Arm flight director modes for a precision approach:
   a) Make sure GPS is the selected navigation source (use CDI Softkey to change navigation source).
   b) Load the Runway 29R LPV approach for KFAT into the flight plan.
2) Press the APR Key once clearance for approach has been acquired. GPS Approach Mode is activated (provided the crosstrack error is less than 2 nm) and Glidepath Mode is armed.
3) Once the glidepath is captured, Glidepath Mode becomes active.
4) There are two options available at this point, as the autopilot flies the approach:
   • Push the AP DISC Switch at the Decision Height and land the aircraft.
   • Use the GA Switch to execute a missed approach.

Figure 6-38 LPV Approach to KFAT
Go Around/Missed Approach

**NOTE:** As a result of the calculations being performed by the system while flying the holding pattern, the display may re-size automatically and the aircraft may not precisely track the holding pattern.

Flying a missed approach:

1) Push the **GA** Switch at the Decision Height and apply full power to execute a missed approach. The flight director Command Bars establish a climb to follow.

Note that when the **GA** Switch is pushed, the missed approach is activated and the autopilot disconnects, indicated by the ‘AP’ annunciation flashing yellow for five seconds.

```
Flashes 5 sec
```

2) Start the climb to the prescribed altitude in the published Missed Approach Procedure.
   a) Press the **AP** Key to re-engage the autopilot.
   b) Select the **CDI** Softkey to select GPS as the navigation source.
   c) Press the **NAV** Key to have the autopilot fly to the hold.

```
GPS AP PIT ALT
```

As the aircraft nears the Selected Altitude, the flight director transitions to Selected Altitude Capture Mode, indicated by the green ‘ALT’ annunciation flashing for up to ten seconds.

```
GPS AP ALTS 2000FT ALT
```

The green ‘ALT’ annunciation flashes for up to ten seconds upon reaching 50 ft from the Selected Altitude; the autopilot transitions to Altitude Hold Mode and levels the aircraft.

3) The autopilot flies the holding pattern. Annunciations are displayed in the Navigation Status Box, above the AFCS Status Box.

```
4.0 NM C20 DIS 0.1 NM BRG 152°
```

```
GPS AP ALT 2000FT
```

Figure 6-39 Go Around/Missed Approach
6.6  **AFCS ANNUNCIATIONS AND ALERTS**

**AFCS Status Alerts**

The following annunciations (listed in order of increasing priority) can appear on the PFD above the Airspeed and Attitude indicators. Only one annunciation may occur at a time, and messages are prioritized by criticality.

![Figure 6-40 AFCS Status Annunciation](image)

<table>
<thead>
<tr>
<th>Alert Condition</th>
<th>Annunciation</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>Aileron Mistrim Right</td>
<td>AIL →</td>
<td>Roll servo providing sustained force in the indicated direction</td>
</tr>
<tr>
<td>Aileron Mistrim Left</td>
<td>←AIL</td>
<td>Roll servo providing sustained force in the indicated direction</td>
</tr>
<tr>
<td>Elevator Mistrim Down</td>
<td>↓ELE</td>
<td>Pitch servo providing sustained force in the indicated direction</td>
</tr>
<tr>
<td>Elevator Mistrim Up</td>
<td>↑ELE</td>
<td>Pitch servo providing sustained force in the indicated direction</td>
</tr>
<tr>
<td>Pitch Trim Failure (or stuck MET Switch)</td>
<td>PTRM</td>
<td>If AP engaged, take control of the aircraft and disengage AP</td>
</tr>
<tr>
<td></td>
<td></td>
<td>If AP disengaged, move MET switches separately to unstick</td>
</tr>
<tr>
<td>Roll Failure</td>
<td>ROLL</td>
<td>Roll axis control failure; AP inoperative</td>
</tr>
<tr>
<td>Pitch Failure</td>
<td>PITCH</td>
<td>Pitch axis control failure; AP inoperative</td>
</tr>
<tr>
<td>System Failure</td>
<td>AFCS</td>
<td>AP and MET are unavailable; FD may still be available</td>
</tr>
<tr>
<td>Preflight Test</td>
<td>PFT</td>
<td>Performing preflight system test; aural alert sounds at completion</td>
</tr>
<tr>
<td></td>
<td></td>
<td>Do not press the AP DISC Switch during servo power-up and preflight system tests as this may cause the preflight system test to fail or never to start (if servos fail their power-up tests). Power must be cycled to the servos to remedy the situation.</td>
</tr>
<tr>
<td></td>
<td>PFT</td>
<td>Preflight system test failed; aural alert sounds at failure</td>
</tr>
</tbody>
</table>

Table 6-4  AFCS Status Field Alerts
Overspeed Protection

Overspeed protection is provided in situations where the flight director cannot acquire and maintain the vertical Mode Reference for the selected vertical mode without exceeding the certified maximum autopilot airspeed.

When an autopilot overspeed condition occurs, the Airspeed Reference appears in a box above the Airspeed Indicator, flashing a yellow ‘MAXSPD’ annunciation. Engine power should be reduced and/or the pitch reference adjusted to slow the aircraft. The annunciation disappears when the overspeed condition is resolved.

Figure 6-41  Overspeed Annunciation
SECTION 7: NAVIGATION

The majority of the following discussions pertain to the Multi Function Display. In discussions pertaining to the PFD, the controls are located on the PFD.

7.1 NAVIGATION MAP PAGE

WARNING: The map display should only be used for situational awareness. Any map display indication should be compared with approved navigation sources.

Select the MAP Page Group

1) Turn the large FMS Knob until ‘MAP’ is highlighted in the page group box in the lower right of the MFD display.

2) Turn the small FMS Knob to select the first MAP page (indicated by a solid rectangular icon).

7.2 DIRECT-TO NAVIGATION

Direct-to Navigation from the MFD

Enter a Direct-to Destination

1) Press the Direct-to (D) Key.
2) Enter the destination waypoint identifier.
3) Press the ENT Key to confirm the identifier. The ‘Activate?’ field is highlighted.
4) If no altitude constraint or course is desired, press the ENT Key to activate. To enter an altitude constraint, proceed to step 5.
5) Turn the large FMS Knob to place the cursor over the ‘VNAV’ altitude field.
6) Enter the desired altitude.
7) Press the ENT Key. The option to select ‘MSL’ or ‘AGL’ is now displayed.
8) Turn the small FMS Knob to select ‘MSL’ or ‘AGL’.
9) Press the ENT Key. The cursor is placed in the ‘VNAV’ offset distance field.

10) Enter the desired target altitude offset from the selected Direct-to.

11) Press the ENT Key to highlight ‘Activate?’ or turn the large FMS Knob to highlight the ‘COURSE’ field.

12) Enter the desired course to the waypoint.

13) Press the ENT Key to highlight ‘ACTIVATE?’.

14) Press the ENT again to activate the Direct-to.

**Select a Direct-to Destination to a Flight Plan Waypoint**

1) While navigating an active flight plan, press the Direct-to (D) Key.

2) Turn the small FMS Knob to the left to display a list of flight plan waypoints as shown in Figure 7-3.

3) Turn the large FMS Knob to select the desired waypoint.

4) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.

5) Press ENT again to activate a Direct-to.

**Select a Direct-to Destination to a Nearest Airport**

1) Press the Direct-to (D) Key.

2) Turn the small FMS Knob to the left. Initially, a flight plan waypoint list is displayed as in Figure 7-3. The list is populated only when navigating a flight plan.

3) Turn the small FMS Knob to the right to display the ‘NRST’ airports to the aircraft’s current position as shown in Figure 7-4.

4) Turn the large FMS Knob to select the desired airport.

5) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.

6) Press ENT again to activate a Direct-to.

**Select a Direct-to Destination to a Recently Entered Identifier**

1) Press the Direct-to (D) Key.

2) Turn the small FMS Knob to the left. Initially, a flight plan waypoint list is displayed as in Figure 7-3. The list is populated only when navigating a flight plan.
3) Turn the small FMS Knob to the right to display the ‘RECENT’ waypoints as shown in Figure 7-5.

4) Turn the large FMS Knob to select the desired airport.

5) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.

6) Press ENT again to activate a Direct-to.

Re-center the CDI to the Destination Waypoint

Press the Direct-to (Direct-to) Key, followed by pressing the ENT Key twice. If a missed approach point (MAP) is the current destination, the approach will be canceled.

Manually Define the Active Direct-to

1) Press the Direct-to (Direct-to) Key.

2) Turn the large FMS Knob to highlight the ‘VNAV’ altitude field.

3) Enter the desired altitude.

4) Press the ENT Key. The option to select ‘MSL’ or ‘AGL’ is now displayed.

5) Turn the small FMS Knob to select ‘MSL’ or ‘AGL’.

6) Press the ENT Key. The cursor now highlights the VNAV offset field.

7) Enter the desired offset distance.

8) Press the ENT Key.

9) Turn the large FMS Knob to place the cursor in the ‘COURSE’ field.

10) Enter the desired course.

11) Press the ENT Key. The cursor now highlights ‘ACTIVATE?’.

12) Press the ENT Key again to begin navigation using the selected destination, altitude constraint, and course.

Canceling Direct-to Navigation

1) Press the Direct-to (Direct-to) Key.

2) Press the MENU Key to display the Direct-to options menu.

3) With ‘Cancel Direct-To NAV’ highlighted, press the ENT Key. If a flight plan is still active, the G1000 resumes navigating the flight plan along the closest leg.

Direct-to Navigation from the PFD

Figure 7-6 Canceling Direct-to Navigation

Figure 7-7 PFD Direct-to Window
Enter a Direct-to Destination

1) Press the Direct-to Key (D•).
2) Turn the large FMS Knob to place the cursor in the desired selection field.
3) Turn the small FMS Knob to begin selecting the desired identifier, location, etc.
4) Press the ENT Key.
5) The cursor is now flashing on ‘ACTIVATE?’.

If no altitude constraint or course is desired, press the ENT Key to activate. To enter an altitude constraint, proceed to step 6.

6) Turn the large FMS Knob to place the cursor over the ‘VNAV’ altitude field.
7) Turn the small FMS Knob to enter the desired VNAV altitude.
8) Press the ENT Key. The option to select ‘MSL’ or ‘AGL’ is now displayed.
9) Turn the small FMS Knob to select ‘MSL’ or ‘AGL’.
10) Press the ENT Key. The cursor is placed in the ‘VNAV’ offset distance field.
11) Turn the small FMS Knob to enter the desired target altitude offset from the selected Direct-to.
12) Press the ENT Key to highlight ‘Activate?’ or turn the large FMS Knob to highlight the ‘COURSE’ field.
13) Turn the small FMS Knob to enter the desired course to the waypoint.
14) Press the ENT Key to highlight ‘ACTIVATE?’.
15) Press the ENT again to activate the Direct-to.

Select a Direct-to Destination to a Flight Plan Waypoint

1) While navigating an active flight plan, press the Direct-to (D•) Key.
2) Turn the small FMS Knob to the left to display a list of flight plan waypoints as shown in Figure 7-8.

3) Turn the large FMS Knob to select the desired waypoint.
4) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.
5) Press ENT again to activate a Direct-to.

Select a Direct-to Destination to a Nearest Airport

1) Press the Direct-to (D•) Key.
2) Turn the small FMS Knob to the left. Initially, a flight plan waypoint list is displayed as in Figure 7-8. The list is only populated when navigating a flight plan.

3) Turn the large FMS Knob to select the desired waypoint.
4) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.
5) Press ENT again to activate a Direct-to.
3) Turn the small FMS Knob to the right to display the ‘NRST’ airports to the aircraft’s current position as shown in Figure 7-9.

4) Turn the large FMS Knob to select the desired airport.

5) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.

6) Press ENT again to activate a Direct-to.

Select a Direct-to Destination to a Recently Entered Identifier

1) Press the Direct-to (D-) Key.

2) Turn the small FMS Knob to the left. Initially, a flight plan waypoint list is displayed as in Figure 7-8. The list is only populated when navigating a flight plan.

Fig. 7-10 Recently Entered Waypoints List (PFD)

3) Turn the small FMS Knob to the right to display the ‘RECENT’ waypoints as shown in Figure 7-10.

4) Turn the large FMS Knob to select the desired airport.

5) Press the ENT Key. The cursor is now displayed on ‘ACTIVATE?’.

6) Press ENT again to activate a Direct-to.

Cancelling Direct-to Navigation

1) Press the Direct-to (D-) Key.

2) Press the MENU Key to display the Options Window. The cursor will be flashing on ‘Cancel Direct-to NAV’.

3) Press the ENT Key to cancel the direct-to.

7.3 NAVIGATING A FLIGHT PLAN

The following discussion is an example of navigating a vertical flight plan with the optional WAAS capable GPS system. A lateral flight plan (LNAV) would be navigated in much the same way, but would not include the vertical guidance after the Initial Approach Fix.

The example is a flight plan from KMCI to KDFW. An LPV (WAAS) approach is used for runway 17L. See the Automatic Flight Control section for a discussion on flying a vertical flight plan with the autopilot coupled.

1) Prior to departure, the RACER3 departure is loaded. See Loading and Activating a Departure in the Procedures section. Note the magenta arrow indicating the active departure leg. ‘DPRT’ is the current CDI flight mode displayed on the HSI.

Fig. 7-11 Departure
2) The aircraft is being flown along the active MCI to DOSOA leg.
3) Now that DOSOA is passed and the DOSOA to KDFW leg is active, maintain a track along the magenta line (as in Figure 7-12) keeping the CDI centered. Note the CDI flight mode is in enroute (ENR) mode.

![Figure 7-12 Begin Enroute Leg](image)

4) It is determined that the BYP5 arrival for runway 17L will be used and is loaded and active as shown in Figure 7-13. See Loading and Activating an Arrival in the Procedures section.

5) It is now determined that the aircraft will proceed directly to MLC.
   a) Press the FMS Knob to activate the cursor.
   b) Turn the large FMS Knob to select MLC in the flight plan list.
   c) Press the Direct-to (D) Key.
   d) Press the ENT Key twice to activate a direct-to MLC. Note, in Figure 7-14, the magenta arrow indicating the direct-to MLC. Keep the CDI centered and maintain a track along the magenta line to MLC.

![Figure 7-13 Loaded Arrival](image)
6) While proceeding to MLC, instructions are received to descend to 8,000 feet prior to BYP. An Along-track Offset waypoint will be used to accomplish this.

a) Press the FMS Knob to activate the cursor.

b) Turn the large FMS Knob to highlight the BYP waypoint.

c) Press the ATK OFST Softkey.

d) Enter the desired offset distance. Turn the small FMS Knob to select the distance before (-) or after (+) the waypoint. In this case, 3 nm prior to BYP will be used.

e) Press the ENT Key. The cursor is now displayed in the ‘ALT’ field.

f) Enter the desired altitude constraint, in this case 8,000 feet.

g) Press the ENT Key.
h) Press the VNV PROF Softkey to place the cursor in the target vertical speed field (VS TGT) as shown in Figure 7-17.

i) At this point, the desired descent vertical speed can be selected, or the desired descent flight path angle (FPA) can be selected. Turn the large FMS Knob to select the desired selection field, then turn the small FMS Knob to entered the desired value. In this example, a 500 foot per minute vertical speed is selected.

j) Press the ENT Key.

7) The aircraft is approaching MLC. The upcoming turn and next heading will be annunciated at the top left of the PFD. Initiate the turn and maneuver the aircraft on a track through the turn to intercept the magenta line for the MLC to BYP leg. Keep the CDI centered.

Note the information now displayed in the ‘CURRENT VNAV PROFILE’ box. Also, note the offset waypoint and gray circle are now displayed on the map. The gray circle marks the Top of Descent (TOD). Vertical guidance will be provided at the TOD that will result in a 500 foot per minute descent to an altitude of 8,000 feet upon reaching the offset waypoint.
8) After passing MLC the previous leg will no longer be displayed on the map. The magenta arrow in the flight plan list now indicates the MLC to BYP leg is now active. The CDI flight mode will change from enroute to terminal (TERM). Keep the CDI centered and maintain a track to follow the magenta line.

9) The aircraft has reached TOD. Note the target vertical speed and the required vertical speed that will be required to reached the selected altitude. The Vertical Descent Indicator (VDI) and the Required Vertical Speed Indicator (RVSI) are now displayed on the PFD.

Figure 7-19 Interception of MLC to BYP Leg

Figure 7-20 Reaching Top of Descent (TOD)

Figure 7-21 VDI & RVSI Upon Reaching Top of Descent (TOD)
10) Establish a descent vertical speed which places the VSI pointer in line with the RVSI. Upon reaching the offset waypoint, the aircraft will be at 8,000 feet.

11) The aircraft is approaching BYP. The upcoming turn and next heading will be annunciated at the top left of the PFD. Initiate the turn and maneuver the aircraft on a track through the turn radius to intercept the magenta line for the BYP to KARLA leg and center the CDI.
12) After passing BYP the next leg of the arrival will turn magenta. The magenta arrow in the flight plan list now indicates the MLC to BYP leg is now active. Keep the CDI centered and maintain a track to follow the magenta line.

13) It is now determined that an LPV approach to runway 17L with a transition at RIVET will be used and is loaded. See Loading an Approach in the Procedures section.

If the GPS system is not WAAS capable, or WAAS is not available, the LPV approach will not be displayed in the list of available approaches. Selecting the LNAV approach for runway 17L will give vertical guidance up to the IAF.

The altitudes displayed in white in the ‘VNAV ALT’ field are obtained from the database. Altitudes displayed as light blue are computed by the G1000. The approach may be activated at any point to proceed directly to the IAF. In this example, the aircraft will progress through the final waypoint of the arrival and the flight plan will automatically sequence to the FAF as the active leg.

Manually activate the approach by performing the following steps:

a) Press the **PROC** Key.
b) Turn the large **FMS** Knob to highlight ‘ACTIVATE APPROACH’.
c) Press the **ENT** Key to activate the approach.

14) At the point illustrated in Figure 7-27, the aircraft has passed COVIE and the turn at LEMYN has been accomplished. The IAF is the next waypoint. At the TOD, establish a descent vertical speed shown as required on the PFD. Align the vertical speed pointer and the RVSI.
15) Upon making the descending turn, the RIVET to DRAAK leg is now active. Keep the CDI centered to track the magenta line and maintain 4,000 feet to DRAAK.

16) At DRAAK, establish another descent as done previously to obtain an altitude of 3,000 feet at INWOD.
17) If the installed GPS system is WAAS capable, at (or just prior to) the FAF, the VDI changes to the Glidepath Indicator as shown in Figure 7-34. Continue the descent through INWOD using the Glidepath Indicator, as one would use a Glideslope Indicator, to obtain an altitude “AT” 2,300 feet at the FAF (MENOL). Also, the CDI flight mode will change from terminal mode to LPV mode. Note the altitude restriction lines over and under (‘At’) the altitude in the ‘ALT’ field.

If the installed GPS system is not WAAS capable, the Glidepath Indicator will not be displayed. The VDI will continue to give vertical guidance to the FAF, after which, the VDI will be removed from the display and vertical guidance is no longer provided. The CDI flight phase will display LNAV mode. Note that the altitude restriction lines in the ‘ALT’ field will only indicate ‘Above’ in a non-WAAS system.

18) Continue following the glidepath to maintain the descent to “AT or ABOVE” 1,000 feet at the MAP (RW17L).
19) Upon reaching the MAP, it is decided to execute a missed approach.
   a) Press the PROC Key.
   b) Turn the large FMS Knob to highlight ‘ACTIVATE MISSED APPROACH’.
   c) Press the ENT Key twice to activate the missed approach.

   **NOTE:** Upon reaching the MAP, automatic waypoint sequencing stops and the G1000 will cease to navigate. A missed approach must be activated to continue navigation.

   A direct-to is initiated to POLKE. Begin climbing to 5,000 feet. The CDI flight mode now changes from LPV mode to MAPR mode.

20) At POLKE follow the next turn prompts in the upper left of the PFD. Follow the POLKE to BACKS (MAHP) leg. Continue climbing to “AT or ABOVE” 5,000 feet at BACKS.
21) A holding pattern will be established at the MAHP (BACKS).

22) Maintain 5,000 feet and follow the magenta line through the hold.

**7.4 AIRPORT INFORMATION**

**Select the Airport Information Page**

1) Turn the large FMS Knob to select the ‘WPT’ page group.

2) Turn the small FMS Knob to select the first rectangular page icon. Initially, information for the airport closest to the aircraft’s present position is displayed.

**Select an Airport from the Database**

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Enter the desired airport identifier.

**Select an Airport from the Active Flight Plan**

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Turn the small FMS Knob to the left to display a list of flight plan airports as shown in Figure 7-43.
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3) Turn the large FMS Knob to select the desired airport.

4) Press the ENT Key.

Select a Nearest Airport

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Turn the small FMS Knob to the left. Initially, a flight plan airport list is displayed as in Figure 7-43. The list is populated only when navigating a flight plan.

3) Turn the small FMS Knob to the right to display the ‘NRST’ airports to the aircraft’s current position as shown in Figure 7-44.

4) Turn the large FMS Knob to select the desired airport.

5) Press the ENT Key.

Select a Recently Entered Airport Identifier

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Turn the small FMS Knob to the right to display the ‘RECENT’ airports as shown in Figure 7-45.

3) Turn the large FMS Knob to the right to display the ‘RECENT’ airports as shown in Figure 7-45.

4) Turn the large FMS Knob to select the desired airport.

5) Press the ENT Key.

Select an Airport by Facility Name or City Location

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to the right to select the facility name or location (city) field.

3) Enter the desired facility name or city.

4) Press the ENT Key. If there are duplicate names in the database, a list is displayed from which to choose the desired location.

5) To remove the flashing cursor, press the FMS Knob.

Access Runway Information

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to place the cursor on the ‘RUNWAYS’ identifier field.
3) Turn the small FMS Knob in the direction of the green arrow to display the next runway for the selected airport. Continue turning the small FMS Knob to select the desired runway.

4) To remove the flashing cursor, press the FMS Knob.

**Access Frequency Information**

1) With the Airport Information Page displayed, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to move the cursor to the Frequencies box.

3) Turn either FMS Knob to scroll through the list, placing the cursor on the desired frequency. If a listed frequency has sector or altitude restrictions, the frequency is preceded by an info (‘i’) designation. Press the ENT Key to view the information. The following may be displayed with the frequency:
   - ‘TX’ – transmit only
   - ‘RX’ – receive only
   - ‘PT’ – part time frequency

4) Press the ENT Key to place the selected frequency in the standby field of the COM or NAV box.

5) To remove the cursor, press the FMS Knob.

### 7.5 INTERSECTION INFORMATION

**Select the Intersection Information Page**

1) Turn the large FMS Knob to select the ‘WPT’ page group.

2) Turn the small FMS Knob to select the second rectangular page icon.

**Access Information on an Intersection**

1) With the Intersection Information Page displayed, press the FMS Knob to activate the cursor.

2) Enter an intersection identifier and press the ENT Key.

3) Press the FMS Knob to remove the flashing cursor.
7.6 NDB INFORMATION

Select the NDB Information Page

1) Turn the large FMS Knob to select the ‘WPT’ page group.
2) Turn the small FMS Knob to select the third rectangular page icon

View Information on a Specific NDB

1) With the NDB Information Page displayed, press the FMS Knob to activate the cursor.
2) Turn the large FMS Knob to highlight the desired selection field (identifier, name or closest city).
3) Enter an identifier, name or city and press the ENT Key.
4) Press the FMS Knob to remove the flashing cursor.

7.7 VOR INFORMATION

Select the VOR Information Page

1) Turn the large FMS Knob to select the ‘WPT’ page group.
2) Turn the small FMS Knob to select the fourth rectangular page icon.

Access Information on a VOR

1) With the VOR Information Page displayed, press the FMS Knob to activate the cursor.
2) Turn the large FMS Knob to highlight the desired selection field (identifier, name or closest city).
3) Enter an identifier, name or city and press the ENT Key.
4) The ‘FREQUENCY’ field is now highlighted. If desired, press the ENT Key to place the frequency in the NAV receiver standby field.
5) Press the FMS Knob to remove the flashing cursor.
7.8 USER WAYPOINT INFORMATION PAGE

See the Flight Planning section for a discussion on creating and modifying user defined waypoints.

7.9 NEAREST AIRPORTS

Figure 7-49 Nearest Airports Page

Nearest Airport Information on the MFD

Select the Nearest Airports Page

1) Turn the large FMS Knob to select the ‘NRST’ page group.
2) Turn the small FMS Knob to select the first rectangular page icon.

Initially, the closest airport to the aircraft’s present position is displayed.

Access Information on a Specific Airport

1) With the Nearest Airports Page displayed, press the APT Softkey to place the cursor in the ‘NEAREST AIRPORTS’ field. The first airport in the nearest airports list is highlighted.
2) Press the FMS Knob, then turn the large FMS Knob to highlight the desired airport.
3) Press the FMS Knob to remove the flashing cursor.

Access Runway Information for the Selected Airport

1) With the Nearest Airports Page displayed, press the RNWY Softkey to place the cursor in the ‘RUNWAYS’ field.
2) Turn the small FMS Knob to select the desired runway.
3) Press the FMS Knob to remove the flashing cursor.

Quickly Tune the COM Transceiver to a Nearby Airport Frequency

1) With the Nearest Airports Page displayed, press the FREQ Softkey to place the cursor in the ‘FREQUENCIES’ field.
2) Turn either FMS Knob to select the desired frequency.
3) Press the ENT Key. The selected frequency is placed in the COM standby frequency field.
4) Press the Frequency Transfer Key to place the frequency in the active field.
5) Press the FMS Knob to remove the flashing cursor.
Nearest Airports Information on the PFD

<table>
<thead>
<tr>
<th>Airport Symbol</th>
<th>Bearing TO</th>
<th>Distance</th>
<th>Primary COM Frequency</th>
</tr>
</thead>
<tbody>
<tr>
<td>USG</td>
<td>325°</td>
<td>9.3NM</td>
<td>VFR</td>
</tr>
<tr>
<td>MULTICOM</td>
<td>122.900</td>
<td>RWY</td>
<td>3500FT</td>
</tr>
<tr>
<td>KIDA</td>
<td>261°</td>
<td>9.5NM</td>
<td>ILS</td>
</tr>
<tr>
<td>TOWER</td>
<td>118.500</td>
<td>RWY</td>
<td>9001FT</td>
</tr>
<tr>
<td>KRKE</td>
<td>352°</td>
<td>20.1NM</td>
<td>VOR</td>
</tr>
<tr>
<td>UNICOM</td>
<td>122.800</td>
<td>RWY</td>
<td>4200FT</td>
</tr>
</tbody>
</table>

Figure 7-50 Nearest Airports Window

Press the NRST Softkey to display the PFD Nearest Airports Window.

View Information on a Specific Airport in the List

1) With the Nearest Airports Window displayed, turn either FMS Knob to place the cursor on the desired airport identifier.
2) Press the ENT Key to display airport information.
3) Press the ENT Key again (cursor is on ‘BACK’) to return to the list.

Load an Airport COM Frequency into the Active COM

1) With the Nearest Airports Window displayed, turn either FMS Knob to place the cursor on the desired airport frequency shown in the window.
2) Press the ENT Key and the selected frequency will be placed in the COM standby frequency field.
3) Press the Frequency Transfer Key to make the frequency the active frequency.

7.10 NEAREST INTERSECTIONS

Select the Nearest Intersections Page

1) Turn the large FMS Knob to select the ‘NRST’ page group.
2) Turn the small FMS Knob to select the second rectangular page icon.

View Information on the Nearest Intersection

1) With the Nearest Intersections Page displayed, press the FMS Knob to activate the cursor.
2) Turn either FMS Knob to select the desired intersection.
3) Press the FMS Knob to remove the flashing cursor.
7.11 NEAREST NDB

Select the Nearest NDB Page

1) Turn the large FMS Knob to select the ‘NRST’ page group.
2) Turn the small FMS Knob to select the third rectangular page icon.

Access Information on a Specific NDB

1) With the Nearest NDB Page displayed, press the FMS Knob to activate the cursor.
2) Turn either FMS Knob to select the desired NDB. The remaining information on the Nearest NDB Page pertains to the selected NDB.
3) Press the FMS Knob to remove the flashing cursor.

7.12 NEAREST VOR

Select the Nearest VOR Page

1) Turn the large FMS Knob to select the ‘NRST’ page group.
2) Turn the small FMS Knob to select the fourth rectangular page icon.

View Information on the Nearest VOR

1) With the Nearest VOR Page displayed, press the VOR Softkey to place the cursor in the ‘NEAREST VOR’ box.
2) Turn either FMS Knob to select a VOR.
3) Press the FMS Knob to remove the flashing cursor.

Select and Load a VOR Frequency

1) With the Nearest VOR Page displayed, press the FREQ Softkey to highlight the VOR frequency in the ‘FREQUENCY’ field.
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2) Press the **ENT** Key. The selected VOR frequency is placed in the NAV standby frequency field.

3) Press the **FMS** Knob to remove the flashing cursor.

### 7.13 NEAREST USER WAYPOINT

**Figure 7-54 Nearest User Waypoints Page**

**Select the Nearest User Waypoint Page**

1) Turn the large **FMS** Knob to select the ‘NRST’ page group.

2) Turn the small **FMS** Knob to select the fifth rectangular page icon.

**Select a Nearest User Waypoint**

1) With the Nearest User Waypoint Page displayed, press the **FMS** Knob to activate the cursor. If any previously entered User Waypoints are within 200 nm, these will be displayed with the closest listed first.

2) Turn either **FMS** Knob to select the desired waypoint. The remaining information on the Nearest User Waypoint Page pertains to the selected Nearest User Waypoint.

3) Press the **FMS** Knob to remove the flashing cursor.

### 7.14 NEAREST FREQUENCIES

**Figure 7-55 Nearest Frequencies Page**

**Select the Nearest Frequencies Page**

1) Turn the large **FMS** Knob to select the ‘NRST’ page group.

2) Turn the small **FMS** Knob to select the sixth rectangular page icon.

**Select and Load the Nearest ARTCC, FSS, or Weather Frequency**

1) With the Nearest Frequencies Page displayed, press the **ARTCC**, **FSS**, or **Wx** Softkey to place the cursor in the appropriate field.

2) Turn the **FMS** Knobs to select the desired facility or frequency.

3) Press the **ENT** Key to load the frequency into the
COM frequency standby field.

4) Press the **FMS** Knob to remove the flashing cursor.

# 7.15 NEAREST AIRSPACES

![Figure 7-56 Nearest Airspaces Page](image)

**Select the Nearest Airspaces Page**

1) Turn the large **FMS** Knob to select the ‘NRST’ page group.

2) Turn the small **FMS** Knob to select the seventh rectangular page icon.

**Airspace Alerts Box**

- If the projected course takes the aircraft inside an airspace within the next ten minutes, ‘Ahead’ is displayed.
- If the aircraft is within two nautical miles of an airspace and the current course takes the aircraft inside, ‘Ahead < 2 nm’ is displayed.
- If the aircraft is within two nautical miles of an airspace and the current course will not take the aircraft inside, ‘Within 2 nm’ is displayed.

- If the aircraft has entered an airspace, ‘Inside’ is displayed.

**View Additional Details for a Listed Airspace**

1) With the Nearest Airspace Page displayed, press the **FMS** Knob to activate the cursor.

2) Turn the large **FMS** Knob to scroll through the list, highlighting the desired airspace.

3) Press the **ALERTS** Softkey to place the cursor in the ‘AIRSPACE ALERTS’ field.

4) Turn either **FMS** Knob to select the desired airspace.

5) Press the **FMS** Knob to remove the flashing cursor.

**View and Quickly Load the Frequency for a Controlling Agency**

1) With the Nearest Airspace Page displayed, press the **FREQ** Softkey to place the cursor in ‘FREQUENCIES’ field.

2) Turn either **FMS** Knob to select the desired frequency.

3) Press the **ENT** Key to load the frequency into the COM frequency standby field.

4) Press the **FMS** Knob to remove the flashing cursor.
SECTION 8: FLIGHT PLANNING

The following discussions pertain to the Multi Function Display, unless otherwise indicated.

8.1 USER DEFINED WAYPOINTS

Select the User WPT Information Page

1) Turn the large FMS Knob to select the ‘WPT’ page group.
2) Turn the small FMS Knob to select the fifth rectangular page icon.

Create a New User Waypoint

1) With the User Waypoint Information Page displayed, press the NEW Softkey. A waypoint is created at the current aircraft position.
2) Turn the small FMS Knob to enter first character of the waypoint name.
3) Turn the large FMS Knob to move the cursor to the next character field.
4) Repeat steps 2 and 3 until the desired name is entered.
5) Press the ENT Key.
6) Press the ENT Key again to accept the new waypoint identifier.
7) The cursor is now in the ‘REFERENCE WAYPOINTS’ field. If desired, the waypoint can be defined by a reference waypoint. Turn the FMS Knobs to enter the position coordinates or the radial and distance from the reference waypoint.
8) Press the FMS Knob to remove the flashing cursor.

Modify a User Waypoint

1) With the User Waypoint Information Page displayed, press the FMS Knob to activate the cursor.
2) Turn the large FMS Knob to move the cursor to the desired field and turn the small FMS Knob to make changes.
3) Press the ENT Key to accept the changes.
4) Press the FMS Knob to remove the flashing cursor.

Delete a User Waypoint

1) Select the User Waypoint Information Page and press MENU to display the User Waypoint Information Page Options or press the DELETE Softkey.
2) Turn the large FMS Knob to select ‘Delete User Waypoint’ and press the ENT Key. The message ‘Would you like to delete the user waypoint’ is displayed. With ‘YES’ highlighted, press the ENT Key.
Create User Waypoints from the Navigation Map Page

1) With the Navigation Map Page displayed, push the Joystick to activate the panning function. The target pointer is displayed at the present aircraft position.

2) Move the Joystick to place the pointer at the desired position.

3) Press the ENT Key. The User Waypoint Information Page is now displayed with the captured position.

4) Turn the FMS Knobs to select a waypoint name.

5) Press the ENT Key to accept the selected name. The first reference waypoint field is highlighted.

6) If desired, turn the FMS Knobs to enter the identifier of the reference waypoint and the radial and distance to the reference waypoint. Press the ENT Key to accept.

7) Press the FMS Knob to remove the flashing cursor.

8.2 VIEWING THE ACTIVE FLIGHT PLAN

Press the FPL Key.

8.3 ACTIVATE A STORED FLIGHT PLAN

1) Press the FPL Key and turn the small FMS Knob to display the Flight Plan Catalog Page.

2) Press the FMS Knob to activate the cursor.

3) Turn the large FMS Knob to highlight the desired flight plan and press the ACTIVE Softkey.

4) With ‘OK’ highlighted, press the ENT Key to activate the flight plan. To cancel the flight plan activation, turn the large FMS Knob to highlight ‘CANCEL’ and press the ENT Key.
8.4 ACTIVATE A FLIGHT PLAN LEG

1) From the Active Flight Plan Page, press the FMS Knob to activate the cursor and turn the large FMS Knob to highlight the desired destination waypoint.

2) Press the ACT LEG Softkey (using MFD only).
   OR

3) Press the MENU Key, select the ‘Activate Leg’ option from the page menu and press the ENT Key. This step must be used when activating a leg from the PFD.

4) With ‘Activate’ highlighted, press the ENT Key.

Figure 8-5 Activate Flight Plan Leg Confirmation

8.5 STOP NAVIGATING A FLIGHT PLAN

1) Press the FPL Key to display the Active Flight Plan Page.

2) Press the MENU Key to display the Page Menu window.

3) Turn the large FMS Knob to highlight ‘Delete Flight Plan’ and press the ENT Key. With ‘OK’ highlighted, press the ENT Key to deactivate the flight plan. This will not delete the stored flight plan, only the active flight plan.

Figure 8-6 Delete Flight Plan

8.6 INVERT ACTIVE FLIGHT PLAN

1) From the Active Flight Plan Page, press the MENU Key to display the Page Menu.

2) Turn the large FMS Knob to highlight ‘Invert Flight Plan’ and press the ENT Key. The original flight plan remains intact in its flight plan catalog storage location.

3) With ‘OK’ highlighted, press the ENT Key to invert the flight plan.

Figure 8-7 Delete Flight Plan Confirmation

Figure 8-8 Invert Flight Plan

Figure 8-9 Invert Flight Plan Confirmation
8.7 CREATE A NEW FLIGHT PLAN

Create a New Flight Plan Using the MFD

1) Press the FPL Key and turn the small FMS Knob to display the Flight Plan Catalog Page.
2) Press the NEW Softkey to display a blank flight plan page for the first empty storage location.
3) Turn the small FMS Knob to display the Waypoint Information Window.
4) Turn the small FMS Knob to the right enter the first character of the identifier of the departure waypoint. Turning the knob to the left will access the FPL, NRST, and RECENT waypoint list.
5) Turn the large FMS Knob to move the cursor to the next character field. Repeat steps 4 and 5 until the desired identifier has been entered.
6) Press the ENT Key.
7) Repeat step number 3, 4, and 5 to enter the identifier for each additional flight plan waypoint.
8) When all waypoints have been entered, press the FMS Knob to return to the Flight Plan Catalog Page. The new flight plan is now in the list.

Create a New Flight Plan Using the PFD

- **NOTE:** A flight plan cannot be entered using the PFD if another flight plan is active.

- **NOTE:** After the first leg is entered (using the PFD only), it is immediately activated.

1) Press the FPL Key, then press the FMS Knob to activate the cursor.
2) Turn the small FMS Knob to enter the first letter of the destination waypoint identifier. Turn the large FMS Knob to the right to move the cursor to the next character position.
3) Repeat step 2 to spell out the rest of the waypoint identifier.
4) Press the ENT Key and the cursor is now ready for entering of the next flight plan waypoint.
5) Repeat steps 2 through 4 to enter the identifier for each additional flight plan waypoint.
6) Once all waypoints have been entered, press the FMS Knob remove the cursor. The new flight plan is now active.
8.8 LOAD A DEPARTURE

See the Procedures section for a discussion on loading and activating departure procedures.

8.9 LOAD AN ARRIVAL

See the Procedures section for a discussion on loading and activating arrival procedures.

8.10 LOAD AN APPROACH

See the Procedures section for a discussion on loading and activating approach procedures.

8.11 REMOVE A DEPARTURE, ARRIVAL, OR APPROACH FROM A FLIGHT PLAN

1) With the Active or Stored Flight Plan Page displayed, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to highlight the title for the approach, departure or arrival to be deleted. Titles appear in white directly above the procedure’s waypoints.

3) Press the CLR Key to display a confirmation window. With ‘OK’ highlighted, press the ENT Key to remove the selected procedure.

8.12 STORE A FLIGHT PLAN

1) After creating a flight plan on either the PFD or MFD, it may be saved by pressing the MENU Key.

2) Turn the large FMS Knob to highlight ‘Store Flight Plan’ and press the ENT Key.

3) With ‘OK’ highlighted, press the ENT Key to store the flight plan.

8.13 EDIT A STORED FLIGHT PLAN

1) Press the FPL Key and turn the small FMS Knob to display the Flight Plan Catalog Page.

2) Press the FMS Knob to activate the cursor.

3) Turn the large FMS Knob to highlight the desired flight plan and press the ENT Key.

4) Turn the large FMS Knob to place the cursor in the desired locations for entering changes.

5) Turn the FMS Knobs to make the desired changes, then press the ENT Key.

6) Press the FMS Knob to return to the Flight Plan Catalog Page.

8.14 DELETE A WAYPOINT FROM THE FLIGHT PLAN

1) With either the Active or Stored Flight Plan displayed, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to select the waypoint to be deleted.

3) Press the CLR Key to display a ‘REMOVE (Wpt Name)’ confirmation window.
4) With ‘OK’ highlighted, press the ENT Key to remove the waypoint. To cancel the delete request, turn the large FMS Knob to highlight ‘CANCEL’ and press the ENT Key.

5) Once all changes have been made, press the FMS Knob to remove the cursor.

8.15 INVERT AND ACTIVATE A STORED FLIGHT PLAN

1) From the Flight Plan Catalog Page, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to highlight the desired flight plan.

3) Press the INVERT Softkey. ‘Invert and activate stored flight plan?’ will be displayed.

4) With ‘OK’ highlighted, press the ENT Key. The selected flight plan is now inverted and activated. The original flight plan remains intact in its flight plan catalog storage location.

8.16 COPY A FLIGHT PLAN

1) From the Flight Plan Catalog press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to highlight the flight plan to be copied.

3) Press the COPY Softkey.

4) A ‘Copy to flight plan #?’ confirmation window is displayed. With ‘OK’ highlighted, press the ENT Key to copy the flight plan. To cancel, turn the large FMS Knob to highlight ‘CANCEL’ and press the ENT Key.

8.17 DELETE A FLIGHT PLAN

1) From the Flight Plan Catalog Page, press the FMS Knob to activate the cursor.

2) Turn the large FMS Knob to highlight the flight plan to be deleted.

3) Press the DELETE Softkey.

4) A ‘Delete flight plan #?’ confirmation window is displayed. With ‘OK’ highlighted, press the ENT Key to delete the flight plan. To cancel, turn the large FMS Knob to highlight ‘CANCEL’ and press the ENT Key.

8.18 GRAPHICAL FLIGHT PLAN CREATION

1) Press the FPL Key to display the Active Flight Plan Page.

2) Press the Joystick to activate the map pointer. Use the Joystick to move the pointer to the desired point on the map to be inserted as a waypoint in the flight plan.

3) Press the LD WPT Softkey. The selected waypoint will be inserted at the end of the flight plan. The default user waypoint naming is USR000, USR001, USR002 and so on.

4) If the selected waypoint is to be placed elsewhere in the flight plan, press the FMS Knob to activate the cursor. Waypoints are inserted ABOVE the cursor.

5) After placing the cursor at the desired point in the list of waypoints, press the LD WPT Softkey.

6) To change the user waypoint name, follow the procedure for modifying a user waypoint.
8.19 TRIP PLANNING

1) Turn the large FMS Knob to select the ‘AUX’ page group.

2) Turn the small FMS Knob to select the first rectangular page icon.

3) The current page mode is displayed at the top of the page: ‘AUTOMATIC’ or ‘MANUAL’. To change the page mode, press the AUTO or MANUAL Softkey.

4) For Direct-to planning, press the WPTS Softkey and verify that the starting waypoint field indicates ‘P.POS’ (present position). If necessary, press the MENU Key and select ‘Set WPT to Present Position’ to display ‘P.POS’. Press the ENT Key and the flashing cursor moves to the ending waypoint field. Turn the FMS Knobs to enter the identifier of the ending waypoint and press the ENT Key to accept the waypoint. OR,

5) For point-to-point planning, turn the FMS Knobs to enter the identifier of the starting waypoint. Once the waypoints identifier is entered, press the ENT Key to accept the waypoint. The flashing cursor moves to the ending waypoint. Again, turn the FMS Knobs to enter the identifier of the ending waypoint and press the ENT Key to accept the waypoint. OR,

6) For flight plan leg planning, press the FPL Softkey (at the bottom of the display) and turn the small FMS Knob to select the desired flight plan (already stored in memory), by number. Turn the large FMS Knob to highlight the ‘LEG’ field and turn the small FMS Knob to select the desired leg of the flight plan, or select ‘CUM’ to apply trip planning calculations to the entire flight plan. Selecting ‘FPL 00’ will display an active flight plan. If an active flight plan is selected, ‘REM’ will be an available option to display planning data for the remainder of the flight plan.

NOTE: The Page Mode must be set to MANUAL to perform the following steps.

7) Turn the large FMS Knob to highlight the departure time (DEP TIME) field.

NOTE: The departure time on the Trip Planning Page is used for preflight planning. Refer to the Utility Page for the actual flight departure time.

8) Turn the FMS Knobs to enter the departure time. Press the ENT Key when finished. (Departure time may be entered in local or UTC time, depending upon system settings).
9) Turn the **FMS** Knobs to enter the fuel flow. Press the **ENT** Key when finished. Note that in automatic page mode, fuel flow is provided by the system.

10) The flashing cursor moves to the fuel on board field. Turn the **FMS** Knobs to modify the fuel on board. Press the **ENT** Key when finished. In ‘AUTOMATIC’ mode, fuel onboard is provided by the entry made in ‘GAL REM’ on the EIS System Page.

11) The flashing cursor moves to the calibrated airspeed field. Turn the **FMS** Knobs to enter a calibrated airspeed. Press the **ENT** Key when finished.
SECTION 9: PROCEDURES

9.1 ARRIVALS AND DEPARTURES

Load and Activate a Departure Procedure

1) Press the PROC Key.

2) Turn the large FMS Knob to highlight ‘SELECT DEPARTURE’.

3) Press the ENT Key.

4) If a flight plan is active, the departure airport will be displayed as the default. A list of available departures is also displayed. If no flight plan is active, use the FMS Knobs to enter the identifier of the desired airport. Press the ENT Key.

5) Turn the large FMS Knob to highlight the Departure field. Turn the small FMS Knob to display a list of available departures.

6) Turn either FMS Knob to select the desired departure and press the ENT Key.

7) A list of runways may be displayed for the departure. Turn either FMS Knob to select the desired runway and press the ENT Key.

8) A list of available transitions is displayed for the departure. Turn either FMS Knob to highlight the desired transition waypoint and press the ENT Key.

9) With ‘LOAD?’ highlighted, press the ENT Key. The departure will be active when the flight plan is active.

Load and Activate An Arrival Procedure

1) Press the PROC Key.

2) Turn the large FMS Knob to highlight ‘SELECT ARRIVAL’.

3) Press the ENT Key.

4) If a flight plan is active, the destination airport will be displayed as the default. A list of available arrivals is also displayed. If no flight plan is active, use the FMS Knobs to enter the identifier of the desired airport. Press the ENT Key.

5) Turn the large FMS Knob to highlight the Arrival field. Turn the small FMS Knob to display a list of available arrivals.

6) Turn either FMS Knob to select the desired arrival and press the ENT Key.

7) A list of available transitions is displayed for the arrival. Turn either FMS Knob to highlight the desired transition waypoint and press the ENT Key.
SECTION 9 – PROCEDURES

7) A second window is displayed listing available transitions for the arrival. Turn either FMS Knob to highlight the desired transition waypoint and press the ENT Key.

![Figure 9-4 Select Arrival Transition](image)

8) A third window is displayed listing the available runways. Turn either FMS Knob to select the desired runway and press the ENT Key.

![Figure 9-5 Select Arrival Runway](image)

9) With ‘LOAD?’ highlighted, press the ENT Key. If a flight plan is active, the selected arrival procedure is inserted after the destination airport and becomes part of the active flight plan. If no flight plan is active when the arrival is loaded, the arrival procedure will become the active flight plan.

9.2 APPROACHES

Not all approaches in the database are approved for GPS use. When selecting an approach, a “GPS” designation to the right of the procedure name indicates the procedure can be flown using the GPS receiver. Some procedures will not have this designation, meaning the GPS receiver can be used for supplemental navigation guidance only. If the GPS receiver cannot be used for primary guidance, the appropriate navigation receiver must be used for the selected approach (e.g., VOR or ILS). The final course segment of ILS approaches, for example, must be flown by tuning the Nav receiver to the proper frequency and selecting that Nav receiver on the CDI.

Load and/or Activate an Approach Procedure

1) Press the PROC Key.

2) Turn the large FMS Knob to highlight ‘SELECT APPROACH’.

3) Press the ENT Key.

4) If a flight plan is active, the destination airport will be displayed as the default. A list of available approaches is also displayed. If no flight plan is active, use the FMS Knobs to enter the identifier of the desired airport. Press the ENT Key.

5) Turn the large FMS Knob to highlight the Approach field. Turn the small FMS Knob to display a list of available approaches.

![Figure 9-6 Selecting an Approach Procedure](image)
6) Turn either FMS Knob to highlight the desired approach. Press the ENT Key.

7) The cursor will move to the TRANSITIONS field. Turn the large FMS Knob to highlight the desired transition waypoint and press the ENT Key. (The “Vectors” option assumes vectors will be received to the final course segment of the approach and will provide navigation guidance relative to the final approach course.)

---

**Activate An Approach in the Active Flight Plan**

1) With the Navigation Map Page displayed, press the PROC Key.

2) Turn the large FMS Knob to highlight ‘ACTIVATE APPROACH’.

3) Press the ENT Key. The approach procedure is now active.

---

8) Turn the large FMS Knob to highlight ‘Activate?’ and press the ENT Key to activate the approach. Selecting ‘Load?’ will add the procedure to the flight plan without immediately using it for navigation guidance.
SECTION 10: HAZARD AVOIDANCE

10.1 CUSTOMIZING THE HAZARD DISPLAYS ON THE NAVIGATION MAP

1) With the Navigation Map Page displayed, press the **MENU** Key to display the Navigation Map Page Menu. The cursor flashes on the ‘Map Setup’ option.

2) Press the **ENT** Key. The Map Setup Menu is displayed. Turn the small **FMS** Knob to select the ‘Weather’ group (Figure 10-2) to customize the display of weather features. Select ‘Traffic’ to customize the display of traffic.

3) Press the small **FMS** Knob to return to the Navigation Map Page.

10.2 STORMSCOPE® (OPTIONAL)

**WARNING:** The Stormscope system is not intended to be used for hazardous thunderstorm penetration. Weather information on the G1000 MFD is approved for weather avoidance only. Refer to the WX-500 Pilot’s Guide for detailed operation.

Displaying Stormscope Lightning Data on the Navigation Map Page

1) Press the **MAP** Softkey.

2) Press the **STRMSCP** Softkey. Press the **STRMSCP** Softkey again to remove Stormscope Lightning Data from the Navigation Map Page.

<table>
<thead>
<tr>
<th>Lightning Age</th>
<th>Symbol</th>
</tr>
</thead>
<tbody>
<tr>
<td>Strike is less than 6 seconds old</td>
<td>⚡</td>
</tr>
<tr>
<td>Strike is between 6 and 60 seconds old</td>
<td>⚡</td>
</tr>
<tr>
<td>Strike is between 1 and 2 minutes old</td>
<td>⚡</td>
</tr>
<tr>
<td>Strike is between 2 and 3 minutes old</td>
<td>⚡</td>
</tr>
</tbody>
</table>

Figure 10-4  In-Flight Navigation Map Page Displaying Stormscope Lightning Data
At a map range of less than 25 nm, Stormscope lightning data is not displayed, but can still be present. The presence of Stormscope lightning data is indicated by the annunciation ‘LTNG < 25 nm’ in the upper right corner.

**Stormscope Page**

1) Turn the large FMS Knob until the Map Page group is selected.

2) Turn the small FMS Knob until the Stormscope Page is selected. This is the third rectangular page icon.

**Change the Stormscope lightning mode between ‘cell’ and ‘strike’:**

1) Select the Stormscope Page.

2) Press the MODE Softkey. The CELL and STRIKE Softkeys are displayed. Press the CELL Softkey to display ‘CELL’ data or press the STRIKE Softkey to display ‘STRIKE’ data. ‘CELL’ or ‘STRIKE’ is displayed in the mode box located in the upper left corner of the Stormscope Page.

**Clear Stormscope lightning data from the Navigation Map Page:**

1) Press the MENU Key (with the Navigation Map Page displayed).

2) Turn either FMS Knob to highlight the ‘Clear Lightning Data’ field and press the ENT Key.

**NOTE: If heading input is lost, strikes and/or cells must be cleared manually after the execution of each turn. This is to ensure that the strike and/or cell positions are depicted accurately in relation to the nose of the aircraft.**

**NOTE: “Cell mode” uses a clustering program to identify clusters of electrical activity that indicate cells.**
Change the viewing mode between 360° and 120°:

1) Select the Stormscope Page.
2) Press the VIEW Softkey. The 360 and ARC Softkeys are displayed. Press the 360 Softkey to display a 360° viewing area or press the ARC Softkey to display a 120° viewing area.

Press the CLEAR Softkey to remove all Stormscope lightning data from the display.

10.3 XM WEATHER (SERVICE OPTIONAL)

WARNING: XM Weather is not intended to be used for hazardous weather penetration. Weather information provided by XM Radio Service is approved only for weather avoidance, not penetration.

1) From the Navigation Map Page, press the MAP Softkey.
2) Press the NEXRAD or XM LTNG Softkey to display the desired weather. Press the applicable softkey again to remove weather data from the Navigation Map Page.

NOTE: Weather is not displayed on the Navigation Map Page at zoom levels less than 10 nm.

Displaying METAR and TAF information on the Airport Information Page

1) Turn the large FMS Knob to select the WPT Page Group.
2) Turn the small FMS Knob to select the Airport Information Page.
3) Press the WX Softkey to display METAR and TAF text (METAR and TAF information is updated every 12 minutes).

Figure 10-7 Navigation Map Page Displaying NEXRAD Weather

Figure 10-8 METAR and TAF Text Displayed on the Airport (Weather) Information Page
SECTION 10 – HAZARD AVOIDANCE

Displaying Weather on the Weather Data Link Page

Select the Weather Data Link Page:
1) Turn the large FMS Knob to select the Map Page Group.
2) Turn the small FMS Knob to select the fourth rectangular page icon.
3) Press the available softkeys to select the desired XM weather product.
4) Press the LEGEND Softkey to view the legends for the selected products. If necessary, turn either FMS Knob to scroll through the list. Press the small FMS Knob or the ENT Key to return to the map.

---

**NEXRAD** – Press the NEXRAD Softkey to show NEXRAD weather and radar coverage information. Areas where radar coverage is not available will be shown in grayish-purple. The display is updated every five minutes.

**ECHO TOP** – Press the ECHO TOP Softkey to show the location, elevation, and direction the highest radar echo. This may not indicate the top of a storm or clouds, only the highest radar return echo. ECHO TOPS cannot be displayed along with NEXRAD and CLOUD TOPS. When ECHO TOPS is activated, NEXRAD and CLOUD TOPS are removed. Refer to the Legend for a description of the ECHO TOPS coding. The display is updated every 7.5 minutes.

**CLD TOP** – Press the CLD TOP Softkey to show the cloud top altitude determined from satellite imagery. The display is updated every 15 minutes.

**LTNG** – Pressing the LTNG Softkey shows the location of cloud-to-ground lightning strikes. The display is updated every five minutes.

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**NOTE:** Strikes depicted represent cloud to ground strikes within a 2 kilometer radius of the actual strike location. Therefore, the exact location of the strike is not displayed.
CELL MOV – Pressing the CELL MOV Softkey shows storm cell movement by displaying an arrow pointing in the direction of predicted movement. The display is updated every 12 minutes.

SIG/AIR – Pressing the SIG/AIR Softkey shows SIGMET and AIRMET information. The display is updated every 12 minutes.

METAR – Press the METAR Softkey to graphically display METARs. METARS are shown as colored flags at airports providing METAR reports. The display is updated every 12 minutes.

MORE WX – Press the MORE WX Softkey to display the following group of softkeys for additional weather control:

SFC – Pressing the SFC Softkey for Surface Analysis shows current or forecast conditions. Forecasts are available for intervals of Current, 12, 24, 36, and 48 hours. Press the softkey corresponding to the desired forecast. The closest city forecast information is displayed in the legend. The display is updated every 12 minutes.

FRZ LVL – Press the FRZ LVL Softkey to display contour lines for freezing levels. The display is updated every 12 minutes.

WIND – Press the WIND Softkey to show wind speed and direction at a selected altitude from the ground up to 42,000 feet in 3,000 foot increments. After pressing the WIND Softkey, press the softkey corresponding to the desired winds aloft altitude. The display is updated every 12 minutes.

COUNTY – Pressing the COUNTY Softkey provides specific public awareness and protection weather warnings for Tornado, Severe Thunderstorm, and Flood conditions provided by the National Weather Service (NWS). The display is updated every 5 minutes.

CYCLONE – Pressing the CYCLONE Softkey shows the current location of cyclones (hurricanes and tropical storms) and their projected track at various time intervals. The update rate is every 12 minutes.

Map Panning Information – Weather Data Link Page

1) Push in the Joystick to display the panning arrow.

2) Move the Joystick to place the panning arrow on AIRMETs, TFRs, METARs, or SIGMETs. Press the ENT Key to display pertinent information for the selected product.

Note that pressing the ENT Key when panning over an AIRMET or a SIGMET displays an information box that displays the text of the report. Panning over an airport with METAR information does not display more information but allows the user to press the ENT Key and select that Airport’s Information Page to display the text of the report. Pressing the ENT Key when panning over a TFR displays TFR specific information.

Weather Products & Symbols

Figure 10-10 depicts the symbol for each weather product (read from left to right). When a weather product is active, the product symbol is displayed in the lower right of the screen.

- NEXRAD
- Cloud Top/Echo Top
- XM Lightning
- Cell Movement
- SIGMETs/AIRMETs
- METARs
- City Forecast
- Surface Analysis
SECTION 10 – HAZARD AVOIDANCE

- Freezing Levels
- Winds Aloft
- County Warnings
- Cyclone Warnings

Echo Top (Cloud Top and Echo Top Mutually Exclusive)

The XM Information Page in the AUX Page Group will display the weather products available for the current subscription. A green box by the weather product means that it is available.

From within the AUX - XM INFORMATION Page, the pilot may switch to the AUX - XM RADIO Page by pressing the RADIO Softkey. Alternatively, the pilot may switch to the AUX - XM INFORMATION Page from the AUX - XM RADIO Page by pressing the INFO Softkey.

**NOTE:** The LOCK Softkey on the AUX - XM INFORMATION Page is used to save the GDL 69(A) activation data when the XM services are initially set up. It is not used during normal operation of the GDL 69(A), but it should have no adverse effects if inadvertently selected during flight. Refer to the GDL 69/69A XM Satellite Radio Activation Instructions (190-00355-04, Rev E or later) for further information.

### Weather Product Age

The age for each of the enabled products is displayed on the right side of the display. Times are based on GMT time when the data was assembled on the ground, not the time the data was received by the XM receiver. When the age of a weather product has exceeded half of the expiration time, the product time will change from light blue to amber in color.

<table>
<thead>
<tr>
<th>Weather Product</th>
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<tbody>
<tr>
<td>SIGMETs/AIRMETs</td>
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</tr>
<tr>
<td>City Forecasts</td>
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<td>Cyclone Warnings</td>
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<td>60</td>
</tr>
<tr>
<td>TAFs</td>
<td>60</td>
</tr>
<tr>
<td>Clouds Tops</td>
<td>60</td>
</tr>
</tbody>
</table>
10.4 TRAFFIC INFORMATION SERVICE (TIS)

**NOTE:** Traffic Information Service (TIS) is only available when the aircraft is within the service volume of a TIS capable terminal radar site.

**NOTE:** If the G1000 is configured to use a Traffic Advisory System (TAS), TIS will not be available for use.

Displaying Traffic on the Traffic Map Page

1) Turn the large FMS Knob to select the Map Page Group.

2) Turn the small FMS Knob to select the second rectangular page icon.

3) Press the OPERATE Softkey to begin displaying traffic. ‘TIS OPERATING’ is displayed in the Traffic mode field.

4) Press the STANDBY Softkey to place the system in the Standby mode. ‘TIS STANDBY’ will be displayed in the Traffic mode field.

5) Rotate the Joystick clockwise to display a larger area or rotate counter-clockwise to display a smaller area.

If data is not received for a period longer than 6 seconds, the age of the present data will be displayed in the lower left of the screen along with the annunciation that the system has entered Coast mode. The system will maintain the traffic display (up to 60 seconds) until the next data reception. If no data is received after 60 seconds, traffic will be removed from the display.

Displaying Traffic on the Navigation Map

1) Ensure TIS is operating. With the Navigation Map displayed, press the MAP Softkey.

2) Press the TRAFFIC Softkey. Traffic will now be displayed on the map.

TIS Voice Alert

When a Traffic Advisory (TA) is displayed, an voice alert “Traffic” is given.

“Traffic Not Available” will be heard whenever TIS service becomes unavailable. This alert can be muted by pressing the TNA MUTE Softkey. ‘TNA MUTE ON’ will be displayed in the upper left of the display.
10.5 TRAFFIC ADVISORY SYSTEM (TAS) (OPTIONAL)

Refer to the Honeywell KTA 870 Pilot’s Guide for a detailed discussion of the KTA 870 system.

System Self Test

1) Set the range to 2/6 nm.
2) Press the STANDBY Softkey.
3) Press the TEST Softkey.
4) Self test will take approximately eight seconds to complete. When completed successfully, traffic symbols will be displayed as shown in Figure 10-12 and a voice alert “TAS System Test OK” will be heard. If the self test fails, the system will revert to Standby mode and a voice alert “TAS System Test Fail” will be heard.

Displaying Traffic on the Traffic Map Page

1) Turn the large FMS Knob to select the Map Page Group.
2) Turn the small FMS Knob to select the second rectangular page icon.
3) Press the OPERATE Softkey to begin displaying traffic. ‘TAS OPERATING’ is displayed in the Traffic mode field.
4) Press the ALT MODE Softkey to change the altitude volume. Select the desired altitude volume by pressing the BELOW, NORMAL, ABOVE, or UNREST (unrestricted) Softkey. The selection is displayed in the Altitude mode field.
5) Press the STANDBY Softkey to place the system in the Standby mode. ‘TAS STANDBY’ will be displayed in the Traffic mode field.
6) Rotate the Joystick clockwise to display a larger area or rotate counter-clockwise to display a smaller area.
Displaying Traffic on the Navigation Map

1) Ensure the KTA 870 system is operating. With the Navigation Map displayed, press the MAP Softkey.

2) Press the TRAFFIC Softkey. Traffic will now be displayed on the map.

![Traffic Advisory](image)

Figure 10-14 TAS Traffic on Navigation Map

10.6 TERRAIN AND OBSTACLE PROXIMITY

NOTE: Terrain data is not displayed when the aircraft latitude is greater than 75 degrees north or 60 degrees south.

Displaying Terrain and Obstacles on the Terrain Proximity Page

1) Turn the large FMS Knob to select the Map Page Group.

2) Turn the small FMS Knob to select the last rectangular page icon.

3) If desired, press the VIEW Softkey to access the ARC and 360 Softkeys. When the ARC Softkey is pressed, a radar-like 120° view is displayed. Press the 360 Softkey to return to the 360° default display.

4) Rotate the Joystick clockwise to display a larger area or rotate counter-clockwise to display a smaller area.

![Terrain Proximity Page](image)

Figure 10-15 Terrain Proximity Page

<table>
<thead>
<tr>
<th>Color</th>
<th>Indication</th>
</tr>
</thead>
<tbody>
<tr>
<td>RED</td>
<td>Terrain/Obstacle at or within 100’ below or above aircraft altitude.</td>
</tr>
<tr>
<td>YELLOW</td>
<td>Terrain/Obstacle between 100’ and 1000’ below current aircraft altitude.</td>
</tr>
</tbody>
</table>
SECTION 10 – HAZARD AVOIDANCE

<table>
<thead>
<tr>
<th>Unlighted Obstacle (Height is less than 1000’ AGL)</th>
<th>Lighted Obstacle (Height is less than 1000’ AGL)</th>
<th>Unlighted Obstacle (Height is greater than 1000’ AGL)</th>
<th>Lighted Obstacle (Height is greater than 1000’ AGL)</th>
</tr>
</thead>
<tbody>
<tr>
<td>![Unlighted Obstacle Symbol]</td>
<td>![Lighted Obstacle Symbol]</td>
<td>![Unlighted Obstacle Symbol]</td>
<td>![Lighted Obstacle Symbol]</td>
</tr>
</tbody>
</table>

Figure 10-16 Obstacle Symbols

**Displaying Terrain and Obstacles on the Navigation Map**

1) With the Navigation Map displayed, press the MAP Softkey.
2) Press the TERRAIN Softkey. Terrain and obstacle proximity will now be displayed on the map.

**10.7 TERRAIN AWARENESS & WARNING SYSTEM (TAWS) DISPLAY (OPTIONAL)**

- **NOTE:** Terrain data is not displayed when the aircraft latitude is greater than 75 degrees north or 60 degrees south.
- **NOTE:** TAWS operation is only available when the G1000 is configured for a TAWS-B installation.

**Displaying Terrain on the TAWS Page**

1) Turn the large FMS Knob to select the Map Page Group.
2) Turn the small FMS Knob to select the TAWS Page.
3) If desired, press the VIEW Softkey to access the ARC and 360 Softkeys. When the ARC Softkey is pressed, a radar-like 120° view is displayed. Press the 360 Softkey to return to the 360° default display.
4) Rotate the Joystick clockwise to display a larger area or rotate counter-clockwise to display a smaller area.
Figure 10-17  TAWS Page (360˚ View)

Figure 10-18  TAWS Page (ARC View)

<table>
<thead>
<tr>
<th>Color</th>
<th>Terrain/Obstacle Location</th>
</tr>
</thead>
<tbody>
<tr>
<td>Red</td>
<td>Terrain/Obstacle at or within 100’ below or above aircraft altitude.</td>
</tr>
<tr>
<td>Yellow</td>
<td>Terrain/Obstacle between 100’ and 1000’ below current aircraft altitude.</td>
</tr>
<tr>
<td>Black</td>
<td>Terrain/Obstacle is more than 1000’ below aircraft altitude.</td>
</tr>
</tbody>
</table>
Enable/Disable Aviation Data

1) While the TAWS Page is displayed, press the MENU Key.
2) Turn the small FMS Knob to select “Show (or Hide) Aviation Data”.
3) Press the ENT Key.

Figure 10-19 TAWS Page Menu

Manual System Test

A system test is automatically performed at power-up. After successful completion of the test, “TAWS System Test, OK” will be heard.

The system test may also be initiated manually, but only when the aircraft is on the ground. To manually verify proper operation of the aural and visual annunciations of the system, perform the following steps.

1) While the TAWS Page is displayed, press the MENU Key.
2) Turn the small FMS Knob to select ‘Test TAWS’.
3) Press the ENT Key. During the test ‘TAWS TEST’ is displayed in the center of the TAWS Page. When all is in working order, “TAWS System Test, OK” will be heard.

Forward Looking Terrain Avoidance (FLTA)

The Forward Looking Terrain Avoidance alert is composed of two sub-functions:

**Reduced Required Terrain Clearance (RTC) and Reduced Required Obstacle Clearance (ROC)**

This provides alerts when the aircraft flight path is above terrain and/or obstacles, yet is projected to come within minimum clearance values outlined in the following table. When an RTC or ROC alert is issued, a potential impact point is displayed on the TAWS Page as a yellow or red ‘X’.

**Imminent Terrain Impact (ITI) and Imminent Obstacle Impact (IOI)**

This provides alerts when the aircraft is below the elevation of terrain in the aircraft’s projected path. ITI and IOI alerts are accompanied by a potential impact point displayed on the TAWS Page as a yellow or red ‘X’. The alert is given when the projected vertical flight path is
calculated to come within minimum clearance altitudes in the following table.

<table>
<thead>
<tr>
<th>Phase of Flight</th>
<th>Level Flight</th>
<th>Descending</th>
</tr>
</thead>
<tbody>
<tr>
<td>Enroute</td>
<td>700 ft.</td>
<td>500 ft.</td>
</tr>
<tr>
<td>Terminal</td>
<td>350 ft.</td>
<td>300 ft.</td>
</tr>
<tr>
<td>Approach</td>
<td>150 ft.</td>
<td>100 ft.</td>
</tr>
<tr>
<td>Departure</td>
<td>100 ft.</td>
<td>100 ft.</td>
</tr>
</tbody>
</table>

During the final approach phase of flight, RTC/ROC/ITI/IOI alerts are automatically inhibited when the aircraft is below 200 feet AGL while within 0.5 nm of the approach runway or is below 125 feet AGL while within 1 nm of the runway.

**Premature Descent Alert (PDA)**

A Premature Descent Alert is issued when the system detects that the aircraft is significantly below the normal approach path to a runway. The PDA alert mode functions only during descent to land.

PDA alerting begins when the aircraft is within 15 nm of the destination airport and ends when the aircraft is either 0.5 nm from the runway threshold OR is at an altitude of 125 feet AGL while within 1 nm of the threshold. During the final descent, algorithms will set a threshold for alerting based on speed, distance, and other parameters.

**Excessive Descent Rate Alert (EDR)**

The purpose of the Excessive Descent Rate alert is to provide suitable alerts when the aircraft is determined to be closing (descending) upon terrain at an excessive speed. Figure 10-21 displays the correlation between height above terrain and descent rate, resulting in the two EDR alerts. EDR alerts have two levels of severity, caution (sink rate) and warning (pull-up).

**Negative Climb Rate After Takeoff Alert (NCR)**

The purpose of the Negative Climb Rate After Takeoff alert is to provide suitable alerts to the pilot when the system determines that the aircraft is losing altitude (closing upon terrain) after takeoff. The aural message “Don’t Sink” is given for NCR alerts, accompanied by an annunciation and a pop-up terrain alert on the display.
**“Five-Hundred” Aural Alert**

The purpose of the aural alert message “Five-hundred” is to provide an advisory alert to the air crew that the aircraft is five-hundred feet above terrain. When the aircraft descends within 500 feet of terrain, the aural message “Five-hundred” is heard. There are no display annunciations or pop-up alerts that accompany the aural message.

**Displaying Terrain and Obstacles on the Navigation Map**

1) With the Navigation Map displayed, press the MAP Softkey.

2) Press the TERRAIN Softkey. Terrain and obstacle proximity will now be displayed on the map.

---

**Pop-up Alerts**

When a terrain or obstacle alert is issued, a pop-up window is displayed on the MFD with the appropriate alert.

---

Unlighted Obstacle (Height is less than 1000’ AGL) | Lighted Obstacle (Height is less than 1000’ AGL) | Unlighted Obstacle (Height is greater than 1000’ AGL) | Lighted Obstacle (Height is greater than 1000’ AGL) | Potential Impact Points

---

| ![Symbol](image1) | ![Symbol](image2) | ![Symbol](image3) | ![Symbol](image4) | ![Symbol](image5) |

---

**Figure 10-22 TAWS Symbols**

---

Press the **ENT** Key to display the TAWS Page, or press the **CLR** Key to remain on the existing page.
# TAWS Alerts Summary

The following table shows the possible TAWS alert types with corresponding annunciations and aural messages.

<table>
<thead>
<tr>
<th>Alert Type</th>
<th>PFD/MFD TAWS Page Annunciation</th>
<th>MFD Pop-Up Alert</th>
<th>Aural Message</th>
</tr>
</thead>
<tbody>
<tr>
<td>Excessive Descent Rate Warning (EDR)</td>
<td>PULL UP</td>
<td>PULL-UP</td>
<td>“Pull Up”</td>
</tr>
<tr>
<td>Reduced Required Terrain Clearance Warning (RTC)</td>
<td>PULL UP</td>
<td>TERRAIN - PULL-UP</td>
<td>“Terrain, Terrain; Pull Up, Pull Up”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Terrain Ahead, Pull Up; Terrain Ahead, Pull Up”</td>
</tr>
<tr>
<td>Imminent Terrain Impact Warning (ITI)</td>
<td>PULL UP</td>
<td>TERRAIN AHEAD - PULL-UP</td>
<td>Terrain Ahead, Pull Up; Terrain Ahead, Pull Up</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Terrain, Terrain; Pull Up, Pull Up”</td>
</tr>
<tr>
<td>Reduced Required Obstacle Clearance Warning (ROC)</td>
<td>PULL UP</td>
<td>OBSTACLE - PULL-UP</td>
<td>“Obstacle, Obstacle; Pull Up, Pull Up”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Obstacle Ahead, Pull Up; Obstacle Ahead, Pull Up”</td>
</tr>
<tr>
<td>Imminent Obstacle Impact Warning (IOI)</td>
<td>PULL UP</td>
<td>OBSTACLE AHEAD - PULL-UP</td>
<td>“Obstacle Ahead, Pull Up; Obstacle Ahead, Pull Up”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Obstacle, Obstacle; Pull Up, Pull Up”</td>
</tr>
<tr>
<td>Reduced Required Terrain Clearance Caution (RTC)</td>
<td>TERRAIN</td>
<td>CAUTION - TERRAIN</td>
<td>“Caution, Terrain; Caution, Terrain”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Terrain Ahead; Terrain Ahead”</td>
</tr>
<tr>
<td>Imminent Terrain Impact Caution (ITI)</td>
<td>TERRAIN</td>
<td>TERRAIN AHEAD</td>
<td>“Terrain Ahead; Terrain Ahead”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Caution, Terrain; Caution, Terrain”</td>
</tr>
<tr>
<td>Reduced Required Obstacle Clearance Caution (ROC)</td>
<td>TERRAIN</td>
<td>CAUTION - OBSTACLE</td>
<td>“Caution, Obstacle; Caution, Obstacle”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Obstacle Ahead; Obstacle Ahead”</td>
</tr>
<tr>
<td>Imminent Obstacle Impact Caution (IOI)</td>
<td>TERRAIN</td>
<td>OBSTACLE AHEAD</td>
<td>“Obstacle Ahead; Obstacle Ahead”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Caution, Obstacle; Caution, Obstacle”</td>
</tr>
<tr>
<td>Premature Descent Alert Caution (PDA)</td>
<td>TERRAIN</td>
<td>TOO LOW - TERRAIN</td>
<td>“Too Low, Terrain”</td>
</tr>
<tr>
<td>Altitude Callout “500”</td>
<td>None</td>
<td>None</td>
<td>“Five-Hundred”</td>
</tr>
<tr>
<td>Excessive Descent Rate Caution (EDR)</td>
<td>TERRAIN</td>
<td>SINK RATE</td>
<td>“Sink Rate”</td>
</tr>
<tr>
<td>Negative Climb Rate Caution (NCR)</td>
<td>TERRAIN</td>
<td>DON’T SINK</td>
<td>“Don’t Sink”</td>
</tr>
<tr>
<td></td>
<td></td>
<td>or</td>
<td>“Too Low, Terrain”</td>
</tr>
</tbody>
</table>
The following system status annunciations may also be issued.

<table>
<thead>
<tr>
<th>Alert Type</th>
<th>PFD/MFD TAWS Page Annunciation</th>
<th>MFD Pop-Up Alert</th>
<th>Aural Message</th>
</tr>
</thead>
<tbody>
<tr>
<td>TAWS System Test Fail</td>
<td>TAWS FAIL</td>
<td>None</td>
<td>“TAWS System Failure”</td>
</tr>
<tr>
<td>TAWS Alerting is disabled</td>
<td>TAWS INHB</td>
<td>None</td>
<td>None</td>
</tr>
<tr>
<td>No GPS position or excessively degraded GPS signal</td>
<td>TAWS N/A</td>
<td>None</td>
<td>“TAWS Not Available”&lt;br&gt;“TAWS Available” will be heard when sufficient GPS signal is re-established.</td>
</tr>
<tr>
<td>System Test in progress</td>
<td>TAWS TEST</td>
<td>None</td>
<td>None</td>
</tr>
<tr>
<td>System Test pass</td>
<td>None</td>
<td>None</td>
<td>“TAWS System Test OK”</td>
</tr>
</tbody>
</table>

**Alert Annunciations**

![Alert Annunciation on the TAWS Page](image1)

![Alert Annunciation on the PFD](image2)

![Figure 10-25 TAWS Alert Annunciation on the PFD](image3)
SECTION 11: ABNORMAL OPERATION

11.1 REVERSIONARY MODE

Should a system detected failure occur in either display, the G1000 automatically enters reversionary mode. In reversionary mode, critical flight instrumentation is combined with engine instrumentation on the remaining display. Minimal navigation capability is available on the reversionary mode display.

Reversionary display mode can also be manually activated by the pilot if the system fails to detect a display problem. The reversionary mode is activated manually by pressing the red DISPLAY BACKUP button on the bottom of the audio panel (GMA 1347). Pressing the red DISPLAY BACKUP button again deactivates reversionary mode.

NOTE: The Cessna Pilot’s Operating Handbook (POH) always takes precedence over the information found in this section.
11.2 ABNORMAL COM OPERATION

When a COM tuning failure is detected by the system, the emergency frequency (121.500 MHz) is automatically loaded into the active frequency field of the COM radio for which the tuning failure was detected. In the event of a dual display failure, the emergency frequency (121.500 MHz) automatically becomes the active frequency to the pilot through the pilot headset.

11.3 UNUSUAL ATTITUDES

The PFD will ‘declutter’ when the aircraft enters an unusual attitude. Only the primary functions will be displayed in these situations.

The following information is removed from the PFD (and corresponding softkeys are disabled) when the aircraft experiences unusual attitudes:

- Traffic Annunciations
- AFCS Annunciations
- Flight director Command Bars
- Inset Map
- Temperatures
- DME Information Window
- Wind Data
- Selected Heading Box
- Selected Course Box
- Transponder Status Box
- System Time
- PFD Setup Menu
- Windows displayed in the lower right corner of the PFD:

  - Timer/References
  - Nearest Airports
  - Flight Plan
  - Messages
  - Procedures
  - DME Tuning
  - Barometric Minimum Descent Altitude Box
  - Glideslope, Glidepath, and Vertical Deviation Indicators
  - Altimeter Barometric Setting
  - Selected Altitude
  - VNV Target Altitude

Red extreme pitch warning chevrons pointing toward the horizon are displayed starting at 50 degrees above and 30 degrees below the horizon line.

11.4 STORMSCOPE OPERATION WITH LOSS OF HEADING INPUT

If heading is lost, strikes and/or cells must be cleared manually after the execution of each turn. This is to ensure that the strike and/or cell positions are depicted accurately in relation to the nose of the aircraft.

11.5 HAZARD DISPLAYS WITH LOSS OF GPS POSITION

If GPS position is lost, or becomes invalid, selected hazards being displayed on the Navigation Map Page will be removed until GPS position is again established. The icons in the lower right of the screen, indicating the selected functions for display, will show an ‘X’, as shown in Figure 11-3.
11.6 DEAD RECKONING

If at any time the G1000 detects an invalid GPS solution or is unable to calculate a GPS position, the system will automatically revert to Dead Reckoning (DR) mode. In DR mode, the G1000 uses its last-known position combined with continuously updated airspeed and heading data (when available) to calculate and display the aircraft’s current estimated position.

DR mode is indicated on the G1000 by the appearance of the letters ‘DR’ superimposed in yellow over the ‘own aircraft’ symbol as shown in Figure 11-4. In addition, ‘DR’ will be prominently displayed, also in yellow, on the HSI slightly above and to the right of the aircraft symbol on the CDI as shown in Figure 11-4. Lastly, but at the same time, a ‘GPS NAV LOST’ alert message will appear on the PFD.

Normal navigation using GPS/WAAS source data will resume automatically once a valid GPS solution is restored.

It is important to note that estimated navigation data supplied by the G1000 in DR mode may become increasingly unreliable and must not be used as a sole means of navigation. If while in DR mode airspeed and/or heading data is also lost or not available, the DR function may not be capable of accurately tracking your estimated position and, consequently, the system may display a path that is different than the actual movement of the aircraft. Estimated position information displayed by the G1000 through DR while there is no heading and/or airspeed data available should not for navigation.

DR mode is inherently less accurate than the standard GPS/WAAS mode due to the lack of satellite measurements needed to determine a position. Changes in wind speed and/or wind direction will compound the relative inaccuracy of DR mode. Because of this degraded accuracy, the crew must maintain position awareness using other navigation equipment until GPS-derived position data is restored.

As a result of operating in DR mode, all GPS-derived data will be computed based upon an estimated position and will be displayed as yellow text on the map to denote degraded navigation source information. This data includes the following:

- Navigation Status Box fields except Active Leg, TAS, and DTK
- GPS Bearing Pointer
- Wind data and pointers in the Wind Data Box on the PFD
- Track Bug
- All Bearing Pointer Distances
- Active Flight Plan distances, bearings, and ETE values

Also, while the G1000 is in DR mode, the autopilot will not couple to GPS, and both TAWS and Terrain Proximity will be disabled. Additionally, the accuracy of all nearest information (airports, airspaces, and waypoints) will be questionable. Finally, airspace alerts will continue to function, but with degraded accuracy.
SECTION 12: ANNUNCIATIONS & ALERTS

NOTE: The Cessna aircraft Pilot’s Operating Handbook (POH) supersedes information found in this document.

The G1000 Alerting System conveys alerts to the pilot using a combination of the following items:

- **Annunciation Window:** The Annunciation Window displays abbreviated annunciation text. Text color is based on alert levels described later in the Alert Levels Definitions section. The Annunciation Window is located to the right of the Altitude and Vertical Speed boxes on the display. All Cessna Nav III annunciations can be displayed simultaneously in the Annunciation Window. A white horizontal line separates annunciations that are acknowledged from annunciations that are not yet acknowledged. Higher priority annunciations are displayed towards the top of the window. Lower priority annunciations are displayed towards the bottom of the window.

- **Alerts Window:** The Alerts Window displays alert text messages. Up to 64 prioritized alert messages can be displayed in the Alerts Window. Pressing the ALERTS Softkey displays the Alerts Window. Pressing the ALERTS Softkey a second time removes the Alerts Window from the display. When the Alerts Window is displayed, the pilot can use the large FMS Knob to scroll through the alert message list.

- **Softkey Annunciation:** During certain alerts, the ALERTS Softkey may appear as a flashing annunciation to accompany an alert. The ALERTS Softkey assumes a new label consistent with the alert level (WARNING, CAUTION, or ADVISORY).

By pressing the softkey annunciation, the pilot acknowledges awareness of the alert. The softkey then returns to the previous ALERTS label. If alerts are still present, the ALERTS label will be displayed in inverse video (white background with black text). The pilot can press the ALERTS Softkey a second time to view alert text messages.

- **System Annunciations:** Typically, a large red ‘X’ appears in windows when a failure is detected in the LRU providing the information to the window. See the G1000 System Annunciations section for more information.

- **Audio Alerting System:** The G1000 system issues audio alert tones when specific system conditions are met. See the Alert Levels Definitions section for more information.

![Figure 12-1 G1000 Alerting System](image-url)
12.1 ALERT LEVEL DEFINITIONS

The G1000 Alerting System, as installed in Cessna Nav III aircraft, uses three alert levels.

- **WARNING:** This level of alert requires immediate pilot attention. A warning alert is annunciated in the Annunciation Window and is accompanied by a continuous aural tone. Text appearing in the Annunciation Window is RED. A warning alert is also accompanied by a flashing **WARNING** Softkey annunciation, as shown in Figure 12-2. Pressing the **WARNING** Softkey acknowledges the presence of the warning alert and stops the aural tone, if applicable.

- **CAUTION:** This level of alert indicates the existence of abnormal conditions on the aircraft that may require pilot intervention. A caution alert is annunciated in the Annunciation Window and is accompanied by a single aural tone. Text appearing in the Annunciation Window is YELLOW. A caution alert is also accompanied by a flashing **CAUTION** Softkey annunciation, as shown in Figure 12-3. Pressing the **CAUTION** Softkey acknowledges the presence of the caution alert.

- **MESSAGE ADVISORY:** This level of alert provides general information to the pilot. A message advisory alert does not issue annunciations in the Annunciation Window. Instead, message advisory alerts only issue a flashing **ADVISORY** Softkey annunciation, as shown in Figure 12-4. Pressing the **ADVISORY** Softkey acknowledges the presence of the message advisory alert and displays the alert text message in the Alerts Window.
12.2 NAV III AIRCRAFT ALERTS

The following alerts are configured specifically for the Cessna Nav III aircraft. See the Cessna Pilot’s Operating Handbook (POH) for information regarding pilot responses.

WARNING Alerts

<table>
<thead>
<tr>
<th>Annunciation Window Text</th>
<th>Audio Alert</th>
</tr>
</thead>
<tbody>
<tr>
<td>OIL PRESSURE</td>
<td>Continuous Aural Tone</td>
</tr>
<tr>
<td>LOW VOLTS</td>
<td>Continuous Aural Tone*</td>
</tr>
<tr>
<td>HIGH VOLTS</td>
<td>Continuous Aural Tone</td>
</tr>
<tr>
<td>CO_LVL_HIGH</td>
<td>Continuous Aural Tone</td>
</tr>
<tr>
<td>PITCH TRIM**</td>
<td>No Tone</td>
</tr>
</tbody>
</table>

* Aural tone is inhibited while the aircraft is on the ground.
** KAP 140 installations only

CAUTION Alerts

<table>
<thead>
<tr>
<th>Annunciation Window Text</th>
<th>Audio Alert</th>
</tr>
</thead>
<tbody>
<tr>
<td>LOW VACUUM</td>
<td>Single Aural Tone</td>
</tr>
<tr>
<td>LOW FUEL L</td>
<td>Single Aural Tone</td>
</tr>
<tr>
<td>LOW FUEL R</td>
<td>Single Aural Tone</td>
</tr>
<tr>
<td>STBY BATT</td>
<td>Single Aural Tone</td>
</tr>
</tbody>
</table>

CAUTION Alerts (T182, T206, and 206 with Prop De-Ice Only)

<table>
<thead>
<tr>
<th>Annunciation Window Text</th>
<th>Audio Alert</th>
</tr>
</thead>
<tbody>
<tr>
<td>PROP HEAT</td>
<td>Single Aural Tone</td>
</tr>
</tbody>
</table>

Safe Operating Annunciation (T182, T206, and 206 with Prop De-Ice Only)

<table>
<thead>
<tr>
<th>Annunciation Window Text</th>
<th>Audio Alert</th>
</tr>
</thead>
<tbody>
<tr>
<td>PROP HEAT</td>
<td>No Tone</td>
</tr>
</tbody>
</table>

12.3 CO GUARDIAN MESSAGES

<table>
<thead>
<tr>
<th>Alerts Window Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>CO DET SRVC – The carbon monoxide detector needs service.</td>
<td>There is a problem within the CO Guardian that requires services.</td>
</tr>
<tr>
<td>CO DET FAIL – The carbon monoxide detector is inoperative.</td>
<td>Loss of communication between the G1000 and the CO Guardian.</td>
</tr>
</tbody>
</table>
### 12.4 AFCS ALERTS (182 & 206 ONLY)

#### System Status Annunciation

The following alert annunciations appear in the AFCS System Status Annunciation on the PFD.

<table>
<thead>
<tr>
<th>Condition</th>
<th>Annunciation</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>Pitch Failure</td>
<td>PITCH</td>
<td>Pitch axis control failure. AP is inoperative.</td>
</tr>
<tr>
<td>Roll Failure</td>
<td>ROLL</td>
<td>Roll axis control failure. AP is inoperative.</td>
</tr>
<tr>
<td>MET Switch Stuck, or Pitch Trim Axis Control Failure</td>
<td>PTRM</td>
<td>If annunciated when AP is engaged, take control of the aircraft and disengage the autopilot. If annunciated when AP is not engaged, move each half of the MET switch separately to check if a stuck switch is causing the annunciation.</td>
</tr>
<tr>
<td>System Failure</td>
<td>AFCS</td>
<td>AP and MET are unavailable. FD may still be available.</td>
</tr>
<tr>
<td>Elevator Mistrim Up</td>
<td>↑ELE</td>
<td>A condition has developed causing the pitch servo to provide a sustained force. Be prepared to apply nose up control wheel force upon autopilot disconnect.</td>
</tr>
<tr>
<td>Elevator Mistrim Down</td>
<td>↓ELE</td>
<td>A condition has developed causing the pitch servo to provide a sustained force. Be prepared to apply nose down control wheel force upon autopilot disconnect.</td>
</tr>
<tr>
<td>Aileron Mistrim Left</td>
<td>←AIL</td>
<td>A condition has developed causing the roll servo to provide a sustained left force. Ensure the slip/skid indicator is centered and observe any maximum fuel imbalance limits.</td>
</tr>
<tr>
<td>Aileron Mistrim Right</td>
<td>AIL→</td>
<td>A condition has developed causing the roll servo to provide a sustained right force. Ensure the slip/skid indicator is centered and observe any maximum fuel imbalance limits.</td>
</tr>
<tr>
<td>Preflight Test</td>
<td>PFT</td>
<td>Performing preflight system test. Upon completion of the test, the aural alert will be heard.</td>
</tr>
<tr>
<td></td>
<td>PFT</td>
<td>Preflight system test has failed.</td>
</tr>
</tbody>
</table>
### 12.5 TAWS ALERTS

The following table shows the possible TAWS alert types with corresponding annunciations and aural messages.

<table>
<thead>
<tr>
<th>Alert Type</th>
<th>PFD/MFD TAWS Page Annunciation</th>
<th>MFD Pop-Up Alert</th>
<th>Aural Message</th>
</tr>
</thead>
<tbody>
<tr>
<td>Excessive Descent Rate Warning (EDR)</td>
<td>PULL UP</td>
<td>PULL UP</td>
<td>“Pull Up”</td>
</tr>
<tr>
<td>Reduced Required Terrain Clearance Warning (RTC)</td>
<td>PULL UP</td>
<td>TERRAIN - PULL UP</td>
<td>“Terrain, Terrain; Pull Up, Pull Up” or “Terrain Ahead, Pull Up; Terrain Ahead, Pull Up”</td>
</tr>
<tr>
<td>Imminent Terrain Impact Warning (ITI)</td>
<td>PULL UP</td>
<td>TERRAIN AHEAD - PULL-UP</td>
<td>“Terrain Ahead, Pull Up; Terrain Ahead, Pull Up” or “Terrain, Terrain; Pull Up, Pull Up”</td>
</tr>
<tr>
<td>Reduced Required Obstacle Clearance Warning (ROC)</td>
<td>PULL UP</td>
<td>OBSTACLE - PULL-UP</td>
<td>“Obstacle, Obstacle; Pull Up, Pull Up” or “Obstacle Ahead, Pull Up; Obstacle Ahead, Pull Up”</td>
</tr>
<tr>
<td>Imminent Obstacle Impact Warning (IOI)</td>
<td>PULL UP</td>
<td>OBSTACLE AHEAD - PULL-UP</td>
<td>“Obstacle Ahead, Pull Up; Obstacle Ahead, Pull Up” or “Obstacle, Obstacle; Pull Up, Pull Up”</td>
</tr>
<tr>
<td>Reduced Required Terrain Clearance Caution (RTC)</td>
<td>TERRAIN</td>
<td>CAUTION - TERRAIN</td>
<td>“Caution, Terrain; Caution, Terrain” or “Terrain Ahead; Terrain Ahead”</td>
</tr>
<tr>
<td>Imminent Terrain Impact Caution (ITI)</td>
<td>TERRAIN</td>
<td>TERRAIN AHEAD</td>
<td>“Terrain Ahead; Terrain Ahead” or “Caution, Terrain; Caution, Terrain”</td>
</tr>
<tr>
<td>Reduced Required Obstacle Clearance Caution (ROC)</td>
<td>TERRAIN</td>
<td>CAUTION - OBSTACLE</td>
<td>“Caution, Obstacle; Caution, Obstacle” or “Obstacle Ahead; Obstacle Ahead”</td>
</tr>
<tr>
<td>Imminent Obstacle Impact Caution (IOI)</td>
<td>TERRAIN</td>
<td>OBSTACLE AHEAD</td>
<td>“Obstacle Ahead; Obstacle Ahead” or “Caution, Obstacle; Caution, Obstacle”</td>
</tr>
<tr>
<td>Premature Descent Alert Caution (PDA)</td>
<td>TERRAIN</td>
<td>TOO LOW - TERRAIN</td>
<td>“Too Low, Terrain”</td>
</tr>
<tr>
<td>Altitude Callout “500”</td>
<td>None</td>
<td>None</td>
<td>“Five-Hundred”</td>
</tr>
<tr>
<td>Excessive Descent Rate Caution (EDR)</td>
<td>TERRAIN</td>
<td>SINK RATE</td>
<td>“Sink Rate”</td>
</tr>
<tr>
<td>Negative Climb Rate Caution (NCR)</td>
<td>TERRAIN</td>
<td>DON’T SINK</td>
<td>“Don’t Sink” or “Too Low, Terrain”</td>
</tr>
</tbody>
</table>
The following system status annunciations may also be issued.

<table>
<thead>
<tr>
<th>Alert Type</th>
<th>PFD/MFD TAWS Page Annunciation</th>
<th>MFD Pop-Up Alert</th>
<th>Aural Message</th>
</tr>
</thead>
<tbody>
<tr>
<td>TAWS System Test Fail</td>
<td>TAWS FAIL</td>
<td>None</td>
<td>“TAWS System Failure”</td>
</tr>
<tr>
<td>TAWS Alerting is disabled</td>
<td>TAWS INHB</td>
<td>None</td>
<td>None</td>
</tr>
<tr>
<td>No GPS position or excessively degraded GPS signal</td>
<td>TAWS N/A</td>
<td>None</td>
<td>“TAWS Not Available”</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>“TAWS Available” will be heard when sufficient GPS signal is re-established.</td>
</tr>
<tr>
<td>System Test in progress</td>
<td>TAWS TEST</td>
<td>None</td>
<td>None</td>
</tr>
<tr>
<td>System Test pass</td>
<td>None</td>
<td>None</td>
<td>“TAWS System Test OK”</td>
</tr>
</tbody>
</table>

### 12.6 G1000 SYSTEM ANNUNCIATIONS

When an LRU or an LRU function fails, a large red ‘X’ is typically displayed on areas associated with the failed data. Refer to the POH for additional information regarding pilot responses to these annunciations.

**NOTE:** Upon power-up of the G1000 system, certain boxes remain invalid as G1000 equipment begins to initialize. All boxes should be operational within one minute of power-up. Should any box continue to remain flagged, the G1000 system should be serviced by a Garmin-authorized repair facility.
## Section 12 – Annunciations & Alerts

<table>
<thead>
<tr>
<th>System Annunciation</th>
<th>Comment</th>
</tr>
</thead>
<tbody>
<tr>
<td><img src="" alt="CALIBRATE AHRS/MAG" /></td>
<td>Indicates a configuration module failure.</td>
</tr>
<tr>
<td><img src="" alt="CHECK ATTITUDE" /></td>
<td>This annunciation will only be seen when the autopilot is engaged. The annunciation indicates an AHRS monitor has detected an abnormal flight parameter, possibly caused by strong turbulence. In this case, the situation should correct itself within a few seconds. If there is an actual failure, a red “X” will soon appear over the Attitude Indicator.</td>
</tr>
<tr>
<td><img src="" alt="AIRSPEED" /></td>
<td>Display system is not receiving airspeed input from air data computer.</td>
</tr>
<tr>
<td><img src="" alt="ALTITUDE" /></td>
<td>Display is not receiving altitude input from the air data computer.</td>
</tr>
<tr>
<td><img src="" alt="VERTICAL SPEED" /></td>
<td>Display is not receiving vertical speed input from the air data computer.</td>
</tr>
<tr>
<td><img src="" alt="HDG" /></td>
<td>Display is not receiving valid heading input from AHRS.</td>
</tr>
</tbody>
</table>
SECTION 12 – ANNUNCIATIONS & ALERTS

<table>
<thead>
<tr>
<th>System Annunciation</th>
<th>Comment</th>
</tr>
</thead>
<tbody>
<tr>
<td><img src="image1" alt="Display" /></td>
<td>Display is not receiving valid transponder information.</td>
</tr>
<tr>
<td><img src="image2" alt="Display" /></td>
<td>‘LOI’ Indicates Loss of Integrity of GPS information. GPS information is either not present or is invalid for navigation use. ‘DR’ may also be seen indicating that GPS is in Dead Reckoning Mode. Note that AHRS utilizes GPS inputs during normal operation. AHRS operation may be degraded if GPS signals are not present (see AFMS).</td>
</tr>
<tr>
<td><img src="image3" alt="Display" /></td>
<td>A red ‘X’ through any other display field, such as engine instrumentation fields, indicates that the field is not receiving valid data.</td>
</tr>
</tbody>
</table>

A red ‘X’ may be the result of an LRU or an LRU function failure. The Figure 12-6 illustrates all possible flags and the responsible LRUs.

**Figure 12-6 G1000 System Failure Annunciations**

Garmin G1000 Cockpit Reference Guide for the Cessna Nav III
**TIS Voice Alerts**

The following voice alerts can be configured for ‘Male’ or ‘Female’ gender by using the Aux System Setup Page on the MFD.

<table>
<thead>
<tr>
<th>Voice Alert</th>
<th>Alert Trigger</th>
</tr>
</thead>
<tbody>
<tr>
<td>“Traffic”</td>
<td>Played when a Traffic Advisory (TA) is issued.</td>
</tr>
<tr>
<td>“Traffic Not Available”</td>
<td>Played when the traffic system fails or cannot communicate.</td>
</tr>
</tbody>
</table>

**NOTE:** Voice alerts are provided to the G1000 by GIA 63/W #1. Should this unit fail, audio and voice alerts will no longer be available.

---

**12.7 G1000 SYSTEM MESSAGE ADVISORIES**

This section describes various G1000 system message advisories. Certain messages are issued due to an LRU or an LRU function failure. Such messages are normally accompanied by a corresponding red ‘X’ annunciation as shown previously in the G1000 System Annunciation section.

**NOTE:** This section provides information regarding G1000 message advisories that may be displayed by the system. Knowledge of the aircraft, systems, flight conditions, and other existing operational priorities must be considered when responding to a message. Always use sound pilot judgment. The Cessna aircraft Pilot’s Operating Handbook (POH) takes precedence over any conflicting guidance found in this section.
### MFD & PFD Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>DATA LOST – Pilot stored</td>
<td>The pilot profile data was lost. System reverts to default pilot profile and settings. The pilot may reconfigure the MFD &amp; PFD with preferred settings, if desired.</td>
</tr>
<tr>
<td>data was lost. Recheck</td>
<td></td>
</tr>
<tr>
<td>settings.</td>
<td></td>
</tr>
<tr>
<td>XTalk ERROR – A flight</td>
<td>The MFD and PFD are not communicating with each other. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>display crosstalk error</td>
<td></td>
</tr>
<tr>
<td>has occurred.</td>
<td></td>
</tr>
<tr>
<td>PFD1 SERVICE – PFD1 needs</td>
<td>The PFD and/or MFD self-test has detected a problem. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>service. Return unit for</td>
<td></td>
</tr>
<tr>
<td>repair.</td>
<td></td>
</tr>
<tr>
<td>MFD1 SERVICE – MFD1 needs</td>
<td></td>
</tr>
<tr>
<td>service. Return unit for</td>
<td></td>
</tr>
<tr>
<td>repair.</td>
<td></td>
</tr>
<tr>
<td>MANIFEST – PFD1 software</td>
<td>The PFD and/or MFD has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>mismatch. Communication</td>
<td></td>
</tr>
<tr>
<td>halted.</td>
<td></td>
</tr>
<tr>
<td>MANIFEST – MFD1 software</td>
<td></td>
</tr>
<tr>
<td>mismatch. Communication</td>
<td></td>
</tr>
<tr>
<td>halted.</td>
<td></td>
</tr>
<tr>
<td>PFD1 CONFIG – PFD1</td>
<td>The PFD and/or MFD configuration settings do not match backup configuration memory. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>configuration error.</td>
<td></td>
</tr>
<tr>
<td>Config service req’d.</td>
<td></td>
</tr>
<tr>
<td>MFD1 CONFIG – MFD1</td>
<td></td>
</tr>
<tr>
<td>configuration error.</td>
<td></td>
</tr>
<tr>
<td>Config service req’d.</td>
<td></td>
</tr>
<tr>
<td>SW MISMATCH – GDU</td>
<td>The MFD and PFD have different software versions installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>software version</td>
<td></td>
</tr>
<tr>
<td>mismatch. Xtalk is off.</td>
<td></td>
</tr>
<tr>
<td>PFD1 COOLING – PFD1</td>
<td>The PFD and/or MFD is overheating and is reducing power consumption by dimming the display. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>has poor cooling. Reducing</td>
<td></td>
</tr>
<tr>
<td>power usage.</td>
<td></td>
</tr>
<tr>
<td>MFD1 COOLING – MFD1</td>
<td></td>
</tr>
<tr>
<td>has poor cooling. Reducing</td>
<td></td>
</tr>
<tr>
<td>power usage.</td>
<td></td>
</tr>
<tr>
<td>PFD1 “KEY” KEYSKT – Key</td>
<td>A key is stuck on the PFD and/or MFD bezel. Attempt to free the stuck key by pressing it several times. The G1000 system should be serviced if the problem persists.</td>
</tr>
<tr>
<td>is stuck.</td>
<td></td>
</tr>
<tr>
<td>MFD1 “KEY” KEYSKT – Key</td>
<td></td>
</tr>
<tr>
<td>is stuck.</td>
<td></td>
</tr>
<tr>
<td>CNFG MODULE – PFD1</td>
<td>The PFD1 configuration module backup memory has failed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>configuration module is</td>
<td></td>
</tr>
<tr>
<td>inoperative.</td>
<td></td>
</tr>
<tr>
<td>PFD1 VOLTAGE – PFD1</td>
<td></td>
</tr>
<tr>
<td>has low voltage. Reducing</td>
<td></td>
</tr>
<tr>
<td>power usage</td>
<td></td>
</tr>
<tr>
<td>MFD1 VOLTAGE – MFD1</td>
<td></td>
</tr>
<tr>
<td>has low voltage. Reducing</td>
<td></td>
</tr>
<tr>
<td>power usage</td>
<td></td>
</tr>
</tbody>
</table>
### Database Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 aviation database error exists.</td>
<td>The MFD and/or PFD detected a failure in the aviation database. Attempt to reload the aviation database. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>PFD1 DB ERR</strong> – PFD1 aviation database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 basemap database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>PFD1 DB ERR</strong> – PFD1 basemap database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 terrain database error exists.</td>
<td>The MFD and/or PFD detected a failure in the terrain database. Ensure that the terrain card is properly inserted in display. Replace terrain card. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>PFD1 DB ERR</strong> – PFD1 terrain database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 obstacle database error exists.</td>
<td>The MFD and/or PFD detected a failure in the obstacle database. Ensure that the data card is properly inserted. Replace data card. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>PFD1 DB ERR</strong> – PFD1 obstacle database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 airport terrain database error exists.</td>
<td>The MFD and/or PFD detected a failure in the airport terrain database. Ensure that the data card is properly inserted. Replace data card. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>PFD1 DB ERR</strong> – PFD1 airport terrain database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 SafeTaxi database error exists.</td>
<td>The MFD and/or PFD detected a failure in the Safe Taxi database. Ensure that the data card is properly inserted. Replace data card. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>PFD1 DB ERR</strong> – PFD1 SafeTaxi database error exists.</td>
<td></td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 Chartview database error exists.</td>
<td>The MFD and/or PFD detected a failure in the ChartView database (optional feature). Ensure that the data card is properly inserted. Replace data card. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>MFD1 DB ERR</strong> – MFD1 FliteCharts database error exists.</td>
<td>The MFD and/or PFD detected a failure in the FliteCharts database (optional feature). Ensure that the data card is properly inserted. Replace data card. If problem persists, the G1000 system should be serviced.</td>
</tr>
</tbody>
</table>
## Database Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>DB MISMATCH – Aviation database version mismatch. Xtalk is off.</td>
<td>The PFD and MFD have different aviation database versions installed. Crossfill is off. Install correct aviation database version in both displays.</td>
</tr>
<tr>
<td>DB MISMATCH – Aviation database type mismatch. Xtalk is off.</td>
<td>The PFD and MFD have different aviation database types installed (Americas, European, etc.). Crossfill is off. Install correct aviation database type in both displays.</td>
</tr>
<tr>
<td>DB MISMATCH – Terrain database version mismatch. Xtalk is off.</td>
<td>The PFD and MFD have different terrain database versions installed. Crossfill is off. Install correct terrain database version in both displays.</td>
</tr>
<tr>
<td>DB MISMATCH – Terrain database type mismatch. Xtalk is off.</td>
<td>The PFD and MFD have different terrain database types installed. Crossfill is off. Install correct terrain database type in both displays.</td>
</tr>
<tr>
<td>DB MISMATCH – Obstacle database version mismatch. Xtalk is off</td>
<td>The PFD and MFD have different obstacle database versions installed. Crossfill is off. Install correct obstacle database version in both displays.</td>
</tr>
<tr>
<td>DB MISMATCH – Airport Terrain database mismatch. Xtalk is off</td>
<td>The PFD and MFD have different airport terrain databases installed. Crossfill is off. Install correct airport terrain database in both displays.</td>
</tr>
</tbody>
</table>

## GMA 1347 Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>GMA1 FAIL – GMA1 is inoperative.</td>
<td>The audio panel self-test has detected a failure. The audio panel is unavailable. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GMA1 CONFIG – GMA1 configuration error. Config service req’d.</td>
<td>The audio panel configuration settings do not match backup configuration memory. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST – GMA1 software mismatch. Communication halted.</td>
<td>The audio panel has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GMA1 SERVICE – GMA1 needs service. Return unit for repair.</td>
<td>The audio panel self-test has detected a problem in the unit. Certain audio functions may still be available, and the audio panel may still be usable. The G1000 system should be serviced when possible.</td>
</tr>
</tbody>
</table>
## GIA 63 Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>GIA1 CONFIG</td>
<td>The GIA1 and/or GIA2 configuration settings do not match backup configuration memory. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GIA2 CONFIG</td>
<td>The GIA1 and/or GIA2 have an error in the audio configuration. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GIA1 COOLING</td>
<td>The GIA1 and/or GIA2 temperature is too low to operate correctly. Allow units to warm up to operating temperature.</td>
</tr>
<tr>
<td>GIA2 COOLING</td>
<td>The GIA1 and/or GIA2 temperature is too high. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>GIA1 SERVICE</td>
<td>The GIA1 and/or GIA2 self-test has detected a problem in the unit. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GIA2 SERVICE</td>
<td>The GIA1 and/or GIA2 has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST</td>
<td>The system has detected an over temperature condition in COM1 and/or COM2. The transmitter will operate at reduced power. If the problem persists, the G1000 system should be serviced.</td>
</tr>
</tbody>
</table>
## GIA 63 Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>COM1 SERVICE – COM1 needs service. Return unit for repair.</td>
<td>The system has detected a failure in COM1 and/or COM2. COM1 and/or COM2 may still be usable. The G1000 system should be serviced when possible.</td>
</tr>
<tr>
<td>COM2 SERVICE – COM2 needs service. Return unit for repair.</td>
<td></td>
</tr>
<tr>
<td>COM1 PTT – COM1 push-to-talk key is stuck.</td>
<td>The COM1 and/or COM2 external push-to-talk switch is stuck in the enable (or “pressed”) position. Press the PTT switch again to cycle its operation. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>COM2 PTT – COM2 push-to-talk key is stuck.</td>
<td></td>
</tr>
<tr>
<td>COM1 RMT XFR – COM1 remote transfer key is stuck.</td>
<td>The COM1 and/or COM2 transfer switch is stuck in the enabled (or “pressed”) position. Press the transfer switch again to cycle its operation. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>COM2 RMT XFR – COM2 remote transfer key is stuck.</td>
<td></td>
</tr>
<tr>
<td>RAIM UNAVAIL – RAIM is not available from FAF to MAP waypoints.</td>
<td>GPS satellite coverage is insufficient to perform Receiver Autonomous Integrity Monitoring (RAIM) from the FAF to the MAP waypoints.</td>
</tr>
<tr>
<td>LOI – GPS integrity lost. Crosscheck with other NAVS.</td>
<td>Loss of GPS integrity monitoring.</td>
</tr>
<tr>
<td>ABORT APR – Loss of GPS navigation. Abort approach.</td>
<td>Abort approach due to loss of GPS navigation.</td>
</tr>
<tr>
<td>TRUE APR – True north approach. All angles are TRUE.</td>
<td>Displayed after passing the first waypoint of a true north approach when the nav angle is set to ‘AUTO’.</td>
</tr>
<tr>
<td>GPS1 FAIL – GPS1 is inoperative.</td>
<td>A failure has been detected in the GPS1 and/or GPS2 receiver. The receiver is unavailable. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GPS2 FAIL – GPS2 is inoperative.</td>
<td></td>
</tr>
</tbody>
</table>
GIA 63 Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>GPS1 SERVICE – GPS1 needs service. Return unit for repair.</td>
<td>A failure has been detected in the GPS1 and/or GPS2 receiver. The receiver may still be available. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GPS2 SERVICE – GPS2 needs service. Return unit for repair.</td>
<td></td>
</tr>
<tr>
<td>NAV1 SERVICE – NAV1 needs service. Return unit for repair.</td>
<td>A failure has been detected in the NAV1 and/or NAV2 receiver. The receiver may still be available. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>NAV2 SERVICE – NAV2 needs service. Return unit for repair.</td>
<td></td>
</tr>
<tr>
<td>NAV1 RMT XFR – NAV1 remote transfer key is stuck.</td>
<td>The remote NAV1 and/or NAV2 transfer switch is stuck in the enabled (or “pressed”) state. Press the transfer switch again to cycle its operation. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>NAV2 RMT XFR – NAV2 remote transfer key is stuck.</td>
<td></td>
</tr>
<tr>
<td>G/S1 FAIL – G/S1 is inoperative.</td>
<td>A failure has been detected in glideslope receiver 1 and/or receiver 2. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>G/S2 FAIL – G/S2 is inoperative.</td>
<td></td>
</tr>
<tr>
<td>G/S1 SERVICE – G/S1 needs service. Return unit for repair.</td>
<td>A failure has been detected in glideslope receiver 1 and/or receiver 2. The receiver may still be available. The G1000 system should be serviced when possible.</td>
</tr>
<tr>
<td>G/S2 SERVICE – G/S2 needs service. Return unit for repair.</td>
<td></td>
</tr>
</tbody>
</table>

GIA 63W Message Advisories

| Message              | Comments                                                                 |
## GIA 63W Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td><strong>GIA1 COOLING</strong> – GIA1 temperature too low.</td>
<td>The GIA1 and/or GIA2 temperature is too low to operate correctly. Allow units to warm up to operating temperature.</td>
</tr>
<tr>
<td><strong>GIA2 COOLING</strong> – GIA2 temperature too low.</td>
<td></td>
</tr>
<tr>
<td><strong>GIA1 COOLING</strong> – GIA1 over temperature.</td>
<td>The GIA1 and/or GIA2 temperature is too high. If problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>GIA2 COOLING</strong> – GIA2 over temperature.</td>
<td></td>
</tr>
<tr>
<td><strong>GIA1 SERVICE</strong> – GIA1 needs service. Return the unit for repair.</td>
<td>The GIA1 and/or GIA2 self-test has detected a problem in the unit. The G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>GIA2 SERVICE</strong> – GIA2 needs service. Return the unit for repair.</td>
<td></td>
</tr>
<tr>
<td><strong>HW MISMATCH</strong> – GIA hardware mismatch. GIA1 communication halted.</td>
<td>A GIA mismatch has been detected, where only one is WAAS capable.</td>
</tr>
<tr>
<td><strong>HW MISMATCH</strong> – GIA hardware mismatch. GIA2 communication halted.</td>
<td></td>
</tr>
<tr>
<td><strong>MANIFEST</strong> – GIA1 software mismatch. Communication halted.</td>
<td>The GIA1 and/or GIA 2 has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>MANIFEST</strong> – GIA2 software mismatch. Communication halted.</td>
<td></td>
</tr>
<tr>
<td><strong>COM1 TEMP</strong> – COM1 over temp. Reducing transmitter power.</td>
<td>The system has detected an over temperature condition in COM1 and/or COM2. The transmitter will operate at reduced power. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td><strong>COM2 TEMP</strong> – COM2 over temp. Reducing transmitter power.</td>
<td></td>
</tr>
<tr>
<td><strong>COM1 SERVICE</strong> – COM1 needs service. Return unit for repair.</td>
<td>The system has detected a failure in COM1 and/or COM2. COM1 and/or COM2 may still be usable. The G1000 system should be serviced when possible.</td>
</tr>
<tr>
<td><strong>COM2 SERVICE</strong> – COM2 needs service. Return unit for repair.</td>
<td></td>
</tr>
</tbody>
</table>
### GIA 63W Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>COM1 PTT – COM1 push-to-talk key is stuck.</td>
<td>The COM1 and/or COM2 external push-to-talk switch is stuck in the enable (or “pressed”) position. Press the PTT switch again to cycle its operation. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>COM2 PTT – COM2 push-to-talk key is stuck.</td>
<td></td>
</tr>
<tr>
<td>COM1 RMT XFR – COM1 remote transfer key is stuck.</td>
<td>The COM1 and/or COM2 transfer switch is stuck in the enabled (or “pressed”) position. Press the transfer switch again to cycle its operation. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>COM2 RMT XFR – COM2 remote transfer key is stuck.</td>
<td></td>
</tr>
<tr>
<td>LOI – GPS integrity lost. Crosscheck with other NAVS.</td>
<td>GPS integrity is insufficient for the current phase of flight.</td>
</tr>
<tr>
<td>ABORT APR – Loss of GPS navigation. Abort approach.</td>
<td>Abort approach due to loss of GPS navigation.</td>
</tr>
<tr>
<td>APR DWNGRADE – Approach downgraded. Use LNAV minima.</td>
<td>Use LNAV minima when approach is downgraded.</td>
</tr>
<tr>
<td>TRUE APR – True north approach. All angles are TRUE.</td>
<td>Displayed after passing the first waypoint of a true north approach when the nav angle is set to ‘AUTO’.</td>
</tr>
<tr>
<td>GPS1 FAIL – GPS1 is inoperative.</td>
<td>A failure has been detected in the GPS1 and/or GPS2 receiver. The receiver is unavailable. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GPS2 FAIL – GPS2 is inoperative.</td>
<td></td>
</tr>
<tr>
<td>GPS1 SERVICE – GPS1 needs service. Return unit for repair.</td>
<td>A failure has been detected in the GPS1 and/or GPS2 receiver. The receiver may still be available. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GPS2 SERVICE – GPS2 needs service. Return unit for repair.</td>
<td></td>
</tr>
</tbody>
</table>
### GIA 63W Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>NAV1 SERVICE – NAV1 needs service. Return unit for repair.</td>
<td>A failure has been detected in the NAV1 and/or NAV2 receiver. The receiver may still be available. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>NAV2 SERVICE – NAV2 needs service. Return unit for repair.</td>
<td></td>
</tr>
<tr>
<td>NAV1 RMT XFR – NAV1 remote transfer key is stuck.</td>
<td>The remote NAV1 and/or NAV2 transfer switch is stuck in the enabled (or “pressed”) state. Press the transfer switch again to cycle its operation. If the problem persists, the G1000 system should be serviced.</td>
</tr>
<tr>
<td>NAV2 RMT XFR – NAV2 remote transfer key is stuck.</td>
<td></td>
</tr>
<tr>
<td>G/S1 FAIL – G/S1 is inoperative.</td>
<td>A failure has been detected in glideslope receiver 1 and/or receiver 2. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>G/S2 FAIL – G/S2 is inoperative.</td>
<td></td>
</tr>
<tr>
<td>G/S1 SERVICE – G/S1 needs service. Return unit for repair.</td>
<td>A failure has been detected in glideslope receiver 1 and/or receiver 2. The receiver may still be available. The G1000 system should be serviced when possible.</td>
</tr>
<tr>
<td>G/S2 SERVICE – G/S2 needs service. Return unit for repair.</td>
<td></td>
</tr>
</tbody>
</table>

### GEA 71 Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>GEA1 CONFIG – GEA1 configuration error. Config service req’d.</td>
<td>The GEA1 configuration settings do not match those of backup configuration memory. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST – GEA1 software mismatch. Communication halted.</td>
<td>The #1 GEA 71 has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST – GEA2 software mismatch. Communication halted.</td>
<td>The #2 GEA 71 has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
</tbody>
</table>
### GTX 33 Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>XPDR1 CONFIG</td>
<td>The transponder configuration settings do not match those of backup configuration memory. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST</td>
<td>The transponder has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>XPDR1 SRVC</td>
<td>The #1 transponder should be serviced when possible.</td>
</tr>
<tr>
<td>XPDR1 FAIL</td>
<td>There is no communication with the #1 transponder.</td>
</tr>
</tbody>
</table>

### GRS 77 Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>AHRS1 TAS</td>
<td>The #1 AHRS is not receiving true airspeed from the air data computer. The AHRS relies on GPS information to augment the lack of airspeed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>AHRS1 GPS</td>
<td>The #1 AHRS is using the backup GPS path. Primary GPS path has failed. The G1000 system should be serviced when possible.</td>
</tr>
<tr>
<td>AHRS1 GPS</td>
<td>The #1 AHRS is not receiving any or any useful GPS information. Check AFMS limitations. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>AHRS1 GPS</td>
<td>The #1 AHRS is not receiving backup GPS information. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>AHRS1 GPS</td>
<td>The #1 AHRS is operating exclusively in no-GPS mode. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>AHRS MAG DB</td>
<td>The #1 AHRS and #2 AHRS magnetic model database versions do not match.</td>
</tr>
<tr>
<td>AHRS1 SRVC</td>
<td>The #1 AHRS earth magnetic field model is out of date. Update magnetic field model when practical.</td>
</tr>
</tbody>
</table>
### GRS 77 Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>GEO LIMITS – AHRS1 too far North/South, no magnetic compass.</td>
<td>The aircraft is outside geographical limits for approved AHRS operation. Heading is flagged as invalid.</td>
</tr>
<tr>
<td>MANIFEST – GRS1 software mismatch. Communication halted.</td>
<td>The #1 AHRS has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
</tbody>
</table>

### GMU 44 Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>HDG FAULT – AHRS1 magnetometer fault has occurred.</td>
<td>A fault has occurred in the #1 GMU 44. Heading is flagged as invalid. The AHRS uses GPS for backup mode operation. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST – GMU1 software mismatch. Communication halted.</td>
<td>The GMU 44 has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
</tbody>
</table>

### GDL 69A Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>GDL69 CONFIG – GDL 69 configuration error. Config service req’d.</td>
<td>GDL 69 configuration settings do not match those of backup configuration memory. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>GDL69 FAIL – GDL 69 has failed.</td>
<td>A failure has been detected in the GDL 69. The receiver is unavailable. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>MANIFEST – GDL software mismatch. Communication halted.</td>
<td>The GDL 69 has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
</tbody>
</table>

### GDC 74A Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>ADC1 ALT EC – ADC1 altitude error correction is unavailable.</td>
<td>GDC1 or GDC2 is reporting that the altitude error correction is unavailable.</td>
</tr>
<tr>
<td>ADC1 AS EC – ADC1 airspeed error correction is unavailable.</td>
<td>GDC1 or GDC2 is reporting that the airspeed error correction is unavailable.</td>
</tr>
<tr>
<td>MANIFEST – GDC1 software mismatch. Communication halted.</td>
<td>The GDC 74A has incorrect software installed. The G1000 system should be serviced.</td>
</tr>
</tbody>
</table>
## Miscellaneous Message Advisories

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td><strong>FPL WPT LOCK</strong> – Flight plan waypoint is locked.</td>
<td>Upon power-up, the G1000 system detects that a stored flight plan waypoint is locked. This occurs when an aviation database update eliminates an obsolete waypoint. The flight plan cannot find the specified waypoint and flags this message. This can also occur with user waypoints in a flight plan that is deleted. Remove the waypoint from the flight plan if it no longer exists in any database, OR update the waypoint name/identifier to reflect the new information.</td>
</tr>
<tr>
<td><strong>FPL WPT MOVE</strong> – Flight plan waypoint moved.</td>
<td>The system has detected that a waypoint coordinate has changed due to a new aviation database update. Verify that stored flight plans contain correct waypoint locations.</td>
</tr>
<tr>
<td><strong>TIMER EXPIRD</strong> – Timer has expired.</td>
<td>The system notifies the pilot that the timer has expired.</td>
</tr>
<tr>
<td><strong>DB CHANGE</strong> – Database changed. Verify user modified procedures.</td>
<td>This occurs when a stored flight plan contains procedures that have been manually edited. This alert is issued only after an aviation database update. Verify that the user-modified procedures in stored flight plans are correct and up to date.</td>
</tr>
<tr>
<td><strong>FPL TRUNC</strong> – Flight plan has been truncated.</td>
<td>This occurs when a newly installed aviation database eliminates an obsolete approach or arrival used by a stored flight plan. The obsolete procedure is removed from the flight plan. Update flight plan with current arrival or approach.</td>
</tr>
<tr>
<td><strong>LOCKED FPL</strong> – Cannot navigate locked flight plan.</td>
<td>This occurs when the pilot attempts to activate a stored flight plan that contains locked waypoint. Remove locked waypoint from flight plan. Update flight plan with current waypoint.</td>
</tr>
<tr>
<td><strong>WPT ARRIVAL</strong> – Arriving at waypoint [xxxx]</td>
<td>Arriving at waypoint [xxxx], where [xxxx] is the waypoint name.</td>
</tr>
<tr>
<td><strong>STEEP TURN</strong> – Steep turn ahead.</td>
<td>A steep turn is 15 seconds ahead. Prepare to turn.</td>
</tr>
<tr>
<td><strong>INSIDE ARSPC</strong> – Inside airspace.</td>
<td>The aircraft is inside the airspace.</td>
</tr>
<tr>
<td><strong>ARSPC AHEAD</strong> – Airspace ahead less than 10 minutes.</td>
<td>Special use airspace is ahead of aircraft. The aircraft will penetrate the airspace within 10 minutes.</td>
</tr>
<tr>
<td><strong>ARSPC NEAR</strong> – Airspace near and ahead.</td>
<td>Special use airspace is near and ahead of the aircraft position.</td>
</tr>
</tbody>
</table>
## Miscellaneous Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td>ARSPC NEAR – Airspace near – less than 2 nm.</td>
<td>Special use airspace is within 2 nm of the aircraft position.</td>
</tr>
<tr>
<td>LEG UNSMooth – Flight plan leg will not be smooth.</td>
<td>The approaching flight plan waypoints are too close to allow for smooth turns. Prepare for steep turns ahead and expect noticeable course deviations.</td>
</tr>
<tr>
<td>APPR INACTV – Approach is not active.</td>
<td>The system notifies the pilot that the loaded approach is not active. Activate approach when required.</td>
</tr>
<tr>
<td>SLCT FREQ – Select appropriate frequency for approach.</td>
<td>The system notifies the pilot to load the approach frequency for the appropriate NAV receiver. Select the correct frequency for the approach.</td>
</tr>
<tr>
<td>SLCT NAV – Select NAV on CDI for approach.</td>
<td>The system notifies the pilot to set the CDI to the correct NAV receiver. Set the CDI to the correct NAV receiver.</td>
</tr>
<tr>
<td>PTK FAIL – Bad geometry, parallel offset not activated.</td>
<td>Bad parallel track geometry.</td>
</tr>
<tr>
<td>PTK FAIL – Invalid leg type, parallel offset not activated.</td>
<td>Invalid leg type for parallel offset.</td>
</tr>
<tr>
<td>PTK FAIL – Past IAF, parallel offset not activated.</td>
<td>IAF waypoint for parallel offset has been passed.</td>
</tr>
<tr>
<td>UNABLE V WPT – Can’t reach current vertical waypoint.</td>
<td>The current vertical waypoint can not be reached within the maximum flight path angle and vertical speed constraints. The system will automatically transition to the next vertical waypoint.</td>
</tr>
<tr>
<td>VNAV – Unavailable. Unsupported leg type in flight plan.</td>
<td>The lateral flight plan contains a procedure turn, vector, or other unsupported leg type prior to the active vertical waypoint. This will prevent vertical guidance to the active vertical waypoint.</td>
</tr>
<tr>
<td>VNAV – Unavailable. Excessive crosstrack error.</td>
<td>The current crosstrack exceeds the limit, causing vertical deviation to go invalid.</td>
</tr>
<tr>
<td>VNAV – Unavailable. Excessive track angle error.</td>
<td>The current track angle error exceeds the limit, causing the vertical deviation to go invalid.</td>
</tr>
<tr>
<td>VNAV – Unavailable. Parallel course selected.</td>
<td>A parallel course has been selected, causing the vertical deviation to go invalid.</td>
</tr>
<tr>
<td>NO WGS84 WPT – Non WGS 84 waypoint for navigation -[xxxx]</td>
<td>The selected waypoint [xxxx] does not use the WGS 84 datum. Cross-check position with alternate navigation sources.</td>
</tr>
</tbody>
</table>
### Miscellaneous Message Advisories (Cont.)

<table>
<thead>
<tr>
<th>Message</th>
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</tr>
</thead>
<tbody>
<tr>
<td>TRAFFIC FAIL – Traffic device has failed.</td>
<td>The G1000 is no longer receiving data from the traffic system. The traffic device should be serviced.</td>
</tr>
<tr>
<td>STRMSCP FAIL – Stormscope has failed.</td>
<td>Stormscope has failed. The G1000 system should be serviced.</td>
</tr>
<tr>
<td>FAILED PATH – A data path has failed.</td>
<td>A data path connected to the GDU or the GIA 63/W has failed.</td>
</tr>
<tr>
<td>MAG VAR WARN – Large magnetic variance. Verify all course angles.</td>
<td>The GDU’s internal model cannot determine the exact magnetic variance for geographic locations near the magnetic poles. Displayed magnetic course angles may differ from the actual magnetic heading by more than 2°.</td>
</tr>
<tr>
<td>SCHEDULER [#] – &lt;message&gt;.</td>
<td>Message criteria entered by the user.</td>
</tr>
</tbody>
</table>
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